

EA02-025

FORD

10/27/03

BOOK 26 OF 61

PART 2 OF 4

BOX 23 OF 28

**Potential
Failure Mode and Effects Analysis
(Design FMEA)**

C-1000
X - Software
— Subsystem
— Component
Model Name/Version/Build: 1.0
Date Test:

Design: Systematic, etc., page, etc.
Key Notes:

Page number:
Page 1 of 1
Prepared by: Mark Albrecht
Print Date: Mar-13 03:00:13 Rev.: 1.00.00.00

Row #	Potential Failure Mode	Severity and Probability of Failure	C	Potential Cause(s) / Mechanism(s) of Failure	S	Current Design Controls	D	Recommended Action(s)	A		
									R	O	M
				Structural damage	9	0					
				Structural load-collapse	9	0					
				Structural fatigue	9	0					
				Structural strength limits	9	0					
				Structural failure under load / vibration	9	0					
				Structural fatigue	9	0					
				Structural fatigue overextended, until no chance to proper protection	9	0					
				Structural protection oil filter shadowcast validation	9	0					
				Structural design load-up	9	0					
				Sharp edges/corner modifications even to allow operator to fully seat	9	0					
				Design review	9	0					
					0	0					
					0	0					
					0	0					
					0	0					
Secondary lock	Secondary lock table not present	Secondary lock missing, bypass pin not used, connector loose	9	Component affected selected	0	0					
				Secondary lock selection incorrect dimension and influencing inadequate	0	0					
				Secondary lock selection incorrect geometry prototype	0	0					
				Secondary lock selection incorrect material	0	0					
Secondary lock can be separated from assembly for use during test or storage position	Connector can not be seated	Secondary lock can be seated	9	Secondary lock impacted by user	0	0					
				Design allows design insertion	0	0					
				Design allows reverse insertion	0	0					
Secondary lock can be removed between deposit and instrument tank finger	Secondary lock not locked into assembly	Secondary lock never connected with no live connection high pressure device to sealing part	9	Design allows reverse insertion	0	0					
				Design allows reverse insertion	0	0					
				Possible unusual parts note: Tropical lock finger damaged	9	0					
				Design review	0	0					
				Secondary lock insertion force limit	0	0					
				Lock finger damage force limit	0	0					

Printed on: 03/03/13 11:41:00

Produced by Relionline Rev Number: W v 1.0

Failure Mode		Potential Failure Modes Analysis (Design FMEA)				FMEA Number: Page 2 of 4 Prepared by: Mark Abromaitis Print Date: May 21, 2004 20:40:39	
		Failure Modes		Mitigation			
Failure Mode	Severity Level	Mitigation Measures	Mitigation Effectiveness	Failure Mode Description	Failure Probability	Impact	Impact Score
Gauge	Importantly minor	Cleaned, sanitized	None	Unacceptable holding tolerance dimensions and tolerancing	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable handle breakage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate part design	<input type="checkbox"/> <input type="checkbox"/> <input checked="" type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate part design	<input type="checkbox"/> <input type="checkbox"/> <input checked="" type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate part design	<input type="checkbox"/> <input type="checkbox"/> <input checked="" type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate part design	<input type="checkbox"/> <input type="checkbox"/> <input checked="" type="checkbox"/>
Cylindrical gauge	Importantly minor, when tested	Open cracks; Damaged threads	None	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
Small cylindrical drill measurements	Importantly minor, when cleaned	Inadequate, contains open cracks	None	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
Large cylindrical drill measurements	Importantly minor, when cleaned	Inadequate, contains open cracks	None	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
Outer shield 1 service	Importantly minor, when cleaned	Cleaned, sanitized, damaged threads/threading	None	Inadequate hold down force and tolerancing	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Acceptable part damage	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate force required to hold shield 1 securely on inner shield	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate force required to hold shield 1 securely on inner shield	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>
				Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>	Inadequate material selection	<input type="checkbox"/> <input checked="" type="checkbox"/> <input type="checkbox"/>

Engineering Specification

I. GENERAL

- A. This specification replaces ES-FUEB-14A464-AA for Ford electrical connectors, both snap together and bolted, and is an extension of the released drawing on which it is referenced. All requirements of this specification must be met for all connectors designed for Ford applications (whether Ford designed or vendor designed) for the 1990 model year & beyond. Any existing connectors upgraded for a 1990 & beyond application must also comply with this specification. Minimum measures necessary for demonstrating compliance are given in each section.

The engineering test schedule contained within this specification, reflects the minimum values established to provide a regular evaluation of conformance to design intent. The schedule is intended as a supplement to normal material inspections, dimensional checking, and in-process controls; and should in no way adversely influence adherence to other more statistically significant inspection operations.

Preparation and submission of an acceptable Control Plan is the responsibility of the manufacturing source. The Control Plan shall include the inspection process and frequency for dimensional checking and other in-process checks. Control Plan approval according to Q-181 is a prerequisite for initial sample review and approval. The manufacturing source will retain the original of the approved Control Plan and any later revisions per Q-181 and provide a copy to the design responsible PEO.

II. CONNECTION VALIDATION AND IN-PROCESS TESTS

- A. Compliance with all of the requirements in this section must be demonstrated as follows:
1. Production Validation (PV) Tests must be completed satisfactorily with parts from production tooling (and processes where possible) before ISIR approval and authorization for shipment of production parts can be effected. The molding facility must also perform a functional test on each part, using a mating part from each vendor currently sourced by Ford to mold the mating part, prior to final part approval. Parts must be revalidated completely, or per Section V whenever any change is made which could possibly affect part function or performance.

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Engineering Specification

II. PRODUCTION VALIDATION AND IN-PROCESS TESTS - continued

2. In-Process Tests - IP tests must be completed with production parts prior to first shipment and at the specified frequency on a continuing basis. Samples for these tests must be selected on a random basis to represent the entire production population as much as possible. In the event that any portion of these tests are not met, Ford Q101, Section III, paragraph I-1, "GS Test Performance Requirements" shall be invoked with the following exception:

If, after an IP test failure, the suspect lot is not shipped and the problem corrected, normal production can be resumed without notifying Ford Purchasing, SQA or the consuming plant Quality activity.

B. Component Performance Characteristics

The following paragraphs describe the performance requirements for components covered by this specification.

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FDI PD 3947-82 Previous editions may NOT be used

Engineering Specification

II. PRODUCTION VALIDATION AND IN-PROCESS TESTS

(1) Test	(2) Test #	PRODUCTION VALIDATION		IN-PROCESS TESTS	
		(3) Minimum Sample Size ^a	(4) Statistical Test Acceptance Criteria	(5) Minimum Sample Size b, Frequency	(6) Statistical Test Acceptance Criteria
Thermal Cycling	1	12	P.70 = .90	none	
Salt Spray	2a	8	P.57 = .90	none	
Soak Shower	2b	8	P.57 = .90	none	
Submersible Air Leak	2c	24	P.70 = .92	1/cav, per lot	No Failures
Dielectric Test	2d	24 ^b	P.70 = .95	1/cavity per lot	No Failures
Vibration	3	12	P.70 = .90	none	
Mechanical Shock	4	24	P.70 = .95	none	
Terminal Retention	5	24	P.70 = .92	3/cav., per week	No Failures
Housing Retention and Lock Fatigue	6	24	P.70 = .92	none	
Push Nut - Push Out and Turning Force	7	24	P.70 = .92	2/lot	No Failures
Threaded Insert Retention	8	24	P.70 = .92	2/lot	No Failures
Over Torque	9	24	P.70 = .92	2/lot	No Failures
Minimum Assembly Torque	10	12	P.70 = .90	2/week	No Failures
Solder Tail Gauge Test	11	24	P.70 = .95	3x Audit	No Failures
Gasket or Grommet Assurance	12	none		100% Inspection	No Failures
Melt Flow Index	13	3	No Failures	See Sec. III.	No Failures
Unclamping Force	14	24	P.70 = .93	none	
Terminal Insertion	15	24	P.70 = .93	none	

¹ or one sample per connector cavity, whichever is greater.
² for connectors with more than 20 positions, test a minimum of 8 samples, or 1 per tool cavity, whichever is greater.

³ for connectors with more than 20 positions, test 3/cavity per week.
⁴ inspection frequencies for any or all in-process tests may be waived in lieu of an approved statistical process control program which monitors the characteristics listed in the "Process Control Parameter" section of each applicable test (see Ford Controlling Process Control and Process Capability Improvement, order # 80-01-291). Identify these characteristics in the control plan, along with a method to be used for controlling them. The control plan must also include a reaction plan in the event that process capability or control is lost after SPC is initiated. This control plan must be approved by Ford SQA and would supersede any reaction plan called out in this spec. The supplier must first demonstrate that the process is in statistical control and that the process is capable.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS

Test 1. Thermal Cycling

PURPOSE

The purpose of this test is to determine the resistance of an electrical connecting system to exposures at extremes of high and low temperatures, and to the shock of alternate exposures to these extremes. Physical damage produced during this test results principally from variations in dimensions and other physical properties.

Procedure

1. **Apparatus** - Separate chambers shall be used for the extreme temperature conditions. The air temperature of each chamber shall be held at the specified temperature by means of air circulation and sufficient heat or cold chamber thermal capacity. The ambient temperature of each chamber shall reach the specified temperature within five minutes after the test samples have been transferred to the appropriate chamber.
2. **Mounting** - The test samples shall be positioned within the chamber such that there is substantially no obstruction to the flow of air across and around the samples. Any additional mounting instructions shall be specified by the requesting design engineer.
3. **Length of Test** - Test samples shall be subjected to 49 thermal cycles for the Product Validation (PV) tests, unless otherwise specified by the design engineer.
4. **Test Temperatures** - Test samples shall be subjected to one of the following test conditions, unless otherwise specified by the design engineer:

Type of Connector	Test Temperature Cycle
High-temp Sealed	-40° ± 2° C (1 hour)/125° ± 2° C (3 hours)
Moderate-temp Sealed	-40° ± 2° C (1 hour)/120° ± 2° C (3 hours)
Unsealed	-40° ± 2° C (1 hour)/100° ± 2° C (3 hours)

Test samples shall be placed in a separate chamber at 55° ± 2°C, with 95-98% Relative Humidity for three hours at the start of the test and after every 7 cycles.

5. For submersible connectors, perform the Submersible Air leak test (Test 2e, Step 6 of this spec) after thermal cycling, in order to check for loss of seal function.

Requirements

The connector shall complete the test successfully. Evidence of deterioration resulting from this test can at times be determined by visual inspection (distortions, cracks, surface delaminations, etc.). However, the effects may be more readily ascertained by measurements or other tests prior to, during, or after thermal cycling. A failure is defined by one or more of the following conditions:

- A. Distortions that affect assembly fit and function (unacceptable change in part dimension).
- B. Loss of lock function.
- C. Visual evidence of cracks and surface delaminations considered harmful to connector function.
- D. Loss of seal function as defined in the Submersible Air leak test. (Test 2e of this spec)
- E. Other, as specified by the design engineer.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Test 2. Seal Performance

Procedure

The purpose of these tests are to establish the electrical characteristics and evaluate the seal performance for a particular electrical connector assembly when subjected to laboratory salt spray. The electrical characteristics to be established are insulation resistance and total terminal connection resistance. NOTE: There are four different tests for evaluating seal performance, the salt spray test, the soap shower test, the submersible air leak test, and the dielectric test. The design engineer shall specify on the engineering drawing which test is to be performed for each specific application.

Test 2a Salt Spray

Procedure

The salt spray (fog) test shall be performed per ASTM designation B117 latest revision with the following additions:

1. Preparation of Test Samples

- 1.1 Crimp to each terminal a convenient length of wire. Use the wire with the smallest outer diameter that is called out for the connector being tested.
- 1.2 Install the terminal/wire subassemblies into the specified connector hardshell test samples. For connectors with more than 10 circuits, instrument a minimum of 10 random locations for this test. These locations must be approved by the Ford design engineer prior to instrumentation of the test. Visually inspect the connector assemblies to make certain that all required packoff, doep, grommets or insert solder weather seals are properly installed. If specified, apply grease to the terminals. Type and amount of grease to be applied shall be defined by the requesting design engineer. Mate connectors.
- 1.3 Solder one millivolt drop lead to each connector assembly wire. All millivolt drop leads shall be single conductor, 28 gauge insulated wire. Install heat shrinkable tubing over each millivolt drop lead to wire attachment.
- 1.4 Secure the connector assemblies to a steel ground plane with a plastic locator or other means of attachment with 4 samples horizontal, 4 samples vertical, and 4 samples at a 45° angle, or as specified by the responsible design engineer.
- 1.5 Position the ground plane in a salt spray chamber per ASTM B117 paragraph 3.1.1 such that the test samples are directly exposed to the salt spray.
- 1.6 Except during electrical measurements, continuously apply 14 ± 0.5 VDC across adjacent terminals in each multicavity connector hardshell. For terminals installed in single cavity connector hardshells, apply 14 ± 0.5 VDC across the ground plane and terminal under test.

2. Electrical Measurements

- 2.1 Measure and record total terminal connection millivolt drop per sample (contact resistance) at the second hour, twenty-fourth hour, and every 36 hours thereafter, for each test sample. Test current shall be 5.0 ± 0.5 amp. Test current shall be "on" only for the time necessary to take all millivolt drop measurements. The power supply voltage shall not exceed 12.0 volts. Total terminal connection millivolt drop values are obtained by taking the total reading obtained above and subtracting the millivolt drop due to the wire between the test leads (point A. to point B. in figure 1). The contact resistance is calculated by dividing the total terminal connection millivolt drop by the test current.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

2. Electrical Measurements - (continued)

- 2.2 Measure and record insulation resistance (IR) at 100 VDC (unless otherwise specified) across adjacent terminals for each multicontact connector hardware. For terminals installed in single cavity connector hardware, measure IR at 100 VDC across the ground plane and terminal under test. During this test, open circuit the millivolt drop leads. Measure IR at the second hour, twenty-fourth hour, and every 24 hours thereafter, for each sample.
- 2.3 Prior to subjecting test samples to salt spray, measure and record the IR and total terminal connection resistance per paragraphs 2.1 and 2.2 for each test sample.

3. Length of Test

The connector samples shall be subjected to the salt spray test per ASTM designation B117 for a period of 96 hours.

4. Requirements

Evidence of deterioration resulting from this test can be determined by careful visual examination and review of the contact resistance and insulation resistance data. A connector assembly failure is defined by one or more of the following conditions:

- A. Total connection contact resistance data and/or insulation resistance data that exceeds the limits in the chart below:

Type of Connector	Contact Resistance		Insulation Resistance
	Before Test	After Test	
Open Housing	.01 ohms	.03 ohms	N/A
Weather Resistant	.002 ohms	.006 ohms	3K ohms
Submersible	.002 ohms	.003 ohms	20M ohms

- B. Harmful disintegration of terminal base metal. Tin or tin-plated surfaces may show a light gray oxide but underplate or base metal must not be visible with 10X magnification (excluding edges and sheared surfaces).
- C. Formation of cracks in terminal base metal that harmful affect mechanical characteristics (normal contact forces, etc.).
- D. Other, as specified by the design engineer.

Test 2b Soap Shower

A more severe test of water repellency is the soap shower test. The connectors are prepared in the same manner as in the salt spray test and the same electrical measurements are taken. However, the salt spray is replaced by a shower head located 200mm to 300mm above the samples, with a flow rate of 30 gallons per hour. The shower shall spray the samples with 3% salt-free dish soap solution heated to 40°C, with a source pressure of 15 psi (gauge). Subject the samples to the spray for 2 seconds of a 4 second cycle for 30 cycles (total test time, 30 minutes). The requirements are the same as those for the salt spray test. Take all electrical measurements without making any special efforts to remove the residual water.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Test 2c Submersible Air Leak

purpose

The purpose of this test is to assure that all submersible connectors are free from leaks which may allow water vapor or other harmful substances to enter the connector and corrode the terminals. For potted connectors, such as the 12-way, this test is also used to demonstrate the effectiveness of the potting material.

Procedure

1. Assemble the connector shell, grammets, and terminals. The terminals must be sealed to prevent air from leaking out through the wires. This can be done in a number of ways, such as potting the ends of the wires or double crimping the wires and using the terminals on both ends of the wire in the connector assembly. For direct connectors, the mating components must be specially prepared. The terminal tails should be cut off at the terminal barrel and then potted. The terminal tails should not extend through the potting material. Connectors that rely on potting material to seal the terminals by design shall be tested without modification. Note the connectors. The process for preparing the connector assembly for this test shall be approved by the responsible design engineer before testing.
2. Remove one of the terminals from either end of the connector assembly, and insert a 3.2 mm O.D. tube through the grammets as in Figure 1.
3. Connect the other end of the tube to a pressure tester, as shown in Figures 3 and 4. Immersa the connector assembly in water. Start with the air regulator closed and slowly open it until the manometer reads 50 Kpa (7 psi).
4. Keep the connector submerged for at least 15 seconds while examining it for leaks. If the mating component of a direct connector leaks, it shall be repeated and the test repeated.
5. For the product verification (PV) test only, place the connector assemblies in a temperature chamber and heat age the connectors for 70 to 74 hours at $15^{\circ} \pm 3^{\circ}\text{C}$.
6. Remove the assemblies from the thermal chambers and repeat steps 3 through 4, using a pressure of 50 Kpa (7 psi).

Requirements

The snap fit and bolted connectors must both be able to hold 50 Kpa (7 psi) for a minimum of 15 seconds.

Test 2d Dielectric

purpose

This test is intended to confirm that all submersible connectors are truly submersible and retain their dielectric properties while submerged in water.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

IMMERSIBLE TEST

Procedure

A. Product Validation Test

For connectors with 20 positions or fewer, 24 samples of each assembly part number to be tested are required. For connectors with more than 20 positions, a minimum of 8 samples (or 1 per tool cavity, whichever is greater) of each assembly part number to be tested are required. Each sample must be mated to its appropriate mating connector assembly thus forming a complete electrical connection including all applicable grounds, seals, wedges, and other parts. Each terminal in each connector shall be appropriately attached to a minimum one-half meter length of wire conforming to MIL-MILS-3 or MIL-MIL-23-A (to be specified by the responsible design engineer), except in the case of terminals which are part of a direct electrical plug-in component assembly and thus do not attach to vehicle wiring. Half of the test connections should use the largest wire sizes, the remaining half should use the smallest wire sizes.

Divide the samples equally into four groups A-D. Groups A and B should include samples using the largest wire sizes, groups C and D should include samples using the smallest wire sizes.

The test procedure is as follows:

1. For each sample in groups A and C, apply 300 volts D.C. across each pair of adjacent wires using a Middle Dielectric Test set, catalog 51005, or equivalent. Measure the shunt resistance, then remove the 300 volt potential.
2. Submerge the samples in groups A-D to a depth of 20 to 48 mm. in a 20° ± 5°C water solution containing 3% salt by volume and 1% dishwashing soap by volume. The exposed ends of the wires must be protected from contamination by the solution and its vapor. When direct plug-in electrical components form part of the electrical connections being tested, any openings in the components that would allow the solution to penetrate to the interior of the connections must be potted. Complete step 3 within 15 minutes of submerging the samples in the solution.
3. For each sample in groups A and C, repeat step 1. For each sample in groups B and D, apply 300 volts D.C. across the solution and every wire in the connection using a Middle Dielectric Test set, catalog 51005, or equivalent. Measure the shunt resistance, then remove the 300 volt potential.
4. Leave the samples submerged in the solution for 4 hours. At the end of the submersion period, while the samples are still submerged, repeat step 3, then remove the samples from the solution.
5. Place the samples in a temperature chamber preheated to 135°C. Heat soak the samples for 4 hours.
6. Immediately transfer the samples to a temperature chamber preheated to -40°C. Cold soak the samples for 1 hour.
7. Repeat the test sequence given in steps 3 and 4 three more times.
8. Remove the samples from the temperature chamber and let them stabilize at room temperature for 1 hour. Then repeat the test sequence given in steps 2 through 4.
9. Place the samples in a temperature chamber preheated to 135°C. Heat soak the samples for 4 hours.
10. Without allowing the samples to stabilize at room temperature, remove them from the 135°C temperature chamber and immediately plunge them into the solution. Repeat steps 2 through 4.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Requirements

The shunt resistance shall not fall below 20 megohms at any time during the entire procedure.

Process Control Parameters - shunt resistance.

Test 3 Vibration

Purpose

The purpose of this test is to determine the effects on component parts of vibration within the predominant frequency ranges and magnitudes that may be encountered during field service. Most vibration encountered in field service is not of harmonic nature, but test based on vibrations of this type have proved satisfactory for determining critical frequencies, modes of vibration and other data necessary for planning protective steps against the effects of undue vibration. Vibration caused by loosening of parts or relative motion between parts in the specimen can produce objectionable operating characteristics, noise, wear, and physical distortion, and often results in fatigue and failure of mechanical parts.

Procedure

1. Mounting

- 1.1 The mounting apparatus used to mount the samples on the vibration machine must be free from resonances over the test frequency range, for example as shown in Figure 1. For test samples with attached brackets, one of the vibration directions shall be parallel to the mounting surface of the bracket. For connectors mounted directly to an electronic component, one of the vibration directions shall be perpendicular to the circuit board that the connector mates to. Vibration input shall be monitored on the mounting fixture in the proximity of the support points of the test sample.
- 1.2 All connectors shall be wired in series and connected to a DC power source allowing a current flow of 100 mA to monitor during the entire test phase for electrical discontinuity greater than 1 microsecond.
- 1.3 The test apparatus must be approved by the responsible design engineer prior to the start of the test.

2. Amplitude and Frequency

- 2.1 All connectors shall be subjected to a simple, harmonic motion having a displacement of 0.06 inches peak-to-peak. The frequency shall be varied logarithmically between the approximate limits of 10 and 35 Hz.
- 2.2 All submersible connectors shall also be subject to a simple harmonic motion of 25 g's while varying the frequency logarithmically between 40 and 300 Hz. This test is to be performed in addition to step 2.1.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

1. Time and Duration

- 3.1 The entire frequency range in step 2.1 of 10 to 25 Hz and back to 10 Hz shall be traversed in 1 minute. The motion shall be applied for a period of 16 hours in each of the three mutually perpendicular directions (total test time 48 hours).
- 3.2 The entire frequency range in step 2.2 of 40 to 300 Hz and back to 40 Hz shall be traversed in 10 \pm 2 minutes. The motion shall be applied for a period of 8 hours in each of the three mutually perpendicular directions (total test time 24 hours).

Requirements

The connector shall complete the test successfully. Evidence of deterioration resulting from this test can at times be determined by visual inspection (distortion, cracks, surface delaminations, etc.). However, the effects may be more readily ascertained by measurements or other tests prior to, during, or after vibration testing. A failure is defined by one or more of the following conditions:

- A. Distortions that affects assembly fit and function (unacceptable change in part dimension).
- B. Loss of lock function.
- C. Visual evidence of cracks and surface delaminations considered harmful to connector function.
- D. Loss of electrical continuity for greater than 1 microsecond.
- E. Other, as specified by the design engineer.

Test 4 Mechanical Shock

Procedure

This test is for the EEC connector only, to ensure that the connector assembly and mounting hardware can withstand the vehicle shock that normally occurs during driving.

Procedure

Assemble the connector shell, grommets, spacers, terminals, locator, and any other mounting hardware. Solder a blank circuit board to the terminal tails if applicable. Mount the assembly to a fixture using the method of attachment provided by the part. Subject the unit to 3 impulses in a direction parallel to the mounting plane, equivalent to 100 Gs force, for a minimum duration of 10 milliseconds each. Apply 3 more impulses of equal strength and duration in the opposite direction. Repeat the procedure in each of the two mutually perpendicular planes.

Requirements

- The connector assembly must remain attached to the fixture for the entire procedure. No visual evidence of cracks or surface delaminations considered harmful to connector function are permissible.

Test 5 Terminal Retention

Procedure

This test ensures that the terminals will not pull out of their cavities when the wire is pulled. This is a required feature of any connector in order to avoid terminal pushouts and intermittent electrical contact during connector assembly and repair.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Procedure

1. A polished steel gauge should be used to test terminal retention. The gauge shall be constructed to nominal terminal dimensions, except for the locking features, which should be worst case. The terminal gauge used for this test must be approved by Ford engineering prior to testing.
2. Pull on the gauge, to make sure that it is securely locked inside the connector and then insert spacer (if applicable).
3. A digital force gauge, approved by Ford SQA, shall be used for all terminal pull testing. This gauge must be properly sized for the force to be inspected. Attach the connector to the gauge and mount assembly to an engagement machine. The engagement machine shall be a motor-driven screw type pull tester with double, vertical columns. Limit switches are recommended to speed the repetitive testing of connectors using gauges to simulate terminals. Use a 0 - 222.4N (0 - 50 lbs.) force gauge for all force measurements, unless the pull forces are expected to exceed 222.4N (50 lbs.), in which case a 0 - 444.8N (0 - 100 lbs.) force gauge shall be used.
4. Run engagement machine with a rate of 25 mm./min. to 100 mm./min. (1-4 inches/min.) on the connector until the terminal comes out of the connector shell. Record retention forces and calculate the three sigma limits for each terminal cavity. (See section IV - STATISTICAL ANALYSIS METHODS)

Requirements

- (E) Terminals designed to lock into premolded connector shells shall withstand a pull test of 80 N (18 lbs.). (lower 3 sigma limit)

Process Control Parameter - Pull Force

Test 6 Housing Retention and Lock Fatigue

Procedure

The purpose of this test is to assure that the connector (pair or direct connect) remains mated after it is subjected to wire tension and vibrations. It also serves to test the strength of the locking arms after multiple connect/disconnect cycles.

Procedure

1. Place one of the connector halves (or the module in the case of a direct connect) in a Vining Olsen, Instron, or equivalent test machine.
Note: If a vise is used, do not apply excessive force to hold the connector in place. Excessive vice force will create incorrect readings. A machine designed for force testing, as stated above, is the preferred method.
2. Mate the connector halves, with terminals, on the test machine. Using a digital force gauge, disconnect the connector slowly (5 cm./min. ± 1 cm./min.) and record the peak force necessary to fully disconnect the connector. Repeat this procedure 10 times for pull apart connectors only. In order to test the fatigue of the locking arms.
3. For all Go-No-Go connectors, the mating force must also be measured. First, take a connector shell with no terminals and mate the connector halves slowly (5 cm./min. ± 1 cm./min.). Using a digital force gauge, measure the force required to fully mate the connector halves. Next, break off the locking arms (or render them inoperative) and insert the full contingent of terminals into each connector shell. Again mate the connectors at the same speed and record the force necessary to fully mate the connector halves. Repeat this procedure 20 times. Calculate the three sigma limits for the first and last mating, for both the connector shell and the terminals.

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TM PD 3947-02 (Previous editions may NOT be used)

Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Requirements

Go-No-Go Lamp Connectors

Mating Force: The lower 3 sigma limit for the connector mating force must be greater than the maximum combined mating force of the terminals. In designs with a gasket or lip seal, the lower 3 sigma limit for the connector mating force must be greater than the sum of the mating force of the terminals plus the force to compress the gasket or overcome the resistance of the lip seal.

Pull Apart Connectors

Disconnect Force: 178N (20 lbs.) minimum initially
67N (15 lbs.) minimum thereafter

Positive Lock Connectors

Disconnect force: 154N (35 lbs.) minimum

Non-Locking Connector Shells

Disconnect force: 27 - 33N (6 - 12 lbs.) per terminal (detented)

9 - 27N (2 - 6 lbs.) per terminal (non-detented)

Connectors with Any Positive Locking Terminals

Disconnect force: 178N (20 lbs.) minimum - 1 to 3 terminals

196N (35 lbs.) minimum - 4 or more terminals

Test 7 Push Out - Push Out and Turning Force

Purpose

This test is for all bolted connectors and ensures that the bolt will withstand the shocks incurred during shipping and assembly without coming loose from the connector shell.

Procedure

- A. Push out - Assemble the bolt and retainer to the appropriate connector shells on production equipment. Mount the connector half on a steel fixture (for example, as shown in figure 3) and use a Timins Olsen machine to push out the bolt. Record the minimum force required to push the retainer off the shaft.
- B. Turning - Assemble the bolt and retainer to the appropriate connector shells on production equipment. Mount the connector half on a steel fixture (for example, as shown in figure 3) and use the Timins Olsen, Ingretex, or equivalent machine to apply a load of 443N ± 45N (100 ± 10 lbs.) for a 6 mm. bolt or 2.22KN ± 45N (500 ± 10 lbs.) for a 6 mm. bolt to the thread end of the bolt. Slowly turn the bolt in a counter clockwise direction for 10 revolutions.

Requirements

- A. The retainer is required to withstand a pushout force of 2.67KN (600 lbs.) for a 6 mm. bolt and 896N (200 lbs.) for a 4 mm. bolt.
- B. The 4 mm. bolt is required to remain within ± 0.5 mm (.02 inch) of its original location. The 4 mm. retainer must remain in the undercut of the bolt.

Process Control Parameter - Pushout Force

Test 8 Threaded Insert Retention

Purpose

This test is for all bolted connectors and ensures that the threaded insert will withstand the shocks incurred during shipping and assembly without coming loose from the connector shell.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Procedure

- A. Push Out - Assemble the threaded insert into the appropriate connector shell and screw in a bolt, engaging all the threads in the insert. Place the assembly into a steel fixture capable of securely holding the connector shell, while allowing room for the threaded insert to push out (for example as shown in figure 6). Using a Tinius Olsen machine, apply a load on top of the bolt head to determine the minimum force required to push the insert out of the connector.
- B. Push Through - Assemble the threaded insert into the appropriate connector shell and thread a bolt into the threaded insert from the shoulder end of the connector until all the threads are engaged. Place the assembly on a steel fixture capable of securely holding the connector shell, while allowing room for the threaded insert to push through the shell (for example as shown in figure 7). Apply a load on the top of the bolt head to determine the minimum force required to push the shoulder of the insert through the connector shell.
- C. Turning Torque - Modify a production bolt as shown in figure 8, adding a square shoulder at the end of the threads. Hand thread this bolt into the threaded insert until the bolt shoulder contacts the brass bushing. Place this assembly on a steel fixture as in step A, and measure the minimum torque required to turn the brass insert one revolution.

Requirements

The threaded insert must withstand a minimum force depending on the bolt size:

Test Step	1 mm. Bolt	4 mm. Bolt
A.	3.56kN (800 lbs.)	1.78kN (400 lbs.)
B.	7.12kN (1600 lbs.)	4.45kN (1000 lbs.)
C.	19.74 Nm (175 in-lbs.)	3.68 Nm (50 in-lbs.)

Procedure Control Parameter - A. and B. minimum force, C. minimum torque

Test 9 Over Torque Capability for Male/Female Assembly

Procedure

Automated bolt and screw drivers are subject to over torquing. This test ensures that the connector assembly will resist over torque.

Procedure

- A. Take a male and a female connector assembly (complete with terminals, spacers, grommets, etc.) and mate them together. Secure the connector assembly in the fixture used for test 3-A (figure 6) or a bench vise. For a 1 mm. bolt, use a certified CM-1 or equivalent torque wrench and torque the bolt to 23 in-lbs (2.63 Nm). For a 4 mm. bolt, use a certified CM-4 or equivalent torque wrench and torque the bolt to 60 in-lbs (6.77 Nm). Disassemble the connectors using the same torque wrench and visually inspect both parts for damage. Inspect for buckled towers, broken or dislodged push nut retainers, movement of the brass insert, and any other damage that could affect the function of either connector.
- B. If, after step A, there is no visible damage, mate the connectors again and torque them down as in step A. Then, continue to torque the bolts slowly until the connector cracks or until 100 in-lbs (11.3 Nm) is reached. Record the torque value at which buckling or cracking occurs if less than 100 in-lbs. (11.3Nm).

Requirements

- A. A connector using a 1 mm. bolt must withstand a torque of 23 in-lbs. (2.63Nm) minimum and a connector using a 4 mm. bolt must withstand a torque of 60 in-lbs. (6.77Nm) minimum. No visual damage is allowed beyond a 0.17 mm. movement of the brass bushing and a slight buckling of the tower (less than 3% of the tower diameter).

Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Requirements - cont.

- B. The data sheets shall contain the torque required to buckle or crack the connector, if less than 100 in-lbs. (11.3Nm), regardless of whether the connector meets the requirements or not.

Process Control Parameter - Minimum torque to buckle or crack connector

Test 10 Assembly Torque Required

Purpose

This test is intended to show that the mating halves of a bolted connector can be easily assembled in production.

Procedure

Take a male and a female connector assembly (complete with terminals, spacers, grommets, etc.) and secure one half of the connector assembly in the fixture used for test #A (Figure 4) or a bench vice. Use a torque wrench to assemble the mating halves and record the maximum torque required to fully mate the connector halves.

Requirements

Connector Description	Drawing Number	Range of Assembly Torque
15-way	F27-14A484-V6	1.9 ± 0.1
HCU-14CH 24-way	F1A-14A484-AB	1.7 ± 0.1
Bulkhead 24-way	F2B-14A484-AB	1.8 ± 0.1
AES 36-way	F2D-14A484-L4	4.0 ± 0.5
AEI-111 32-way	F2A-14A484-2	4.1 ± 0.5
36-way	F2D-14A484-04	4.0 ± 0.5
AES 48-way	F17-14A484-06	6.2 ± 1.0
In-Line 48-way	F1V-14A484-74	4.0 ± 0.5
Bulkhead 32-way	F2A-14A484-AB	4.0 ± 0.5
Door 36-way	F2D-14A484-06	3.7 ± 0.5
24-way	F2D-14A484-06	3.0 ± 1.0
HSC-IV 48-way (sealed)	F41-14A484-01	3.3 ± 0.5
HSC-IV 48-way (unsealed)	F41-14A484-04	3.7 ± 0.5
48-way	F2D-14A484-06	3.0 ± 1.0
Bulkhead 76-way	F2U-14A484-06	4.0 ± 0.5
48-way	F2D-14A484-06	3.0 ± 1.0
HSC 104-way	F43-14A484-01	4.1 ± 1.0
120-way (11 48-ways)	F11-14A484-01	4.1 ± 1.0

Other connectors shall have the assembly torque requirements stated on the assembly drawing.

Process Control Parameter - Minimum assembly torque

Test 11 Solder Tail Gauge Test

Purpose

This test is for all connectors that mate directly to a circuit board and ensures that the solder pins will always mate to that board, even with worst case conditions.

Procedure

The solder tails are to be installed on the appropriate connector half and then gauged in a true position gauge which has been reviewed by the responsible Ford Motor Co. design engineer. This true position gauge is to be designed so that with worst case conditions, the connector will always mate with the printed circuit board.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Requirements

If any of the solder pins will not mate with the gauge, reject the lot and sent for the identified nonconformance. All nonconforming connectors must be immediately removed from production approved stock and clearly marked for rework or scrap. All stock that has been reworked must pass the gauge test. Since the gauge test must be completed before potting to allow rework, a random sample of connectors with potting must be gauged (1 per tray of 60 pieces). If any of these connectors fail the gauge test, they are to be immediately quarantined for scrap and all remaining pieces in the tray are to be checked.

Test 12 Grommet Assurance Test

All connectors with grommets must be inspected for the presence of the grommet and the correct position of the blanked holes in the grommet relative to the connector hardware. This test is to be performed at the site where the grommets are installed in the connector shells.

Test 13 Melt Flow Index

Purpose

This test is to be performed by the part fabricator or molder who molds polyester plastic parts. It is basically necessary in order to confirm raw material properties and insure that excessive material degradation does not occur during the molding process.

Procedure

Five samples of the raw material each of sufficient size for testing the material to ASTM D1512 condition ~~as~~ except as noted below for Gafite 6448, using a Monsanto or Timex Clegg Indenter or equivalent test equipment shall be selected and tested for Production Validation.

When reground is used, the molder shall certify that the combination of reground and raw material still meets the Ford melt flow index requirements for the raw material.

For IP testing, one sample of raw material from each lot of material received shall be tested as specified for Production Validation above. In addition, one sample lot of a sufficient number of finished parts shall be selected and ground to provide material for testing as specified for Production Validation above. The values obtained for the reground parts shall be compared to the values obtained for the lot of raw material from which the parts were molded.

The testing of the material from finished parts shall be performed as follows:

- 1) At the start of each production run (samples taken during the first 30 minutes).
- 2) Each time the molding machine is started up following a shutdown of 4 hours or more.
- 3) Once per 24 hours.
- 4) At the end of the production run (samples taken during the last 30 minutes).

Material conditioning (drying, etc.) prior to the test is to be in accordance with the recommendations of the raw material supplier.

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III. TEST PROCEDURES AND REQUIREMENTS - continued

Requirements

Material	Melt Index or Flow Rate (grams/10 min.)	
	Raw Material	Required of Finished Parts
ESB-H4234-A2 (GAF GAFITE 14422)	8 Max.	24 times the value obtained for raw material (max.)
(Calanex 4300)	10 Max.	2 times the value obtained for raw material (max.)
ESB-H4D27-A1 (GAF GAFITE 14228)	8 Max. ³	2 times the value obtained for raw material (max.)
G. E. Valox 420P ⁴	10 Max.	14 Max.
ESB-H4D334-A1 G. E. Valox 420	20 Max.	A maximum of 2 times the value obtained for the raw material but not over 21
ESB-H4D462-A3 G. E. Valox 308	13 Max.	A maximum of 2 times the value obtained for the raw material.
GAF GAFITE 14322 ¹	8 Max.	2 times the value obtained for raw material (max.)
ESB-H4D334-A3 (G. E. Valox 420-420)	23 Max.	A maximum of 2 times the value obtained for the raw material but not over 21
ESB-H4D467-A3 G. E. Valox 4301 ¹	12.4 Max.	2 times the value obtained for raw material (max.)
ESB-H4D341-A3 (Calanex 3300) - Natural/Grey - Black	20 Max. 23 Max.	2 times the value obtained for raw material (max.)
ESB-H4D462-A1 G. E. Valox 4240	23 Max. ³	2 times the value obtained for raw material (max.)

- 1 Ford material spec to be written and Ford # to be assigned.
- 2 For this test a 5000 gram load is to be used instead of the normal condition "T" load of 2100 grams.
- 3 For this test a 5000 gram load and temperature of 510° F. is to be used. All other conditions per the ASTM-D138 condition "T" should be followed.

Test 14 Detent Force

Equipment

This test is for all connectors that incorporate a release feature to unlock the connector shell for servicing. It ensures that the connector shells can be removed easily after the lock mechanism is released.

Procedure

1. Place the female connector shell (or a module, with the male terminals removed, if the case of a direct connect) in a fixture. The shell should contain all of the grommets, o-rings and seals but no terminals. If necessary, the terminals should be cut off so that they will have no effect on the measuring force.
- Note:** Do not apply excessive force to hold the connector in place. Excessive vice force will create incorrect readings.
2. Load the male connector with a couple of terminals, crimped to a convenient length of wire for pulling, and a wedge, if applicable.
 3. Mate the connectors fully, then release the locking mechanism. Pull the connector halves apart just enough to keep the lock from reengaging.

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Engineering Specification

III. TEST PROCEDURES AND REQUIREMENTS - continued

Test 14 Unlatching Force - continued

4. Using a digital force gauge, measure the force necessary to fully unlatch the connectors.

Requirements

All connectors with a front push-to-release lock beam feature (with a ~~low-profile~~ lock ramp) shall have an unlatching force of 15 pounds maximum. Connectors with a rear push-to-release lock beam feature (with a low profile lock ramp) shall have an unlatching force of 7 pounds maximum.

Test 15 Terminal Insertion

Purpose

This test is run to ensure that the insertion force required to insert the terminals into the connector cavity is not greater than the column strength of the wire and is also low enough to allow easy and consistent assembly.

Procedure

1. Secure the connector shell in a fixture. If the connector being tested uses a grommet, assemble the grommet and load the connector with terminals, crimped to a convenient length of the correct wire, leaving only one cavity open. This is to compress the grommet to its worst case condition.
2. Crimp terminals on the smallest gauge wire applicable. Grip the wire 1 cm. from the back of the terminals (.12 in. on connectors that require grommets), or as specified by the design engineer. Insert the terminal slowly into the connector shell (23 mm. \pm 10 mm./min.).
3. Attach a terminal to a solid rod (or use a terminal gauge) and insert the terminal slowly (23 mm. \pm 10 mm./min.) into a new connector shell. Using a digital force gauge, measure the force required to fully insert the terminal into the connector shell.
4. Repeat steps 3 through 3 for each terminal cavity in the connector. The samples used for steps 1 and 3 shall include a minimum of one part from every connector cavity.

Requirements

The terminal should enter the connector shell smoothly, without hanging up or causing the wire to buckle during insertion. The force required to insert the terminal fully into the connector shell must not be greater than 4 lbs. (17.8N).

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IV. STATISTICAL ANALYSIS METHODS

The following are methods for reducing test data to determine compliance with the Reliability Acceptance Criterion shown in the Test Schedule, Section II.B.

A. Limit Graph

The Reliability Acceptance Criterion (α_{RL} = confidence level, β_{RL} = Reliability) is met if all sample characteristics are within the applicable rating limits.

B. Standard Deviation

1. The standard deviation (sigma) is the square root of the variance. The variance is the sum of the square of the difference between the individual measurements and the average divided by the number of measurements minus 1. The standard deviation is expressed mathematically as:

$$\text{Standard Deviation} = \sqrt{\frac{(X - \bar{X})^2}{N-1}}$$

Where: X = Individual measurement

N = Number of measurements

= Sum

and average = $\bar{X} = \frac{X}{N}$

C. Three Sigma Limits

The three sigma limits are the statistical high and low limits for a set of data. They are calculated by multiplying the standard deviation (sigma) by three and adding or subtracting that from the average, or mean value. For example, to find the lower three sigma limit of this set of 9 numbers:

(30, 32, 33, 30, 31)

calculate \bar{X} (average) = $\frac{X}{N} = 31.4$

calculate the standard deviation = $\sqrt{\frac{(X - \bar{X})^2}{N-1}} = 2.07$

three sigma value = standard deviation $\times 3 = 6.21$

lower three sigma limit = mean(average) - three sigma value = 25.39

upper three sigma limit = mean(average) + three sigma value = 37.61

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AM PD 3947-a2 (Previous version may not be used)

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V. VALIDATION REQUIREMENTS

For any change in processing or material the entire Production Validation (PV) Program must be successfully completed. No change to processing or material may be allowed without prior engineering approval and successful completion of the PV program. For certain process changes, portions of the PV Program may be waived if approval is obtained from the responsible design engineer, SMCX, prior to re-validation.

VI. LOT DEFINITION

A lot shall have been manufactured in a continuous 8 hour time period, on a validated production line, with unchanged process. Lot identification shall be established prior to performing tests. A lot must not overlap a production shift change.

VII. RECORD AND TEST SAMPLE RETENTION

- A. Samples used to demonstrate compliance with the "In-Process Tests" must be retained by the supplier for at least one month and may not be included in production shipments. Samples subjected only to tests performed on 100% of production stock need not be retained as described above.
- B. Data from "In-Process Test" are to be retained for 1 year and be made available to Ford Motor Company representatives upon request.
- C. In the event of test failure, appropriate action as outlined in Ford Quality Control Specification Q-101 must be instituted.

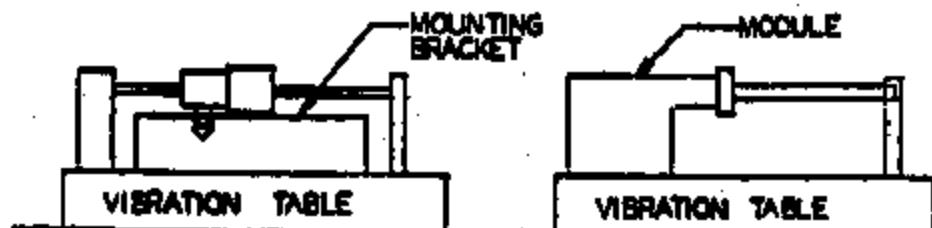
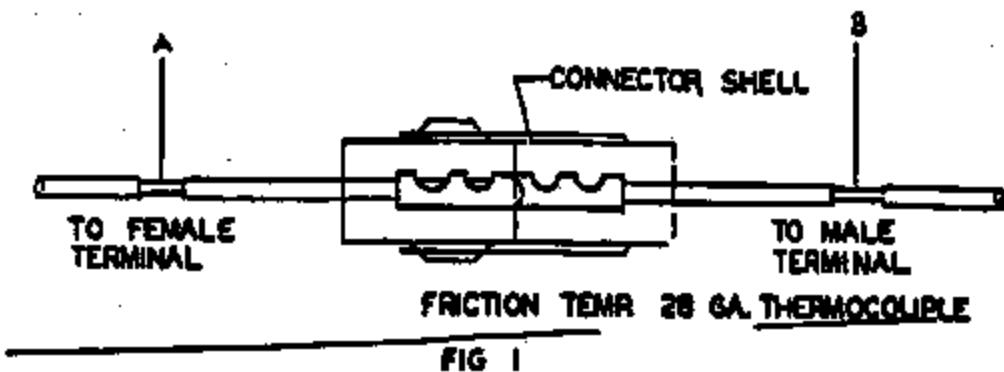
VIII. INSPECTION AND REVIEW

- A. The term "initial production" parts shall mean units which have production level components which have been produced by the production tools and assembled in a sequence and technique equivalent to that used in final production.
- B. Test fixture design, installation plans and operations procedures require review by the responsible SMCX Design Engineer. Pictures made from approved drawings also require review by the responsible SMCX Design Engineer prior to SMCX approval.
- C. For Production Validation, samples are to be randomly selected from a minimum lot size of 300 pieces.

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1000 PD 3947-82 (Previous editions now NOS or used)

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PD 3947-62 Previous editions may be used.

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Engineering Specification

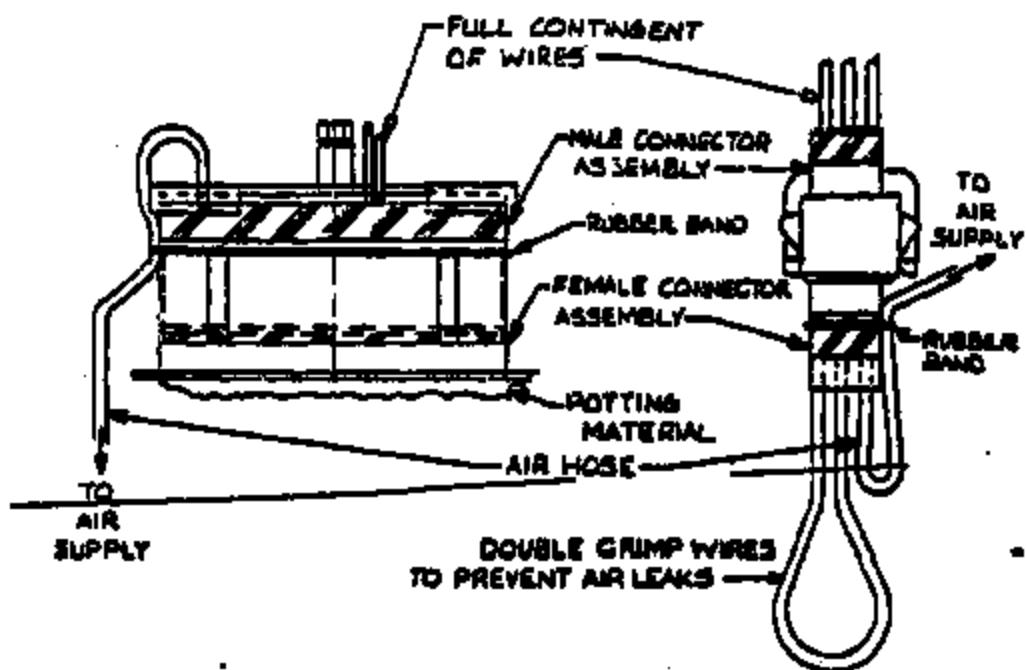


FIGURE 3



FIGURE 4

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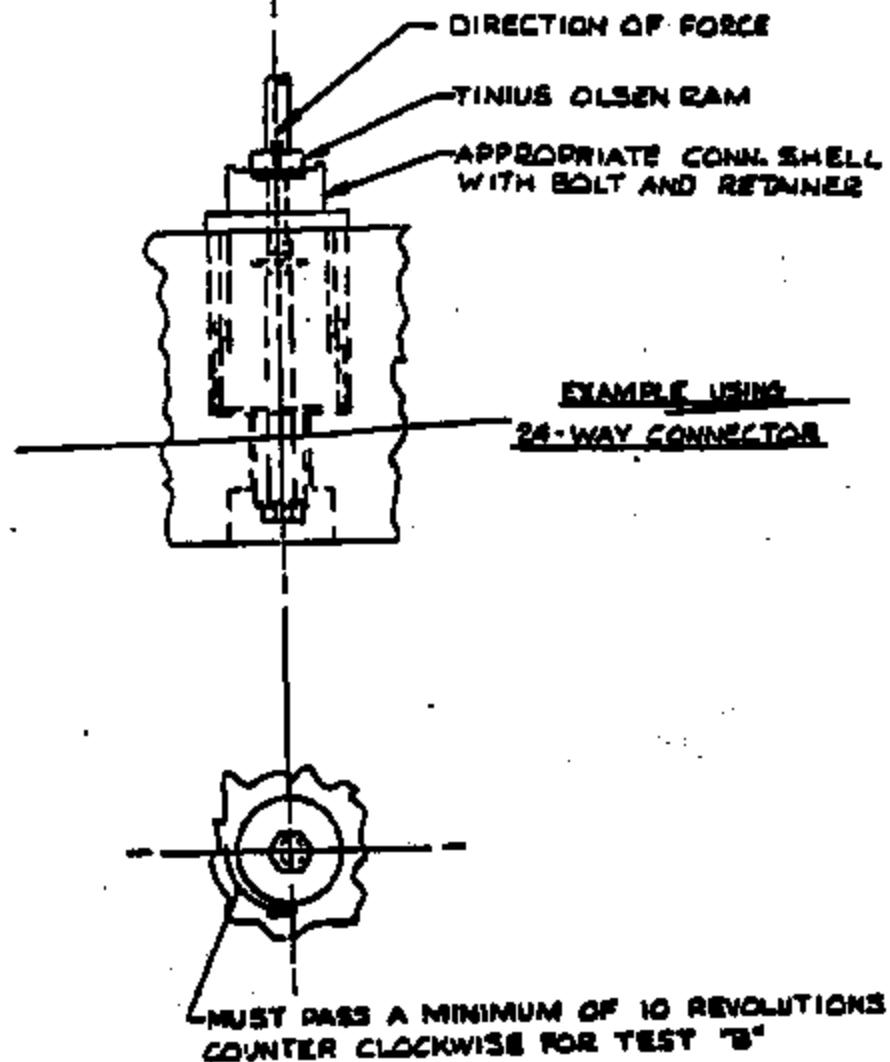
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TMG PD 3947-02 Previous editions will not be used

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FIGURE 5



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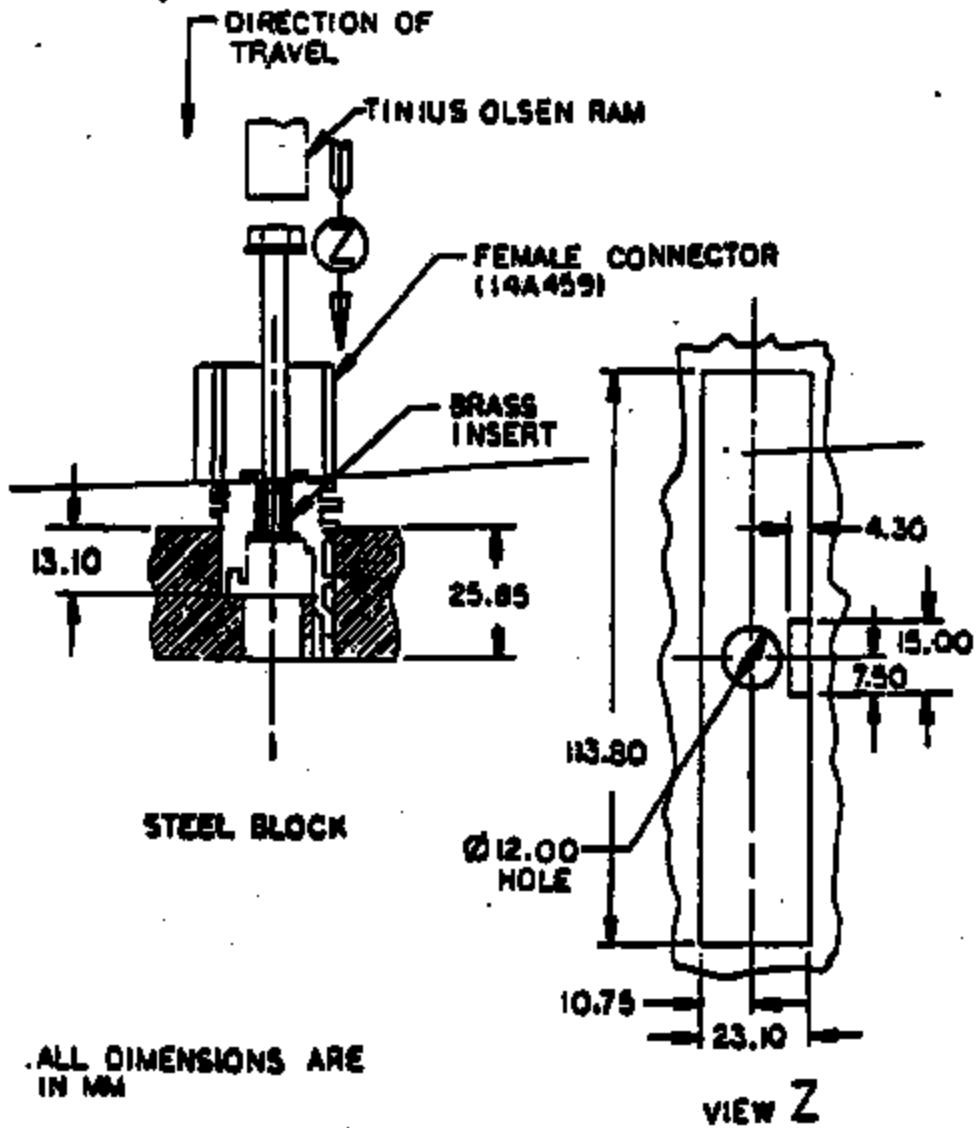
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AM PD 3947-82 Previous editions may not be used

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FIGURE 6
(FOR TEST 'A')
EXAMPLE USING 60 WAY EEC CONNECTOR



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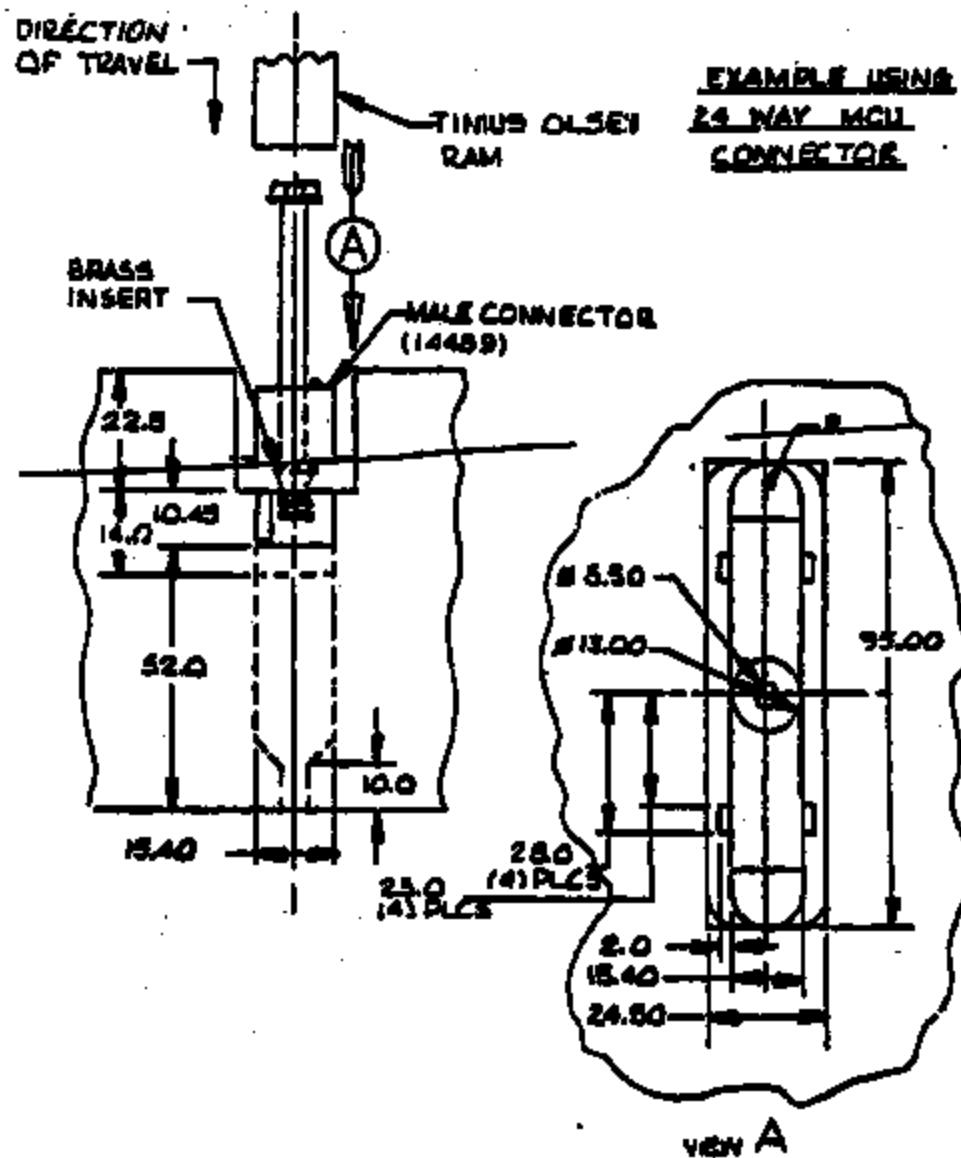
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TM PD 3947-22 Previous editions may not be used

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Engineering Specification

**FIGURE 7
(FOR TEST "B")**



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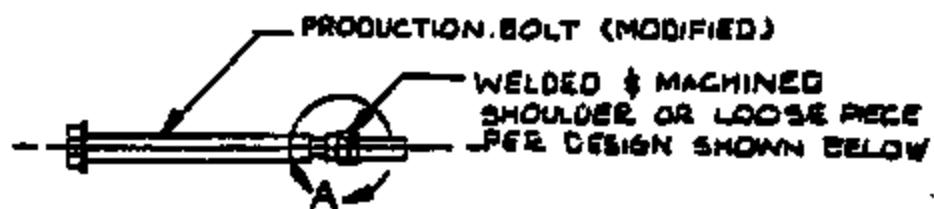
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TM PD 3847-02 Previous editions rev'd are invalid

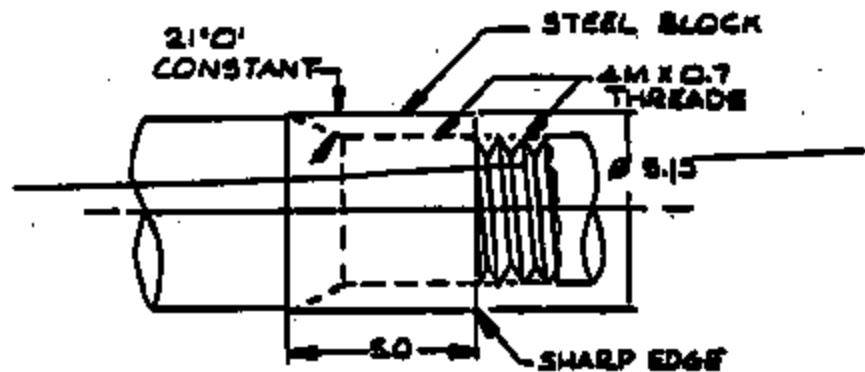
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FIGURE 8
(FOR TEST "C")

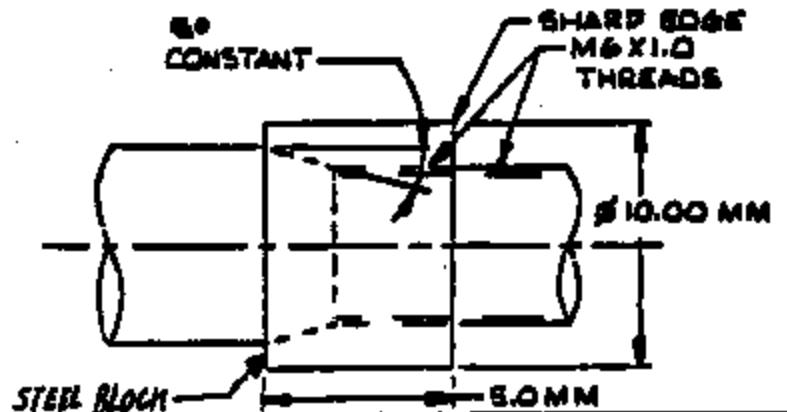


Fixture Design - 4 MM Bolt



VIEW A

Fixture Design - 4 MM Bolt

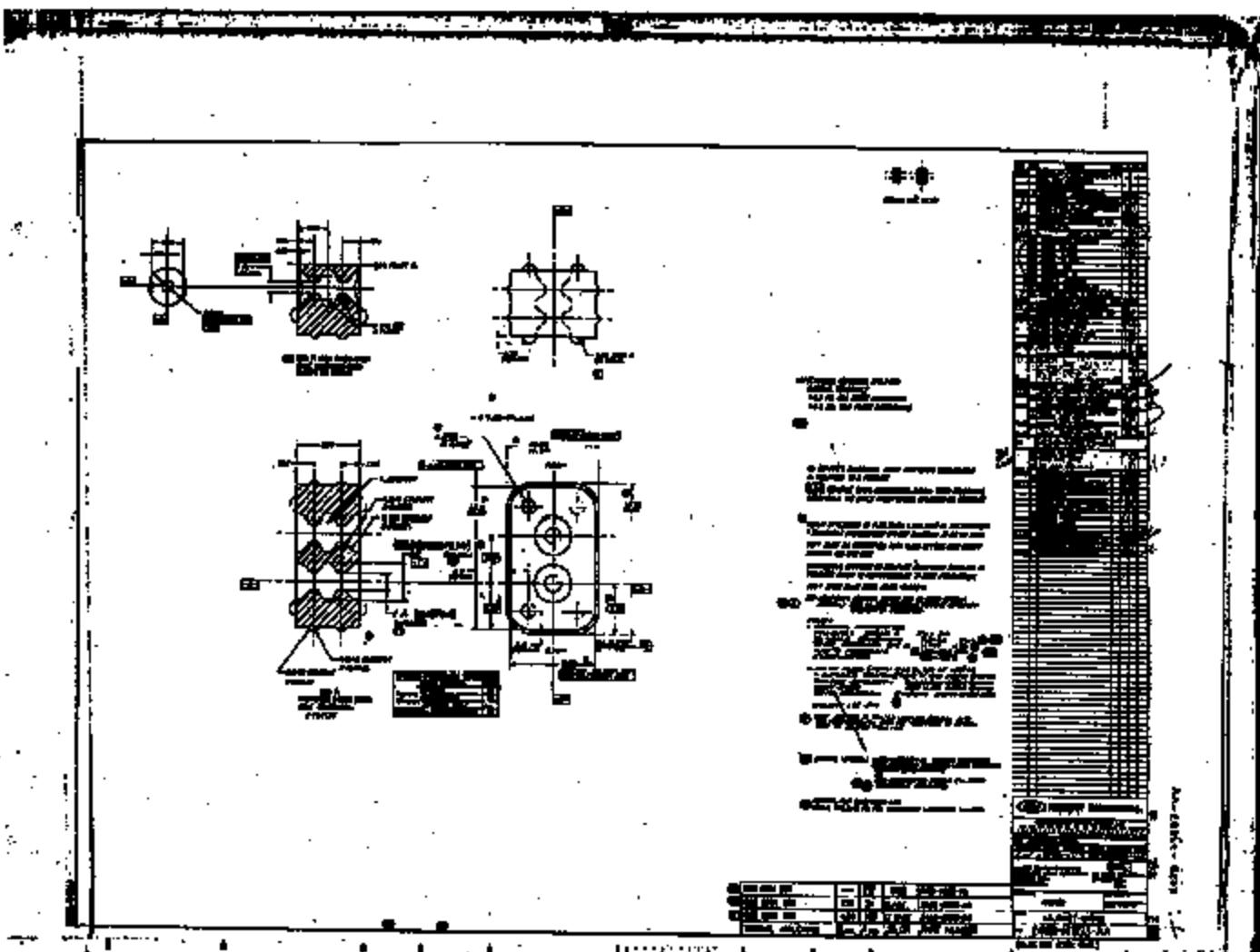


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TAB PD 3947-22 (Previous editions may not be used)

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Legal Request

Drawing Work Request

02/05/1998

Design Analysis

To: MSX International
1464 John Papas Dr.
Lincoln Park, MI 48146-1468
Attn: Dain Pearson ph: x89938, Teresa Platt ph: x58985
Fax: x57776

From: Design Analysis
408 Parklane Towers West (PTW)
Three Parklane Blvd.
Dearborn, MI 48126
Fax: 313-51811 Phone: 59-42897

PLEASE MAKE 1 XEROX COPIES AND 0 SET(S) OF APERTURE CARDS OF THE BELOW LISTED PART NUMBERS.

Work Request #:	191278	Due Date:	02/08/1998	Matter:	412052 RAMIREZ ROBERTO
DAB Manager:		Phone:		Allegation:	Lane Deployment
Contact:	RORSELLI	Phone:		Vehicle Model:	Lincoln Town Car (Full Size)
Requested by:	NLAPOINT	Phone:	(313)594-2686	Year:	1992

Prefix	Base	Suffix	Comments	Entry Order #
ESP33B	14A464	AA	CONNECTOR ES	1

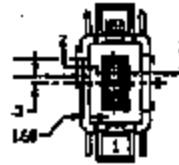
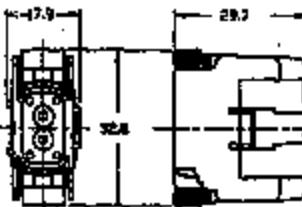


FIG. 2

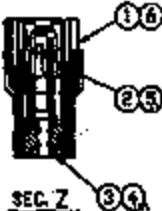
ALY SAME AS FIG. 1, EXCEPT 356
THIS SLOT IS BYSET AS SHOWN



WONDER DIGHT THIS AREA

ASSEY. NO. THIS AREA
CAVITY NO. THIS AREA

FIG. 1



SEC. Z ③ ④

ITEM	DESCRIPTION	REMARKS
		SEE SPECIFICATION FOR DETAILS
1	CONNECTOR-HOLE	BLACK
2	CONNECTOR-HOLE	GRAY
3	CONNECTOR	RED
4	CAV. (CAGE)ASD	RED
5	CONNECTOR-CVO	BLUE
6	CONNECTOR	GRAY

NAME	COLOR
CONNECTOR-HOLE	BLACK
CONNECTOR-HOLE	GRAY
CONNECTOR	RED
CAV. (CAGE)ASD	RED
CONNECTOR-CVO	BLUE
CONNECTOR	GRAY

- (1) F37B-14A464-AA
- (2) F9FB-14A464-AA
- (3) F2AB-14A464-AA
- (4) F3LB-14A464-AA
- (5) F47B-14A464-AA
- F5VB-14A464-AA

PART NUMBER

ITEM	DESCRIPTION	REMARKS
1	CONNECTOR-HOLE	BLACK
2	CONNECTOR-HOLE	GRAY
3	CONNECTOR	RED
4	CAV. (CAGE)ASD	RED
5	CONNECTOR-CVO	BLUE
6	CONNECTOR	GRAY

NAME	AMERICAN ELECTRIC
CAGE	
STOCK	40-140464
SPECIFICATIONS	
NOTES	

AMERICAN ELECTRIC, CORP., A Division of American Electric Power Products Corp.
1110 Broadway, New York, N.Y. 10010
Manufactured for American Electric Power Products Corp.

AMERICAN ELECTRIC
CORPORATION
NEW YORK, N.Y.

DATE 5/22/70
PART NO. F5VB-14A464-AA
CONTRACTOR CODE 53561

A B C D E F G H I J K L M

S3551

NOTES : UNLESS OTHERWISE SPECIFIED

1. GENERAL TOLERANCES
SEE ALL ONE PLACE DIMENSIONS
AND ALL TWO PLACE DIMENSIONS.
2. USE WITH SPACER NO. E9E3-144465-B4 AND TERMINAL
NO. E9E9-14474-BA.
3. VENDOR - USA
4. WHEN MATING TO THE APPLICABLE MALE CONNECTOR
THE FOLLOWING DATA VALUES SHALL APPLY:
 A. INSERTION MATING FORCE MAXIMUM 4-12 LBS
IS 27 LBS.
 B. RETENTION FORCE OF MATED CONNECTORS IS
22-125 NEWTONS 8-35 STOOGES.
5. PARTS MUST BE FREE OF FLASH AND IMPERFECTIONS
THAT MAY AFFECT FIT OR FUNCTION.
6. MUST MEET ENGINEERING SPEC. E9T2B-144464-B4
TESTS 1, 2, 3, 4, 5, 6, 12, 13, 14, & 15.

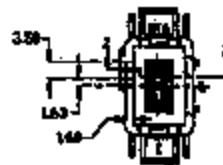
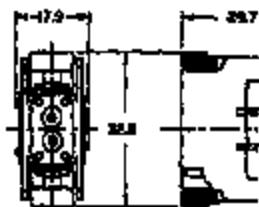


FIG. 2

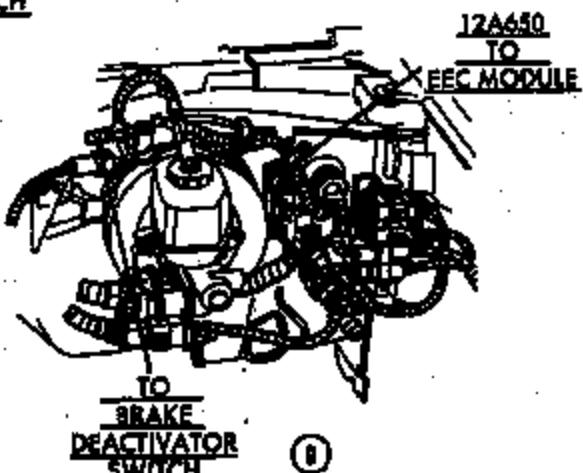
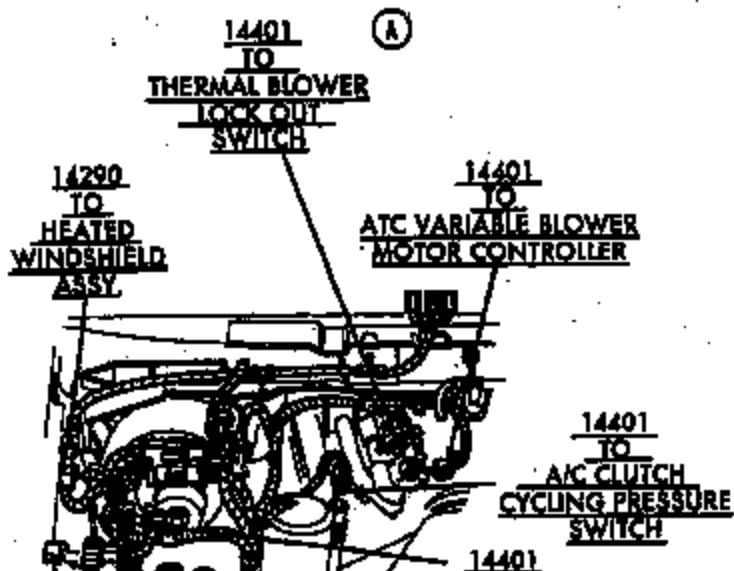
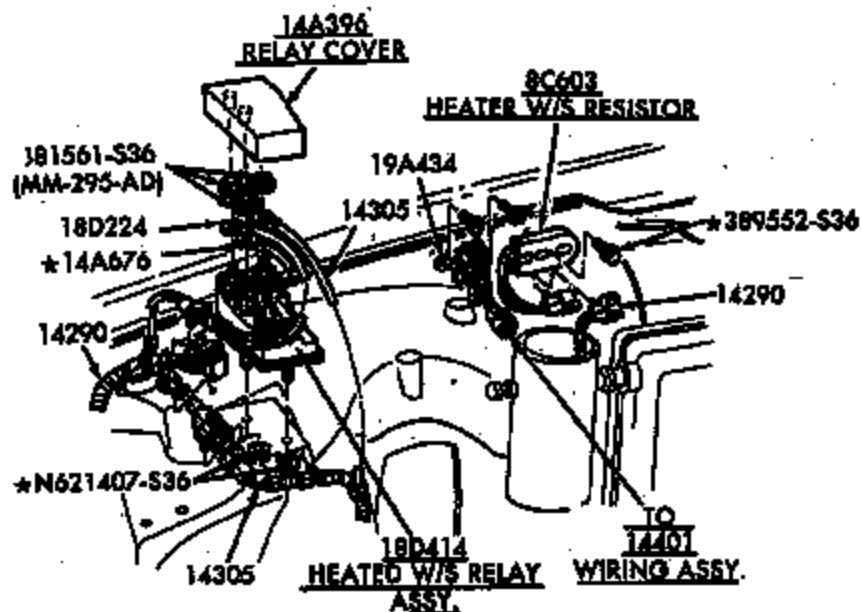
PICTORIALLY SAME AS FIG. 1 EXCEPT 2&3
POLARIZING SLOT IS OFFSET AS SHOWN.



			F18/F16	F16						
			1	1	2	1	1	1	1	2
			F18-144464-RLA							
			E9CB-144464-744							
			F3LB-144464-744							
			F4LB-144464-744							
			F5VB-144464-744							

NO.	REF. NO.	NAME	NO.	REF. NO.	NAME	ITEM	PART NO.	NAME	COLOR
①	②	③	④	⑤	⑥	⑦	26-400-07-1	SLIQUE-VINYL CONNECTOR-MALE	BLACK
⑥	⑦	⑧	⑨	⑩	⑪	⑫	188-900-17-1	SLIQUE-VINYL CONNECTOR-MALE	WHITE
⑨	⑩	⑪	⑫	⑬	⑭	⑮	8007-01-00-1	GARNET-VINYL CONNECTOR	RED
⑩	⑪	⑫	⑬	⑭	⑮	⑯	44757	GARNET-VINYL CORD HOLDING	RED
⑪	⑫	⑬	⑭	⑮	⑯	⑰	7429-14503-1A	CONNECT-VINYL CONNECTOR (TVO)	BLUE
⑫	⑬	⑭	⑮	⑯	⑰	⑱	7429-14503-1A	CONNECT-VINYL CONNECTOR	GRAY

① F57B-1
 ② F9FB-1
 ③ F2AB-1
 ④ F3LB-1
 ⑤ F4LB-1
 ⑥ F5VB-1
 PART



TO
Brake
Deactivator
Switch

•ALSO SUPPLIED IN 12A581 WIRING HARNESS
■ONLY SUPPLIED IN 14290 WIRING HARNESS

P-29535

3713 7049

YEAR	MODEL/RESTRICTIONS	ENG.CYL	PART NO.	DESCRIPTION	QTY.
■ 12A381	WEIRE ASY. (ENGINE CONTROL SENSOR) Also Refer to Group(s): 11A298, 12A522, 14289, 9E724 CONT'D				1

Refer to the Cross Reference section in the front of this Catalog for all LD Cross Reference Information.

L (TEMPO), MB (TOPAZ) CONT'D

92/93	B - w/ sports pkg. or M/T - w/o sport pkg. - 30 states or Canada	6 Cyl. 3.0	P33Z 12A381-A		1
92/93	MB - w/ sports pkg. or M/T w/o sports pkg. - 30 states or Canada	6 Cyl. 3.0	P33Z 12A381-H		1
92/93	MB - w/o sports pkg. or M/T w/o sports pkg. - 30 states or Canada	6 Cyl. 3.0	P33Z 12A381-A		1
92/93	B - w/o sports pkg. or M/T w/o sport pkg. - 30 states or Canada	6 Cyl. 3.0	P33Z 12A381-A		1
94	B, MB - 30 states, A/T. W/o speed control	6 Cyl. 3.0	P43Z 12A381-D		1
94	B, MB - Canadian daytimer rou- tine hours, A/T	6 Cyl. 3.0	P43Z 12A381-G		1
94	B, MB - M/T 30 states or Canada	6 Cyl. 3.0	P43Z 12A381-A		1
90	B, MB - AWD - w/o A/C	P03Z 12A381-B	LH	1	
90	B, MB - exc. AWD - w/o A/C - w/o air	* P03Z 12A381-D	Never Incorporated into Produc- tion	1	
91	B, MB - 2WD - A/T w/o bag. w/o air cond/door	P13Z 12A381-E		1	

DC (LINCOLN CONTINENTAL)

90	DC - w/o electric windshield de- froster	6 Cyl. 3.8	P00Y 12A381-B		1
90	DC - w/o electric windshield de- froster	6 Cyl. 3.8	P00Y 12A381-A		1
92	DC - w/ electric windshield defrost- er	6 Cyl. 3.8	P20Y 12A381-B		1
92	DC - w/o electrical windshield de- froster	6 Cyl. 3.8	P20Y 12A381-A		1
94	DC	6 Cyl. 3.8	P40Y 12A381-A	To be used with R-134A Refriger- ant	1
95	DC - w/ traction assist	SCyl. 4.6	F10Z 12A381-BU		1
95	DC - w/o traction assist	SCyl. 4.6	P00Y 12A381-A		1
95	DC - w/ TA	SCyl. 4.6	P00Z 12A381-BH		1
95	DC - w/o TA	SCyl. 4.6	F10Z 12A381-AH		1
97	DC - w/ anti-slip traction brakes	SCyl. 4.6	F10Z 12A381-BA	Same F70Z-12A381-BA	1
97	DC - w/o anti-slip traction brakes	SCyl. 4.6	* F10Z 12A381-AA		1
97	DC	SCyl. 4.6	F10Z 12A381-BC	Same F70Z-12A381-BC	1
98	DC	SCyl. 4.6	F00Z 12A381-AC		1
99	DC	SCyl. 4.6	XF3Z 12A381-AA		1
91	DC - w/o electric windshield de- froster	All	P00Y 12A381-B		1
91	DC - w/o electric windshield de- froster	All	P00Y 12A381-A		1
93/94	DC	All	P00Y 12A381-A	To be used on vehicles W/o R- 134A Refrigerant	1

DT (LINCOLN TOWN CAR)

91	DT	SCyl. 4.6	P0VY 12A381-A		1
92	DT	SCyl. 4.6	P0VY 12A381-A	P71Z - 4K 742	1
93	DT	SCyl. 4.6	P0VY 12A381-A	P71Z - AZ	1
94	DT	SCyl. 4.6	P0VY 12A381-C		1
95	DT	SCyl. 4.6	P0VY 12A381-A		1
96	DT	SCyl. 4.6	+ P6V2 12A381-AF + P6V2 12A381-AI		1
97	DT	SCyl. 4.6	P7VZ 12A381-AC		1
98	DT	SCyl. 4.6	* 12A381	--For 12A381 refer to section 140. In cold temp 1429°	1
99	DT - Calif	SCyl. 5.0	P0VY 12A381-B		1
99	DT - exc. Calif	SCyl. 5.0	P0VY 12A381-A		1

SMMPPDIA

Releasing Part Detail

01/04/99 16:22:11

--> _____

Analyst: IAA

SERVICE PART: P2VY- 12A581-A WIR ASY ENG CONTR SNS

PLIER LOCATION:	E778W UNITED TECHNOLOGIES	Supplier Role:	M
Supplier Code:	E778	PO Effect Dt:	03/10/95
Purchase Ord Nbr:	SC 00976	Buyer:	181
Vendor Part Nbr:		PO Etab Dt:	03/09/95

Lead Time:	120	Piece/Pkg Value:	430.43	Minimum Run:	0
Auth Days:	120	Overpack Quantity:	0	Minimum Ship:	0
Pct of Bus:	100	Pallet Quantity:	20	Ship Multiple:	0
Unit of Iss:	1	Package Issue Date:	02/17/98	Buy Quantity:	26
		Package Eff Date:	05/17/98	Shelf Stock:	0

F1=Help F4=PrevSuplr F5=NextSuplr F9=FabHold F10=ServEngXref
RECORD FOUND

PD01572

3713 7051

SMMPCPA

Buyer Contact

01/04/99 16:22:45

==> _____

SUPPLIER LOCATION: B778W UNITED TECHNOLOGIES

Contact Name:
Phone Number:
Fax Number:

Buyer Remarks:

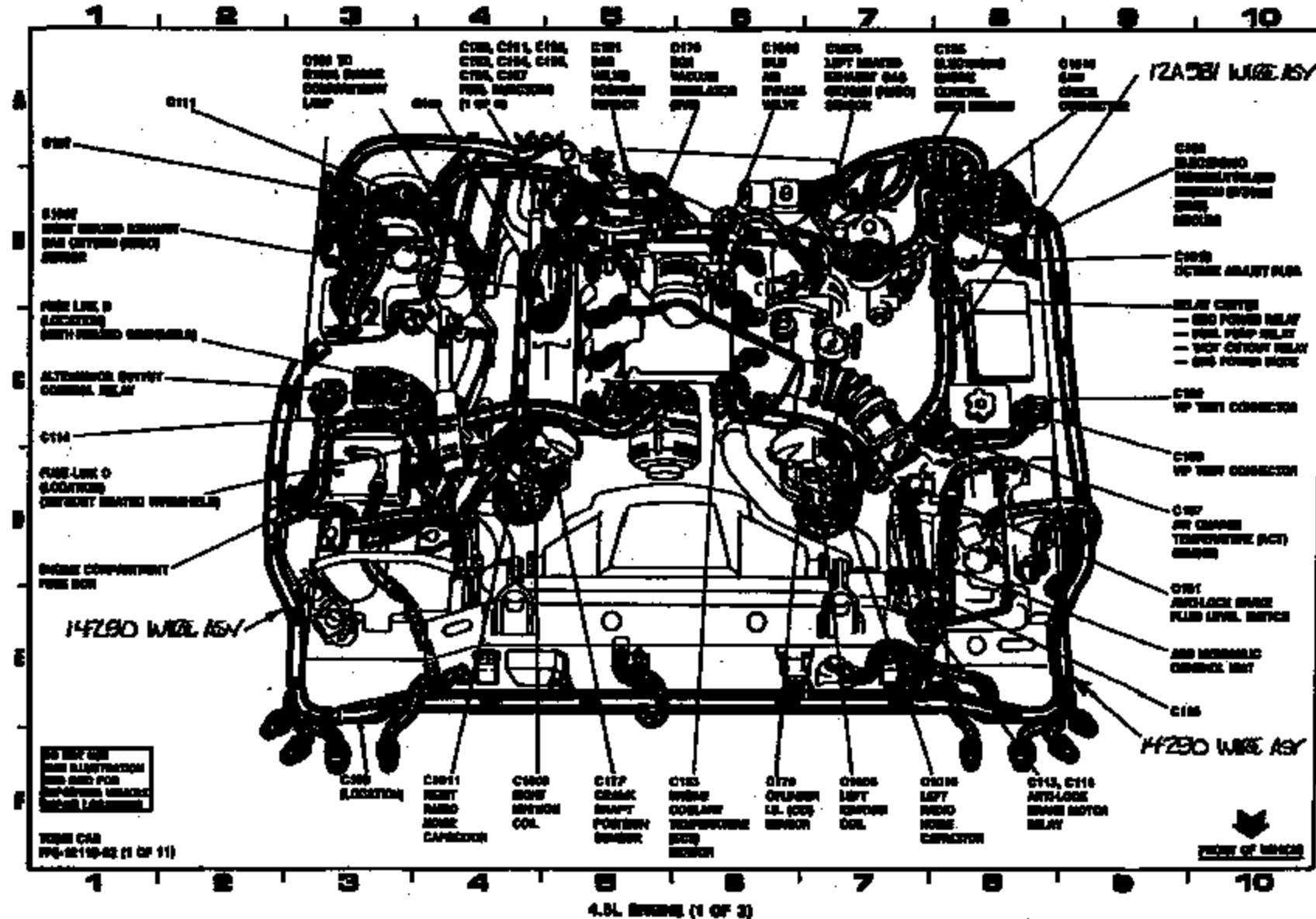
RANDY DREHER FX:614-452-9855 JUDY 593-9171 GARY: 614-452-7541 X235 KRI
JUDY WHITTAKER HANDLES COUNTRY OF MANUFACTURE FOR THIS SITE ONLY.3/98
SERVICE: PAUL TUCKEY 313/240-3041 (7/98)
UPP-JUDY WHITTAKER - SEE #'S ABOVE

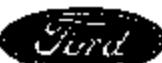
F1=Help
RECORD FOUND

PD01572

3713 7052

151-1 COMPONENT LOCATION VIEWS





Ford Motor Company
Environmental and
Safety Engineering

320 Town Center Drive
Dearborn, Michigan 48124

3M EMT™ 9165 TRANSMISSION
TELEPHONE: (313) 594-2268

AUTOMOTIVE SAFETY OFFICE
PRODUCTION VEHICLE SAFETY
AND COMPLIANCE
FAIRLANE PLAZA SOUTH, Suite 600
DEARBORN, MI 48126

DATE: 99 - JAN - 08

TIME: 8:55 AM

Please deliver to:

Name: NORM LAPONTE

Organization: Excellence D/A

Room # and Building: _____

Telephone: _____

FAX: 78256

Number of Sheets being Transmitted (including this one): 3

Special Instructions/Notes:

Transmission sent by (Name/Telephone):

WM ADAMCZYK 23284

EA02-025

FORD

10/27/03

BOOK 26

DRAWING

DRAWINGS AVAILABLE UPON REQUEST

BRAKE SW.

GEAR CONTROL

FIRE INV.

O.D.I. PEG8-051

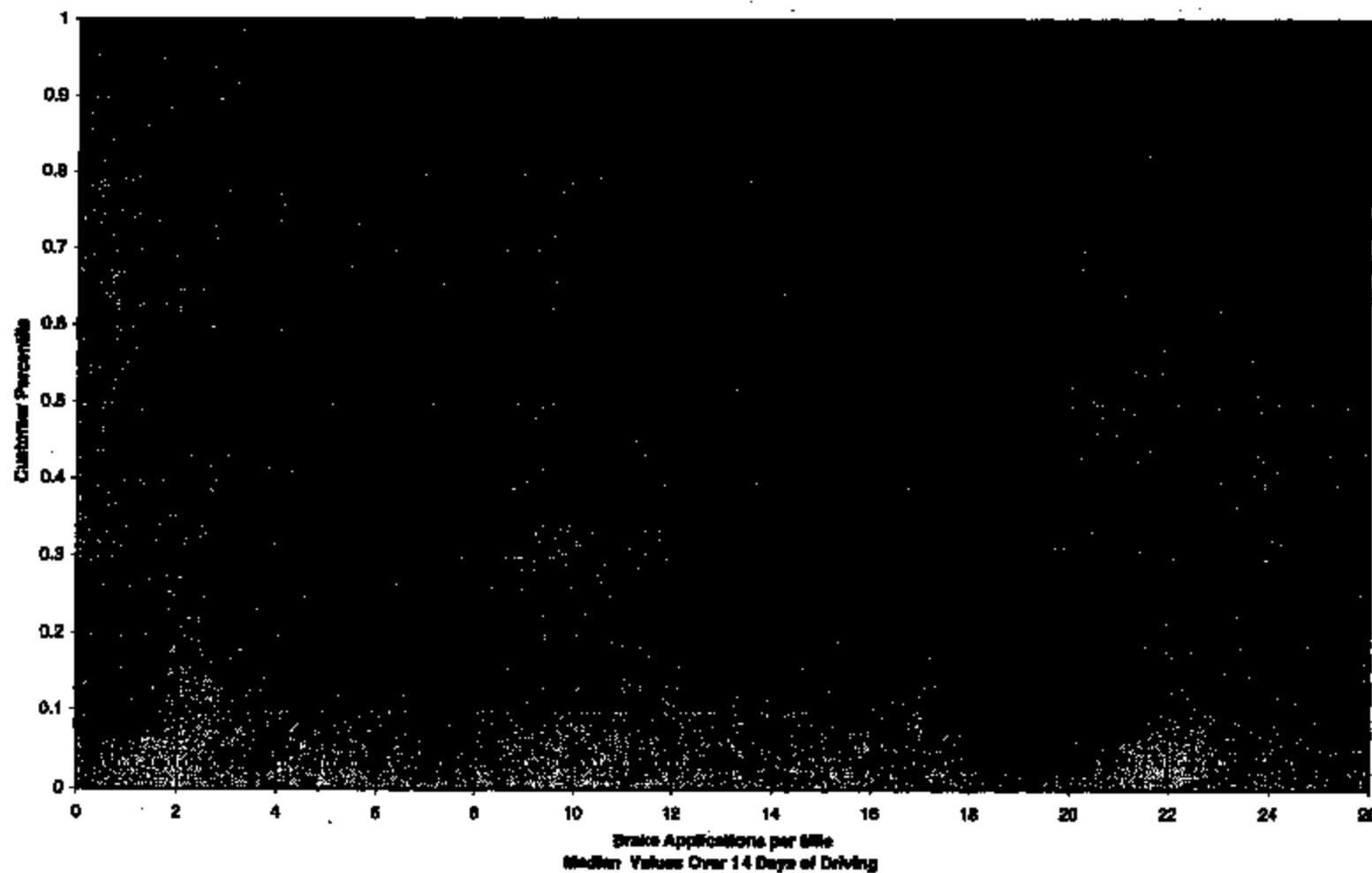
3713 7060

**THIS PAGE ACCOUNTS FOR PAGE(s) REMOVED FOR CONFIDENTIAL FILE
DOCUMENTS CONTAINED IN FILEROOM**

BOX 3713 PAGE(s) 7061-7065 DATE 7/11/2000

Data from 1986 Brake Key Life Test Customer Data Acquisition Study
TO: 020001

Brake Key Life Test - 1986 C.V., G.M. & T.C.

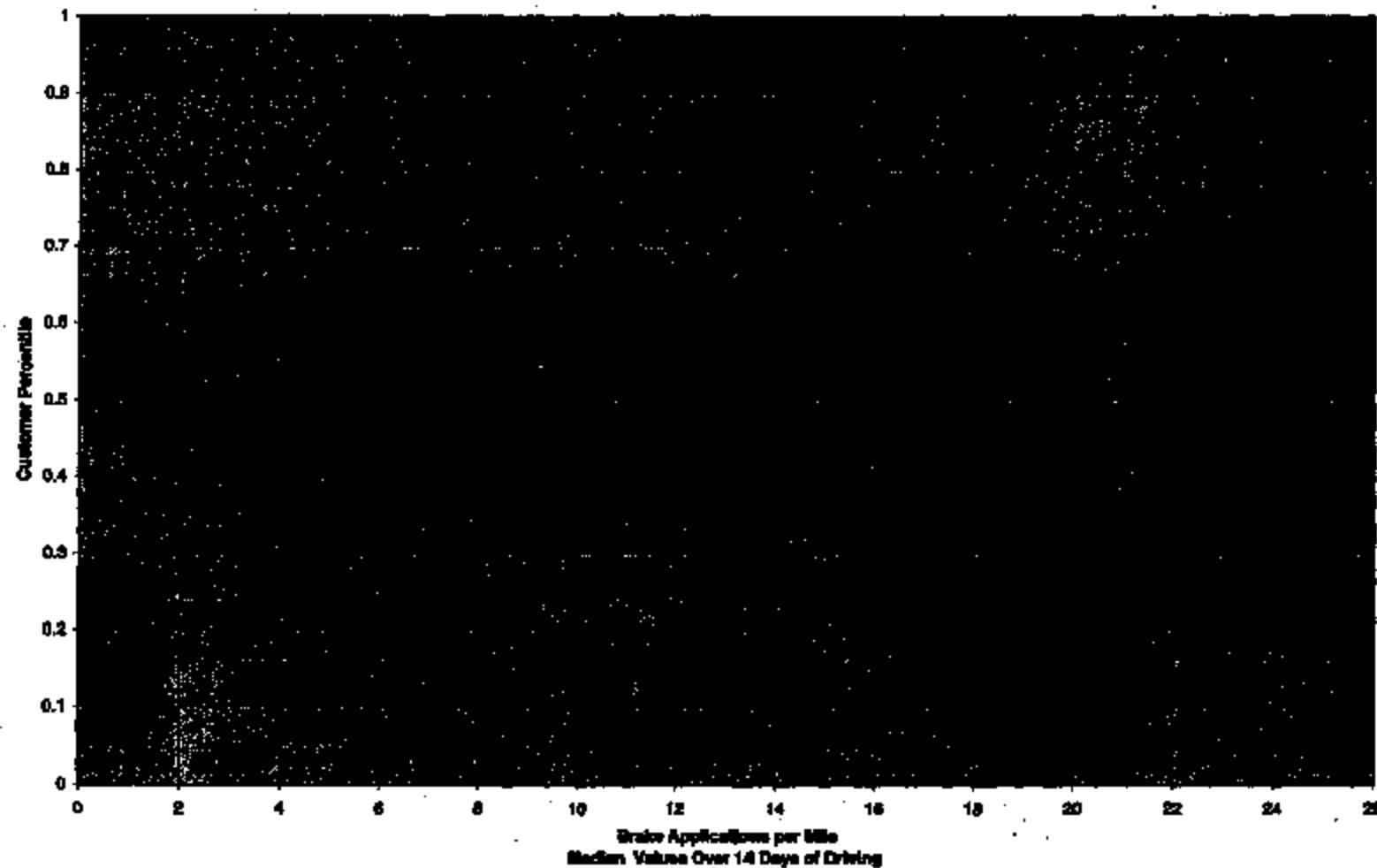


Data from 29 customers total, in cities of Boston, Atlanta, Tampa.

37197986

Data from 1995 Brake Key Life Test Customer Data Acquisition Study
TO: Q20001

Brake Key Life Test - 1995 C.V., G.M. & T.C.

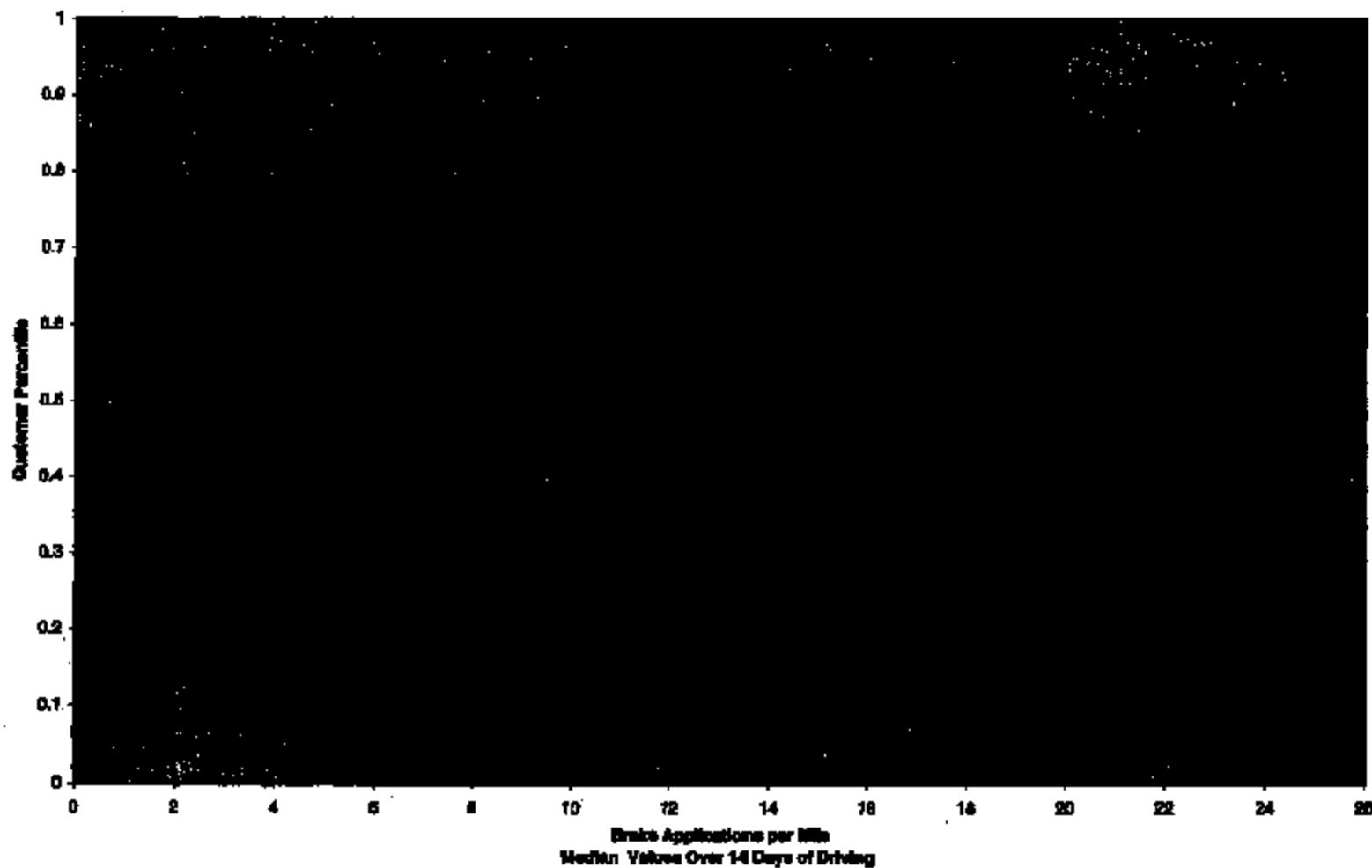


Data from 29 customers total, In cities of Boston, Atlanta, Tampa

3713 7087

Data from 1986 Brake Key Life Test Customer Data Acquisition Study
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Brake Key Life Test - 1986 C.V., G.M. & T.C.

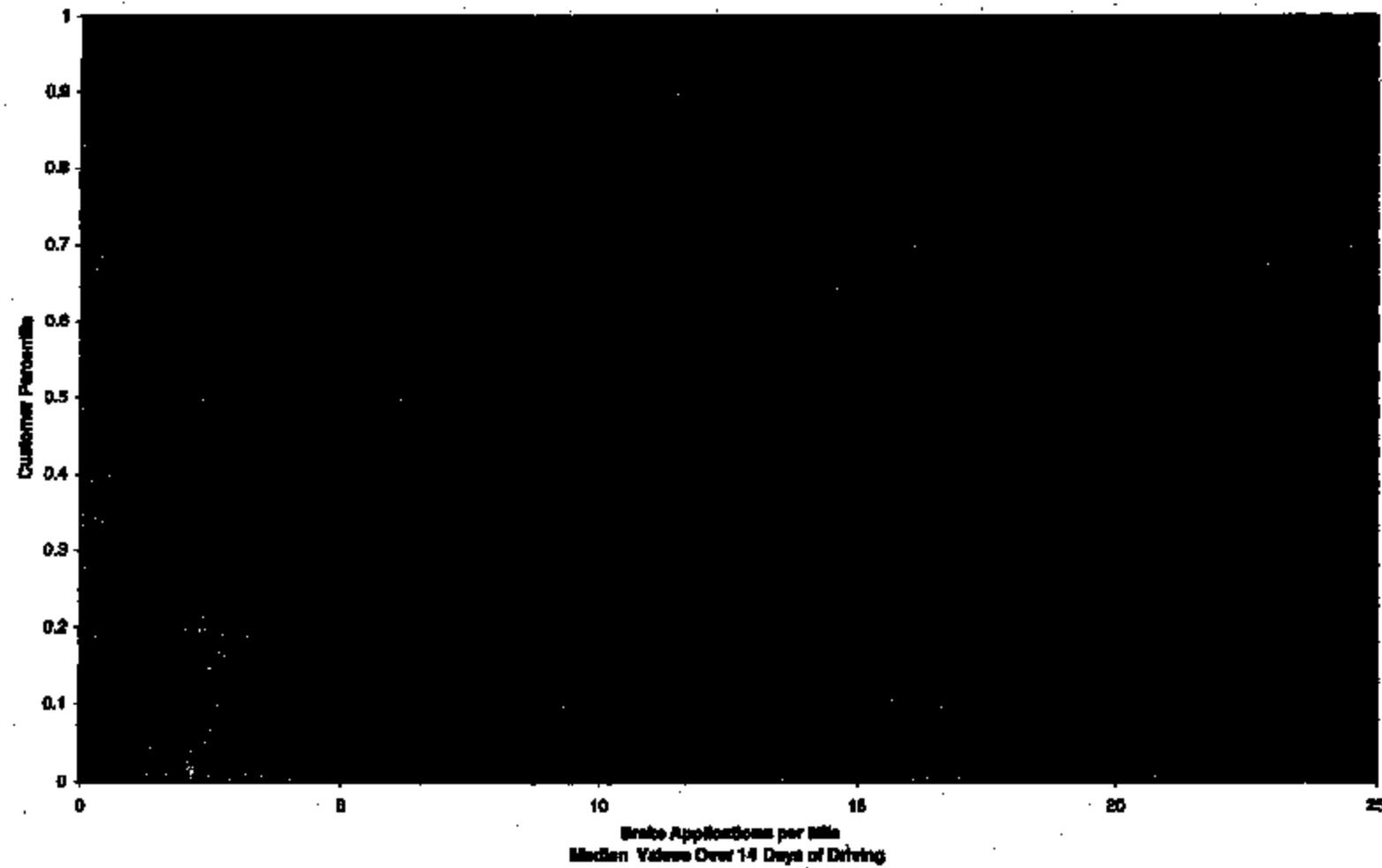


Data from 29 customers total, in cities of Boston, Atlanta Tampa

3713 7069

Data from 1995 Brake Key Life Test Customer Data Acquisition Study
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Brake Key Life Test - 1995 C.V., G.M. & T.C.

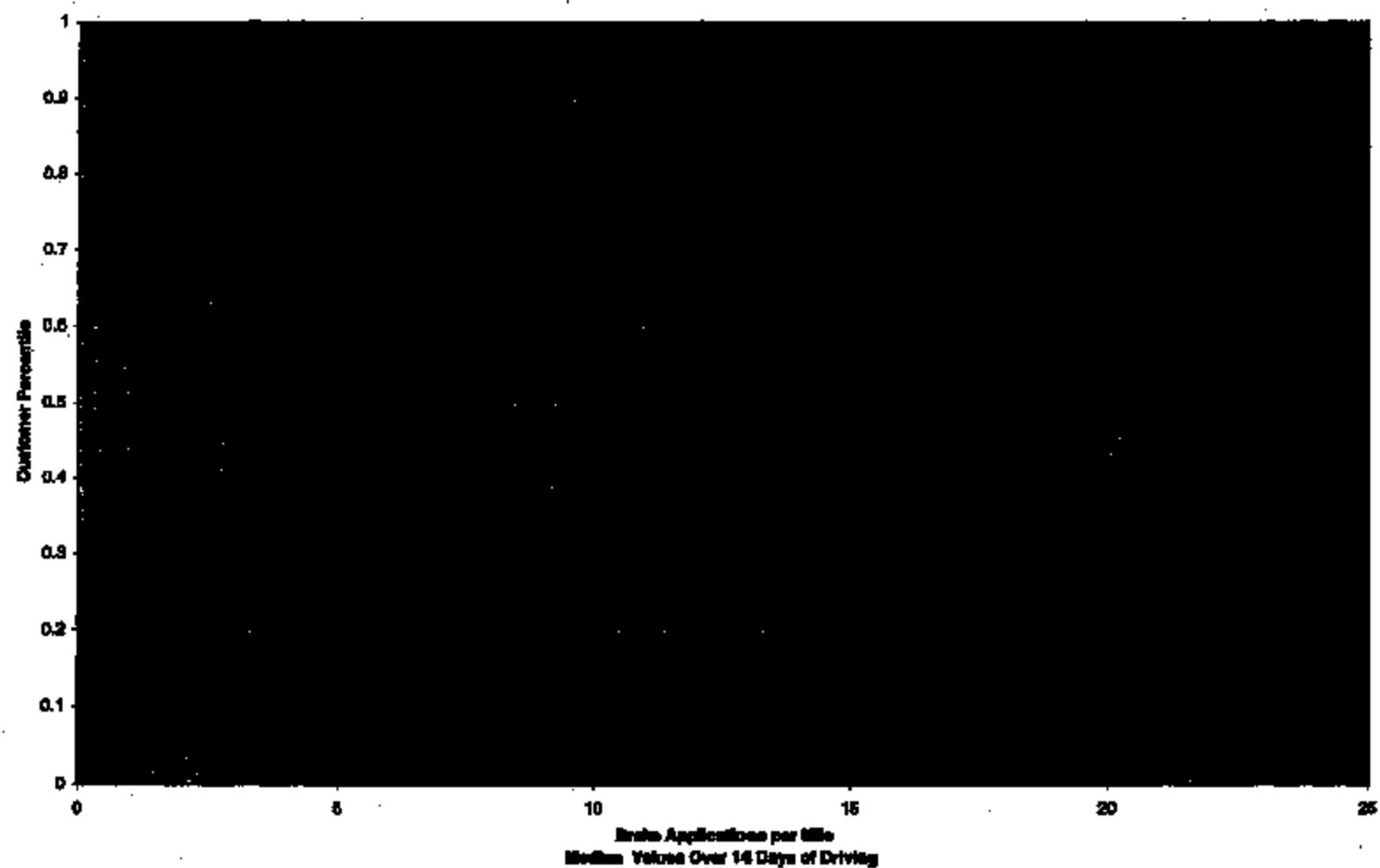


Data from 29 customers total, in cities of Boston, Atlanta, Tampa

37137068

Data from 1996 Brake Key Life Test Customer Data Acquisition Study.
TO: G20001

Brake Key Life Test - 1996 C.V., G.M. & T.C.

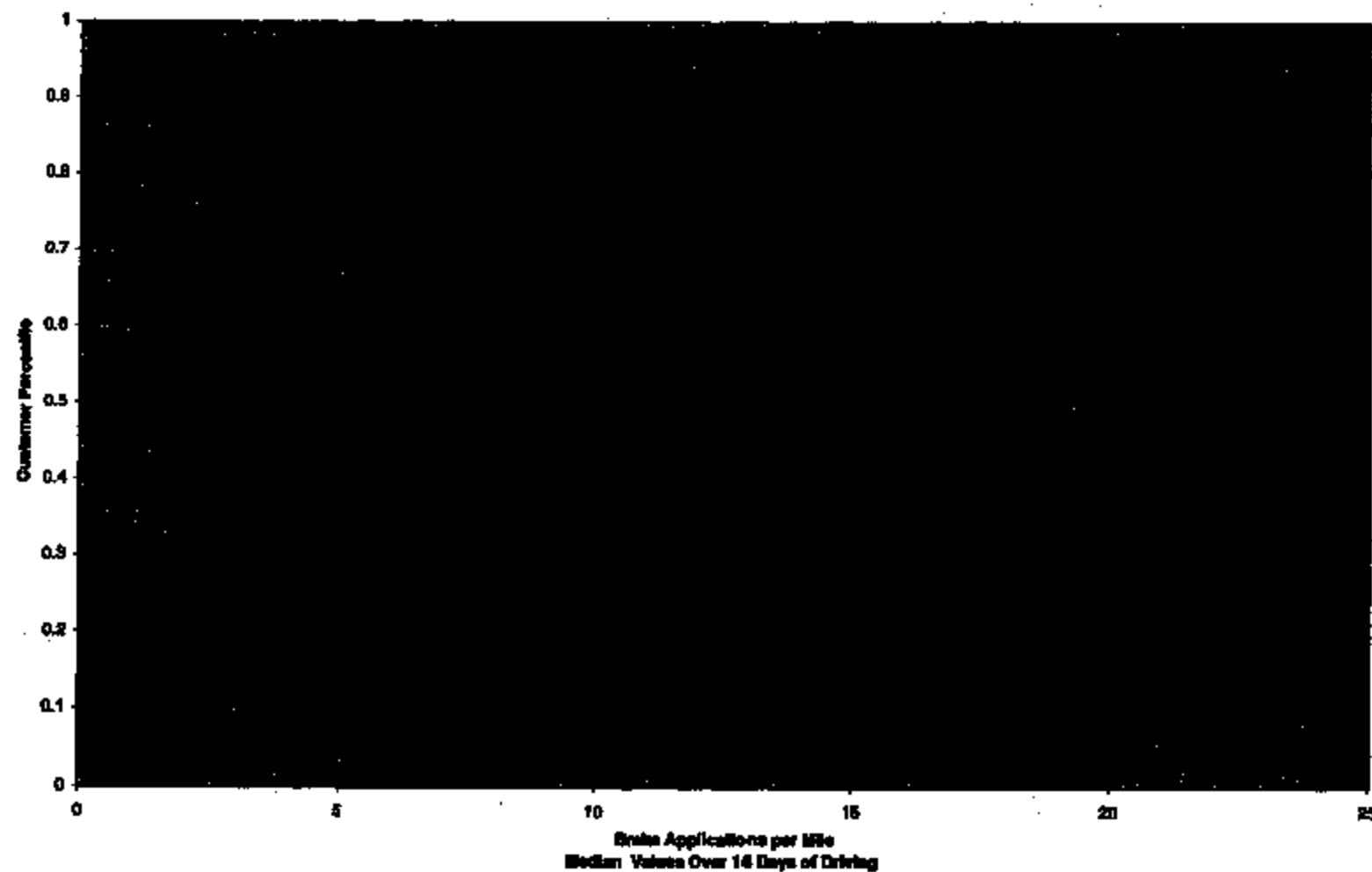


Data from 29 customers total, in cities of Boston, Atlanta Tampa.

3713 7070

Data from 1996 Brake Key Life Test Customer Data Acquisition Study
TD: Q20001

Brake Key Life Test - 1995 C.V., G.M. & T.C.



Data from 29 customers total, in cities of Boston, Atlanta Tampa

3713.071

Table 33 Mean and Median Parameter List

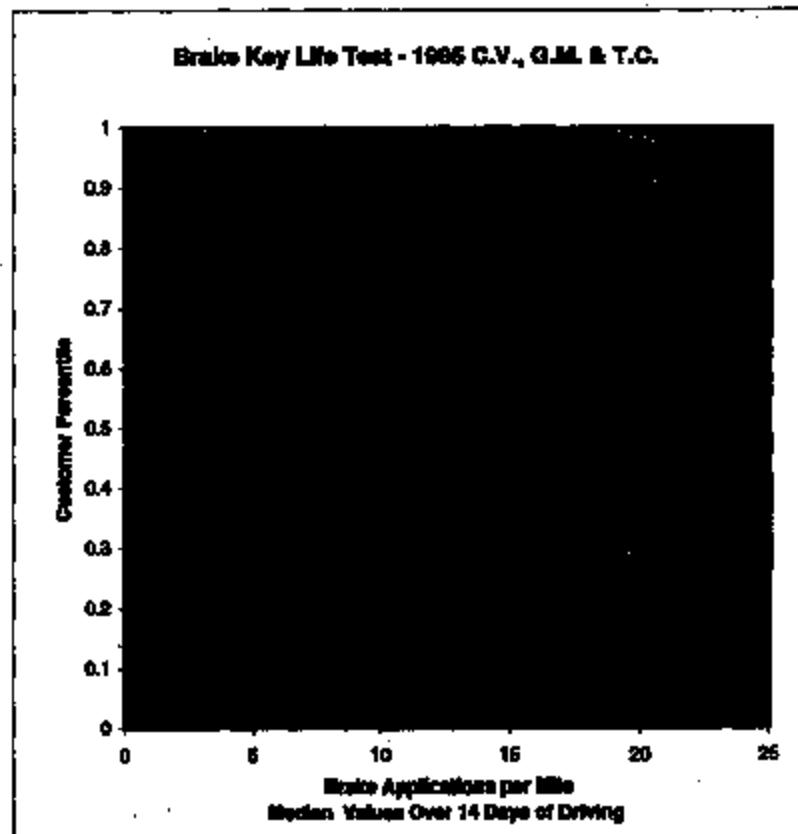
Number	Description
P1	Average Daily Distance Traveled (miles)
P2	Average Daily Vehicle Speed (V<2) (mph)
P3	Average Maximum Daily Vehicle Speed (mph)
P4	Average Daily Idle Time as a Percentage of Operation Time (%)
P6	Average Daily Number of Brake Events
P8	Average Daily Number of Brake Events (V>2)
P7	Average Daily Brake Duration (sec)
P9	Average Daily Brake Duration Standard Deviation (sec)
P10	Average Daily Time between Brake Events (sec)
P11	Average Daily Total Time Braking (sec)
P12	Average Daily Total Time Braking (V>2) (sec)
P13	Average Daily Number of Speed Control Applications
P14	Average Daily Number of Speed Control De-activation Using the Brake
P15	Average Daily Brake Energy per Mile (10^3 lb-ft/mile)
P16	Average Daily Brake Energy per Application (10^6 lb-ft/app)
P17	Average Daily Brake Power per Mile (10^3 hp/mile)
P18	Average Daily Brake Power per Application (10^3 hp/app)
P19	Average Daily Brake Work per Mile per Ton (10^3 lb-ft/m-ton)
P20	Median Daily Distance Traveled (miles)
P21	Median Daily Vehicle Speed (V<2) (mph)
P22	Median Maximum Daily Vehicle Speed (mph)
P23	Median Daily Idle Time as a Percentage of Operation Time (%)
P24	Median Daily Vehicle Operation Time (min)
P25	Median Daily Average Brake Events (V>2) per Mile
P26	Median Daily Number of Brake Events
P27	Median Daily Number of Brake Events (V>2)
P28	Median Maximum Daily Outer Brake Pad Temperature (Deg C)
P29	Median Daily Brake Duration (sec)
P30	Median Daily Brake Duration Standard Deviation (sec)
P31	Median Daily Time between Brake Events (sec)
P32	Median Daily Time between Brake Events Standard Deviation (sec)
P33	Median Daily Total Time Braking (sec)
P34	Median Daily Total Time Braking as a Percentage of Operation Time (%)
P35	Median Daily Total Time Braking (V>2) (sec)
P36	Median Daily Total Time Braking (V>2) as a Percentage of Operation Time (%)
P37	Median Daily Number of Speed Control Applications
P38	Median Daily Number of Speed Control De-activation Using the Brake
P39	Median Daily Brake Energy per Mile (10^3 lb-ft/mile)
P40	Median Daily Brake Energy per Application (10^6 lb-ft/app)
P41	Median Daily Brake Power per Mile (10^3 hp/mile)
P42	Median Daily Brake Power per Application (10^3 hp/app)
P43	Median Daily Brake Work per Mile per Ton (10^3 lb-ft/m-ton)
P44	Maximum Daily Maximum Outer Brake Pad Temperature (Deg C)

Table 34 Overall Average and Median Summary Results for All Braké Key Life Test Study Participants. (Continued)

Participant Number	Type	Site	No. of days of data	P15	P16	P17	P18	P19	P20	P21	P22	P23	P24	P25	P26	P27
1	NC	1	13	194.1	26.68	62.9	13.17	68.5	35.1	35.0	71.0	17.37%	77.1	4.85	227	198
2	R	1	3	137.1	15.18	98.5	10.70	65.5	24.2	31.0	60.0	12.50%	80.7	8.98	328	260
3	N	1	14	388.6	22.80	198.2	18.94	158.8	19.1	21.0	56.0	18.92%	75.1	15.86	438	377
4	R	1	10	192.3	40.38	81.3	18.93	63.6	24.0	31.0	67.5	20.44%	61.0	6.03	140	121
5	N	1	14	230.8	20.98	130.2	11.74	109.9	13.8	25.5	49.5	20.72%	66.0	13.71	274	219
6	N	1	11	322.3	25.27	153.4	11.94	163.8	14.3	20.0	44.0	27.65%	58.7	13.38	200	157
7	R	1	14	121.7	34.03	46.7	12.41	57.7	48.5	34.5	56.5	10.27%	74.3	3.52	180	152
8	N	1	12	249.8	21.74	149.2	12.98	108.5	61.0	31.0	71.0	20.38%	126.8	13.19	674	585
9	R	1	14	243.8	39.08	89.0	13.68	115.2	32.5	27.0	49.8	19.79%	112.0	6.79	276	230
10	NC	1	14	248.6	20.73	160.7	12.65	118.8	15.3	25.0	44.5	13.63%	41.7	12.90	247	172
11	R	2	14	218.3	28.57	77.7	10.04	103.9	20.4	31.0	63.0	22.76%	56.2	8.19	259	198
12	N	2	13	248.8	30.21	103.9	12.79	108.1	43.8	35.0	55.0	23.06%	130.2	6.21	402	349
13	N & R	2	14	228.9	26.97	85.6	9.82	108.1	12.8	25.5	63.0	21.06%	37.2	7.96	157	126
14	R	2	13	219.9	18.08	94.1	7.72	104.4	33.0	26.0	62.0	20.82%	91.7	12.71	423	323
15	NC	2	12	161.8	26.39	91.0	14.93	72.5	56.0	43.0	71.5	9.15%	88.8	5.39	400	346
16	NC	2	13	201.5	29.04	79.4	10.91	87.8	15.8	29.0	66.0	20.56%	49.5	6.92	140	98
17	N	2	11	362.4	37.90	123.5	13.58	167.0	33.8	29.0	60.0	25.86%	126.3	9.27	408	401
18	N	2	13	255.2	32.04	115.6	14.39	121.5	37.8	35.0	67.0	26.06%	89.2	7.98	268	230
19	R	2	6	101.8	29.88	32.8	9.45	48.1	47.8	37.5	60.5	16.78%	98.8	3.56	179	156
20	R	2	11	149.5	36.85	71.7	17.02	71.4	44.1	46.0	77.0	18.20%	70.6	4.22	186	160
21	N	3	14	233.2	42.49	77.1	14.35	110.2	27.8	29.5	66.0	19.23%	79.4	5.56	294	193
22	R	3	6	129.8	34.80	47.4	11.89	61.8	32.9	31.5	50.5	11.78%	61.8	3.90	125	90
23	R	3	14	255.8	26.98	105.4	11.04	121.2	26.1	31.0	67.0	21.79%	70.8	9.15	325	282
24	NC	3	13	178.8	21.88	116.8	11.84	77.7	27.4	23.0	62.0	19.52%	72.2	7.50	228	183
25	N	3	13	240.2	26.00	92.3	10.44	114.3	24.9	29.0	61.0	19.42%	70.3	8.79	372	256
26	R	3	14	186.2	40.18	60.8	16.18	76.0	33.5	41.0	66.8	13.90%	58.7	4.38	180	162
27	R	3	13	250.8	38.92	77.9	11.08	118.8	17.4	26.0	52.0	24.86%	64.0	7.01	180	122
28	N	3	10	248.4	28.44	112.5	10.83	107.9	18.0	26.0	49.0	27.84%	58.6	8.61	248	155
29	N	3	11	298.4	36.08	98.4	13.62	137.7	15.4	31.0	53.0	21.37%	44.8	7.83	195	140

Sheet1

Participant	p20	b-app	p26 miles	b-app./mi	rank	order
19	47.8	179	3.74477	0.02381	1	
7	43.5	180	4.137831	0.057823	2	
20	44.1	196	4.444444	0.091837	3	
22	22.9	125	5.468615	0.12585	4	
27	39.5	189	5.841791	0.159864	5	
4	24	140	5.838833	0.193878	6	
1	36.1	227	6.467238	0.227881	7	
15	56.8	400	7.194245	0.261906	8	
24	27.4	229	8.357684	0.295918	9	
21	27.6	234	8.478281	0.328932	10	
9	32.5	276	8.482808	0.369946	11	
16	16.8	140	8.880759	0.397859	12	
12	43.8	402	9.178032	0.431973	13	
18	37.8	368	9.470889	0.465986	14	
28	17.4	180	10.84468	0.5	15	
23	26.1	325	12.45211	0.634014	16	
13	12.6	157	12.48032	0.588027	17	
30	15.4	185	12.68234	0.602041	18	
11	20.4	259	12.69608	0.636054	19	
14	33	433	13.12121	0.670068	20	
8	51	674	19.21588	0.704082	21	
2	24.2	329	13.59504	0.738095	22	
29	18	248	13.77778	0.772109	23	
6	14.3	200	13.98801	0.806122	24	
17	33.6	496	14.82143	0.840136	25	
25	24.8	372	14.93876	0.87415	26	
10	15.8	247	16.14379	0.908163	27	
5	13.8	274	18.85607	0.942177	28	
3	19.1	436	22.82723	0.97619	29	



file/BSwCU/nit/ 2-18-88

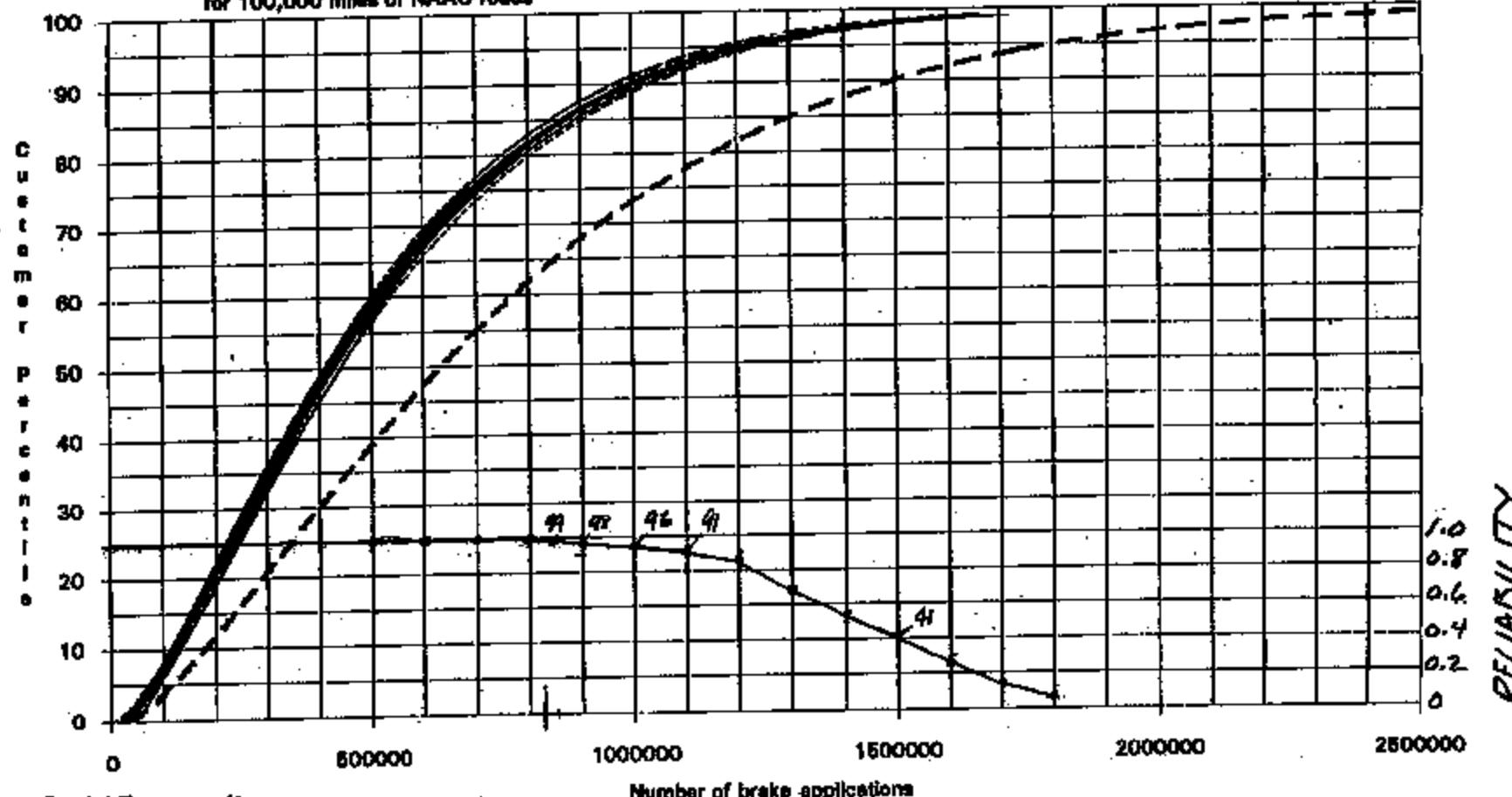
87197074

1991 Thunderbird 6.0L, AOD
NAAO Customer Usage Spectrum

Heavy solid curve represents the
average of 9 Monte Carlo projections
for 100,000 miles of NAAD roads

Heavy dashed curve represents
the 150,000 mile projection

16.~99%



Bandwidth represents
variability of 9 Monte Carlo
projections for 744
Thunderbird owners

3713 7076

AGPRWG W37.XLC 8/8/04

Sheet1

Order	Cycle	Rank
1	994232	0.038043
2	994232	0.092391
3	1131102	0.148739
4	1131102	0.201087
5	1302691	0.256435
6	1328190	0.308763
7	1344673	0.364113
8	1359659	0.418478
9	1359659	0.472828
10	1403522	0.527174
11	1403522	0.581522
12	1403522	0.635887
13	1418250	0.690217
14	1418250	0.744585
15	1472621	0.798913
16	1488221	0.853281
17	1511728	0.907609
18	1587659	0.961957

Cor. Curve 100,000mi
@ 99% 1.6x10⁶ Brake Appl.
∴ 16 Brake Appl./mi

Reddick Veh ~ 51,500mi
∴ B.A. = $\frac{16}{\text{mi}}(51,500) = 824,000 \sim 82\%$

Weibull Curve
@ 824,000 cycles $\Rightarrow 2.2\% F_x$

@ ~ 725000/0.01

Contamination? ↓
High Cycle count \rightarrow Diaphragm \rightarrow Brake Fluid \rightarrow Seal Contact
 F_x Leakage \rightarrow because -

ABS Effects?

Sudden Environmental
Temp
Salt
Humidity?

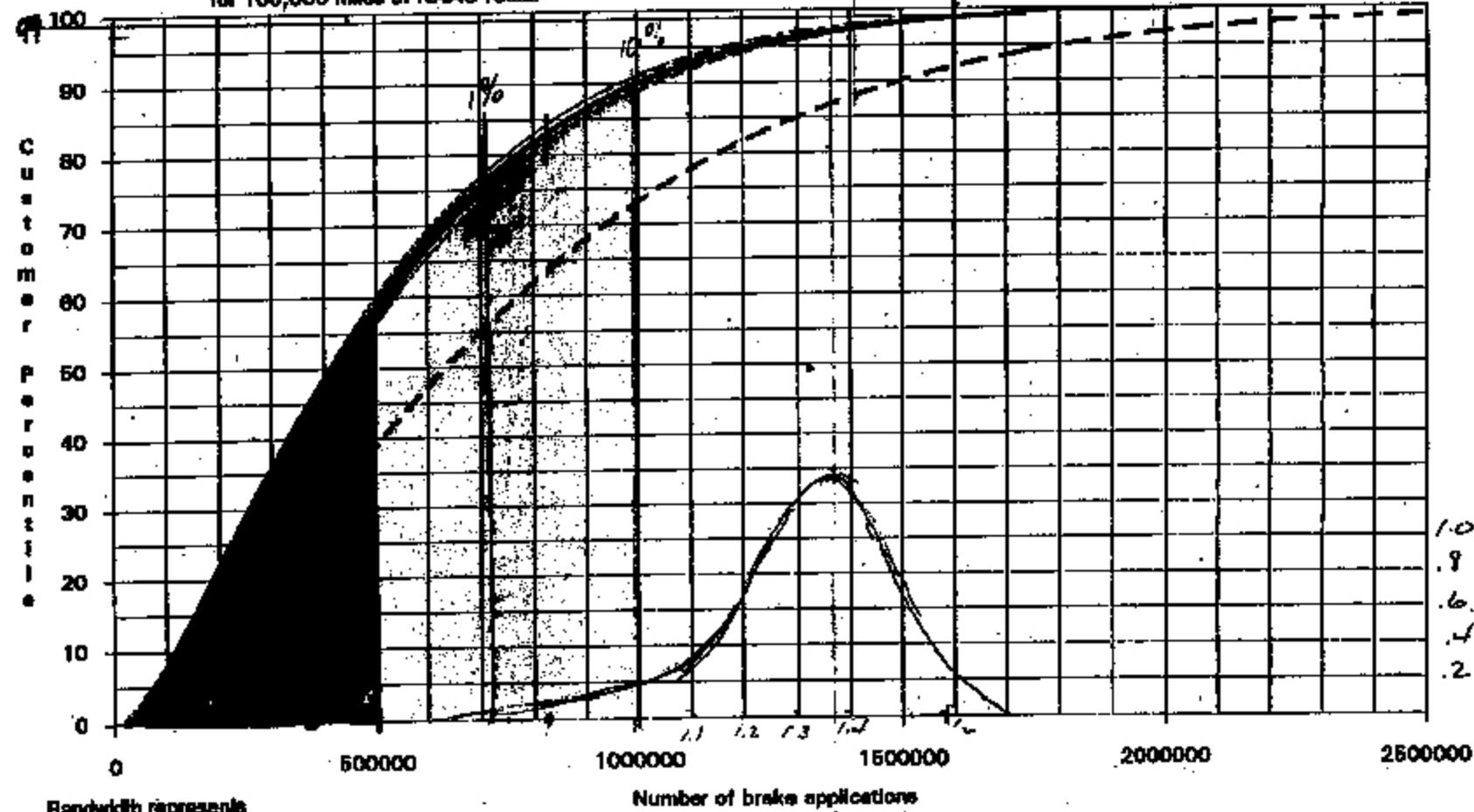
External
accumulation?

? TI & Fluid
unable to
seal fluid

1991 Thunderbird 5.0L, AOD
NAAO Customer Usage Spectrum

Heavy solid curve represents the
average of 9 Monte Carlo projections
for 100,000 miles of NAAO roads

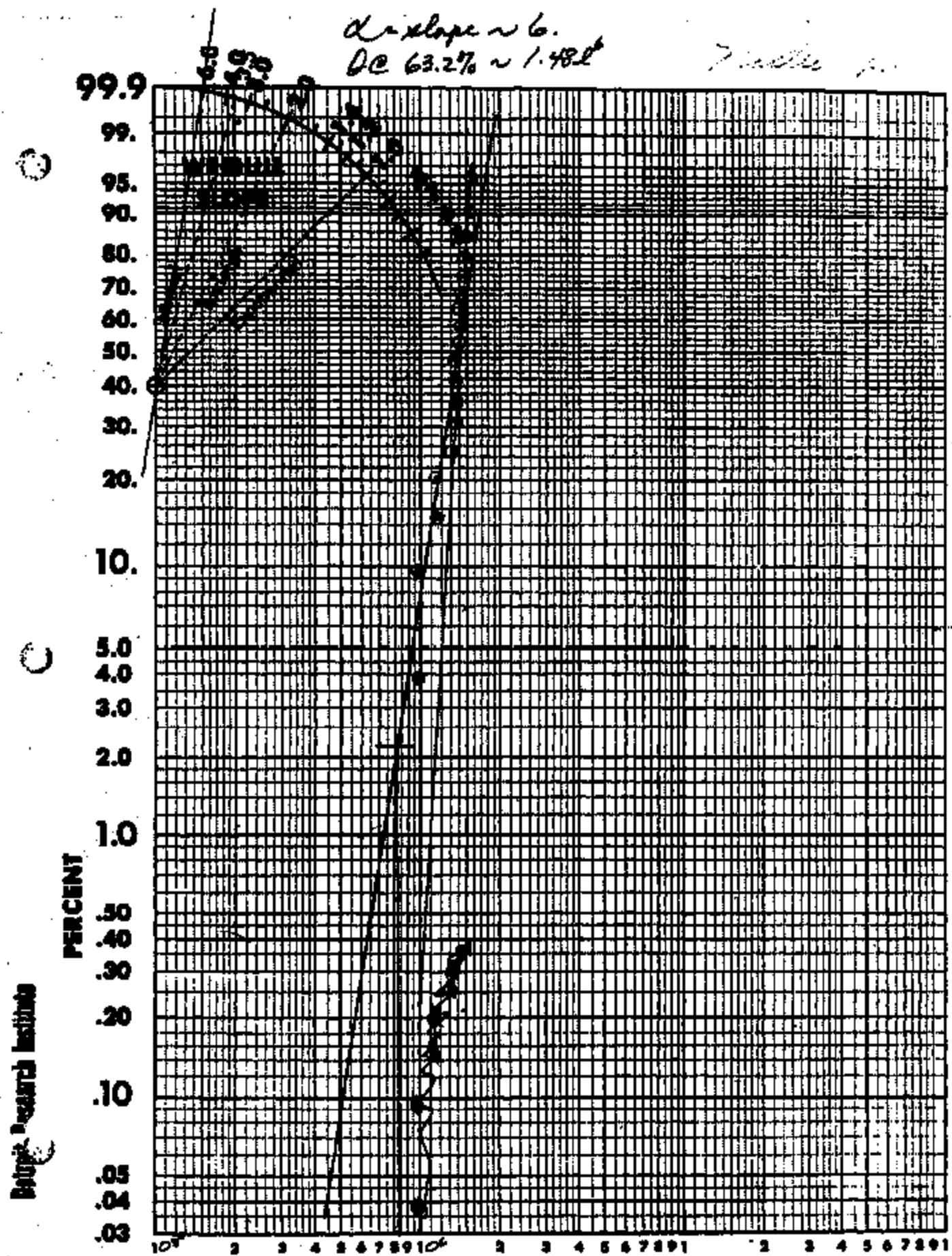
Heavy dashed curve represents
the 150,000 mile projection



Bandwidth represents
variability of 9 Monte Carlo
projections for 744
Thunderbird owners

ASPIRING W27.XLC 6/5/94

37137077



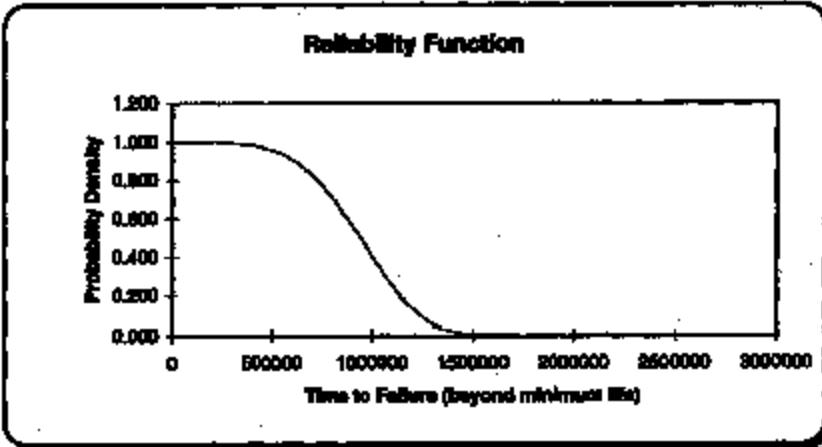
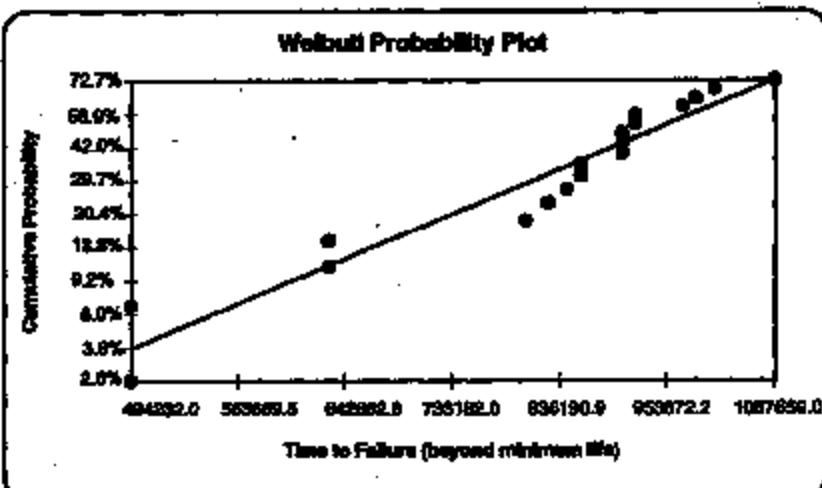
3713 7078

2 and 3 parameter WEIBULL FAILURE ANALYSIS

CLICK here to recalculate W Weib data
or "Minimum life" input value is altered.

131.000	0.000000000
131.250	0.015625000
131.500	0.031250000
131.750	0.046875000
132.000	0.062500000
132.250	0.078125000
132.500	0.093750000
132.750	0.109375000
133.000	0.125000000
133.250	0.140625000
133.500	0.156250000
133.750	0.171875000
134.000	0.187500000
134.250	0.203125000
134.500	0.218750000
134.750	0.234375000
135.000	0.250000000
135.250	0.265625000
135.500	0.281250000
135.750	0.296875000
136.000	0.312500000
136.250	0.328125000
136.500	0.343750000
136.750	0.359375000
137.000	0.375000000
137.250	0.390625000
137.500	0.406250000
137.750	0.421875000
138.000	0.437500000
138.250	0.453125000
138.500	0.468750000
138.750	0.484375000
139.000	0.500000000
139.250	0.515625000
139.500	0.531250000
139.750	0.546875000
140.000	0.562500000
140.250	0.578125000
140.500	0.593750000
140.750	0.609375000
141.000	0.625000000
141.250	0.640625000
141.500	0.656250000
141.750	0.671875000
142.000	0.687500000
142.250	0.703125000
142.500	0.718750000
142.750	0.734375000
143.000	0.750000000
143.250	0.765625000
143.500	0.781250000
143.750	0.796875000
144.000	0.812500000
144.250	0.828125000
144.500	0.843750000
144.750	0.859375000
145.000	0.875000000
145.250	0.890625000
145.500	0.906250000
145.750	0.921875000
146.000	0.937500000
146.250	0.953125000
146.500	0.968750000
146.750	0.984375000
147.000	1.000000000

Calculated Output:

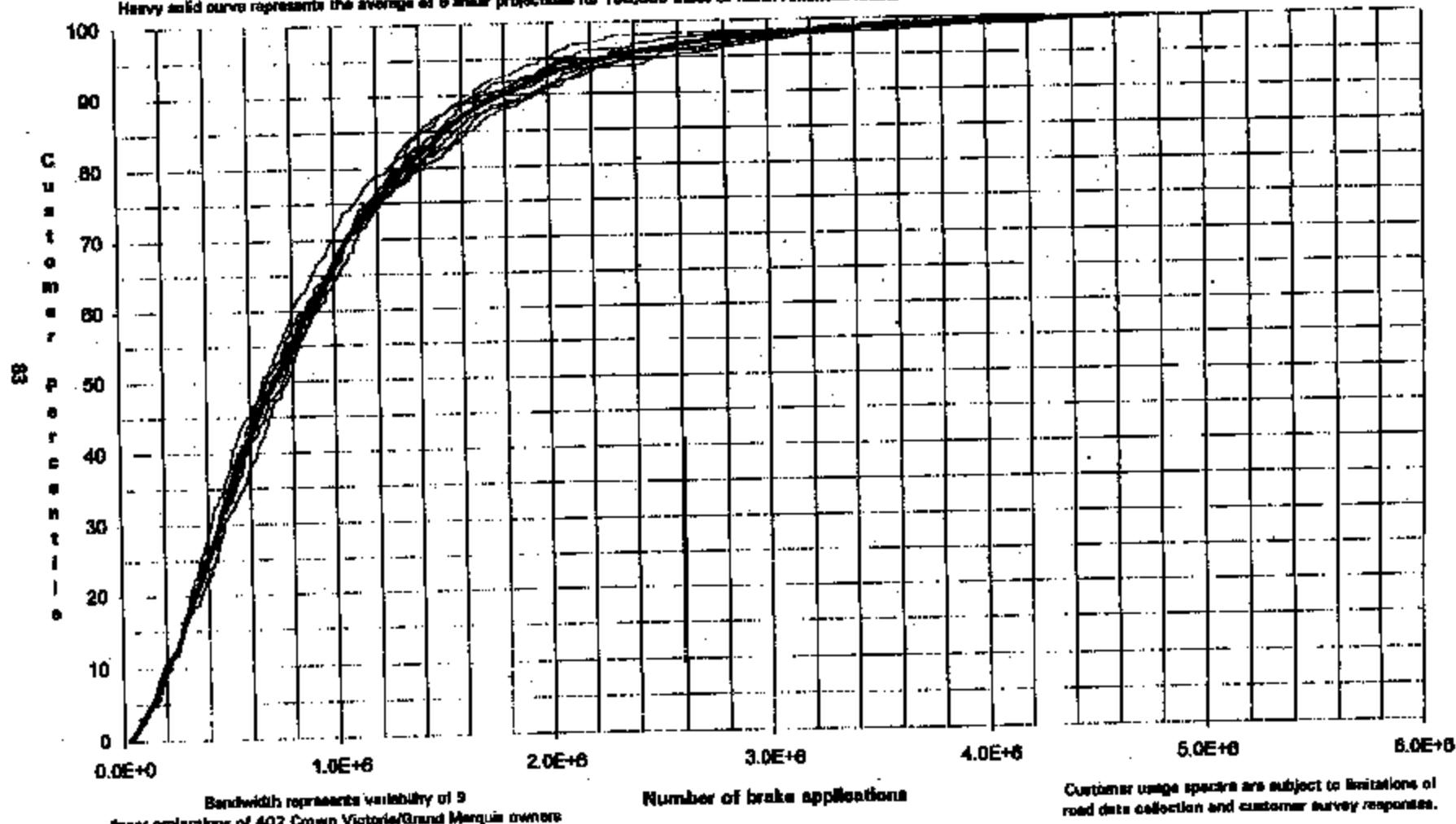


NOTES:

1992 CROWN VICTORIA (4.6L, AODE)
NORTH AMERICAN POWERTRAIN CUSTOMER USAGE SPECTRUM

15.3 14.6 14 17.3 18.7 20

Heavy solid curve represents the average of 9 linear projections for 150,000 miles of North American roads.



3713 7080

1992 Crown Victoria North American Powertrain Customer Correlation
GTO - Customer Correlation - U.S.
Revised: March 1993

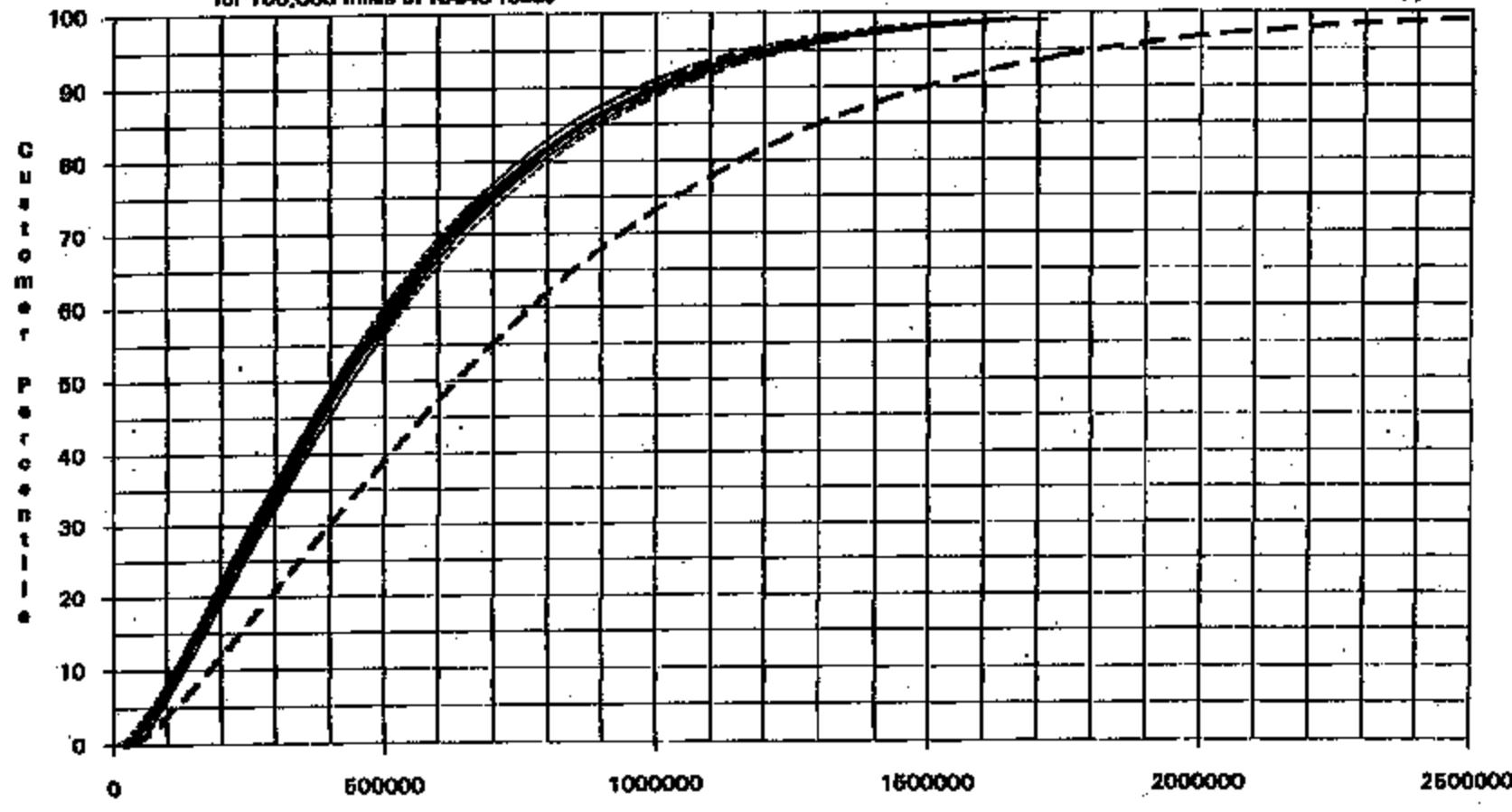
Customer usage spectra are subject to limitations of road data collection and customer survey responses.

WV1.DC
REV6
2/93

1991 Thunderbird 5.0L, AOD
NAAO Customer Usage Spectrum

Heavy solid curve represents the
average of 9 Monte Carlo projections
for 100,000 miles of NAAO roads

Heavy dashed curve represents
the 150,000 mile projection



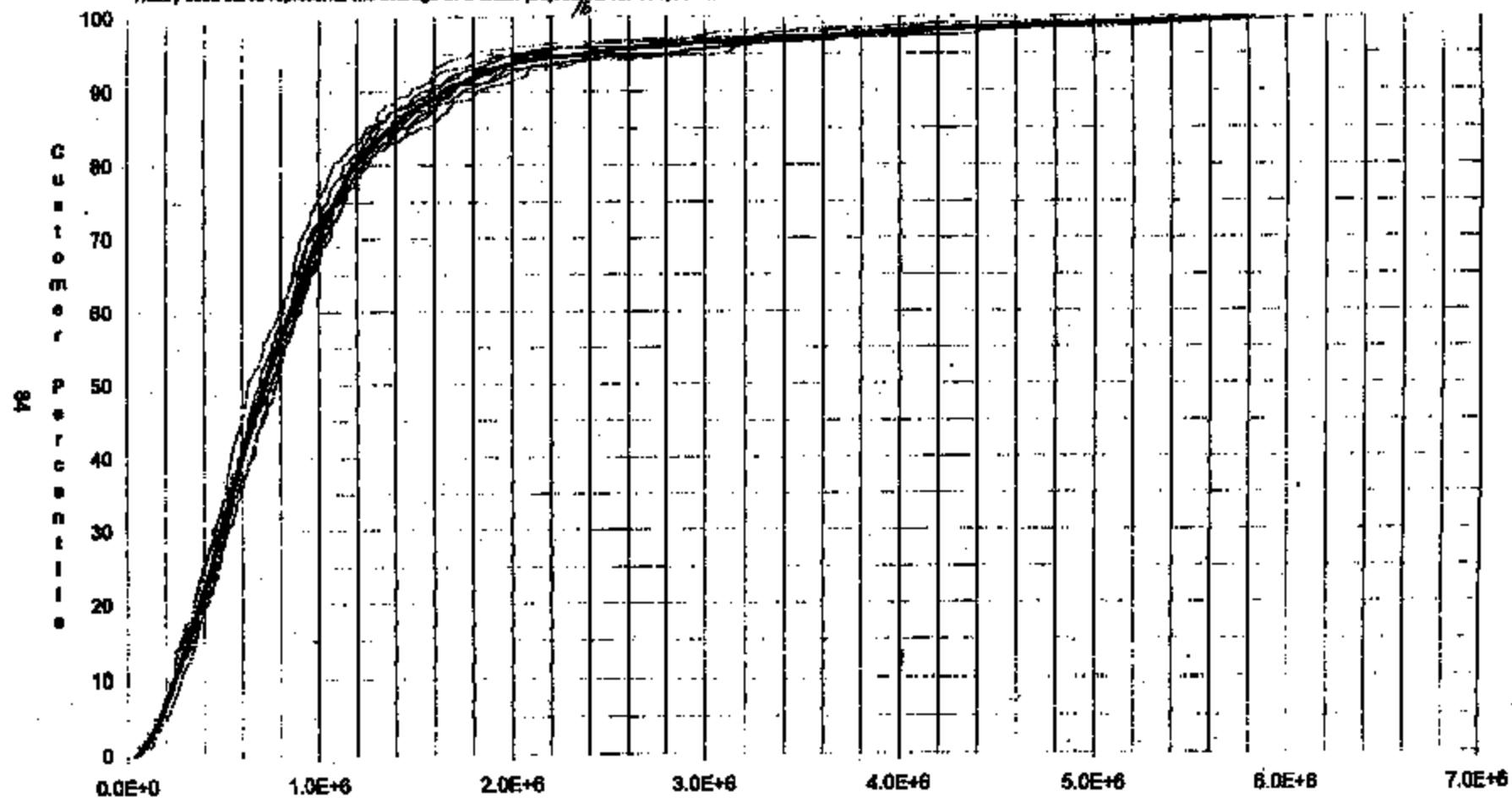
Bandwidth represents
variability of 9 Monte Carlo
projections for 744
Thunderbird owners

ASPIRING.W37.XLC 8/8/94

1991 8/8/94

1992 TAURUS (3.0L, AXODE)
NORTH AMERICAN POWERTRAIN CUSTOMER USAGE SPECTRUM

Heavy solid curve represents the average of 9 linear projections for 150,000 miles of North American roads.



Bandwidth represents variability of 9
linear projections of 400 Taurus/Sable owners

Number of brake applications

Customer usage spectra are subject to limitation of
read data collection and customer survey responses.

Sheet1

Participant	p20	b-app	p20 miles	b-app/mi.	rank	order
19	47.8	179	3.74477	0.02381	1	
7	43.5	180	4.137031	0.057823	2	
20	44.1	196	4.444444	0.091837	3	
22	22.9	125	5.456616	0.12585	4	
27	39.5	189	5.841791	0.159864	5	
4	24	140	5.833333	0.193878	6	
1	36.1	227	6.467236	0.227891	7	
15	55.8	400	7.184245	0.261905	8	
24	27.4	229	8.357684	0.285918	9	
21	27.8	234	8.478281	0.329932	10	
9	32.5	276	8.482308	0.363946	11	
16	15.8	140	8.660758	0.387959	12	
12	43.8	402	9.178082	0.431973	13	
18	37.8	358	9.470899	0.465986	14	
28	17.4	180	10.344483	0.5	15	
23	26.1	325	12.45211	0.534014	16	
13	12.8	157	12.46032	0.568027	17	
30	15.4	196	12.68234	0.502041	18	
11	20.4	269	12.69608	0.538054	19	
14	33	438	13.12121	0.570068	20	
8	51	674	13.21569	0.704082	21	
2	24.2	329	13.58504	0.788085	22	
29	18	248	13.77778	0.772109	23	
6	14.3	200	13.98801	0.808122	24	
17	33.6	498	14.82143	0.640138	25	
26	24.9	372	14.93976	0.87415	26	
10	16.9	247	16.14378	0.908163	27	
5	19.8	274	19.85507	0.942177	28	
3	19.1	436	22.82723	0.97619	29	

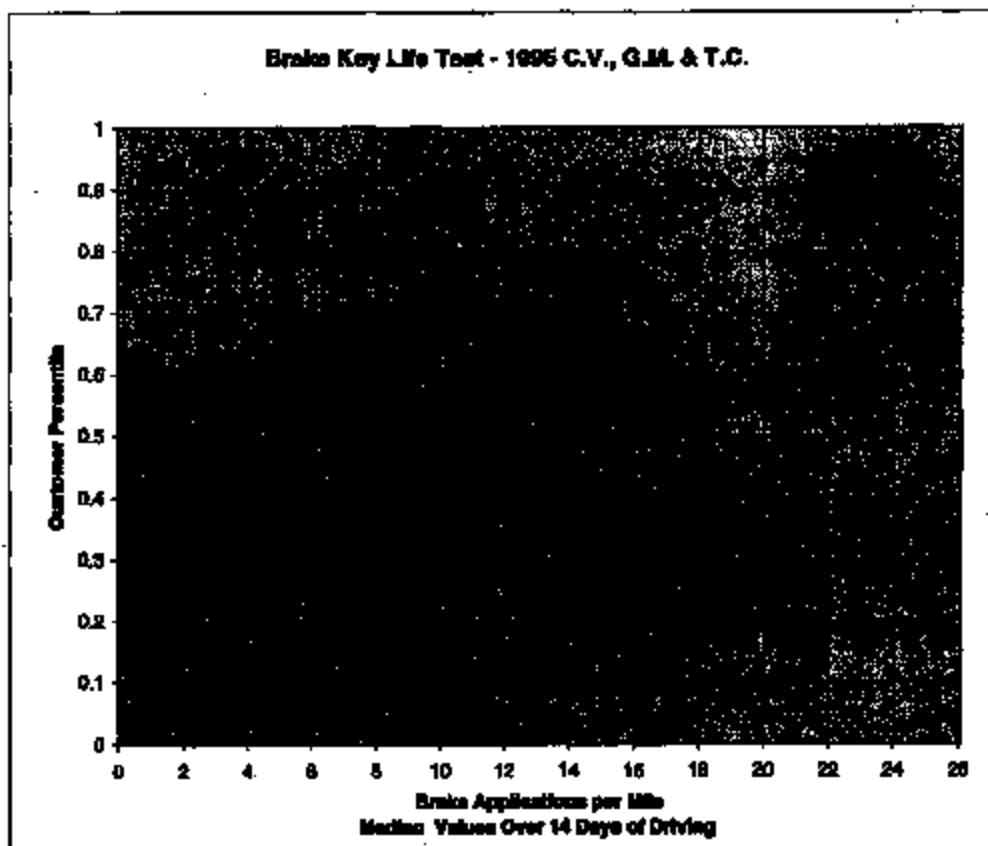
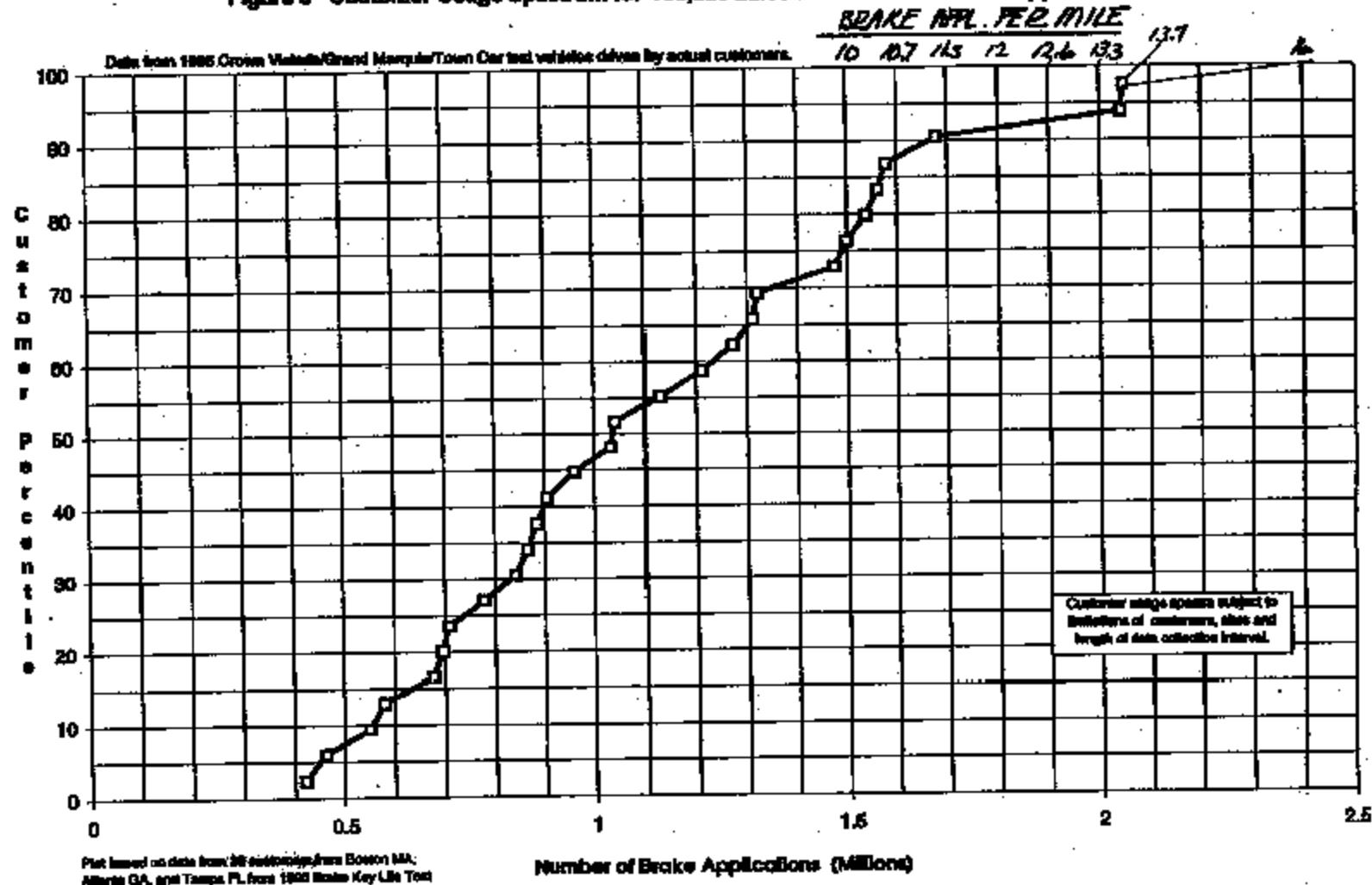
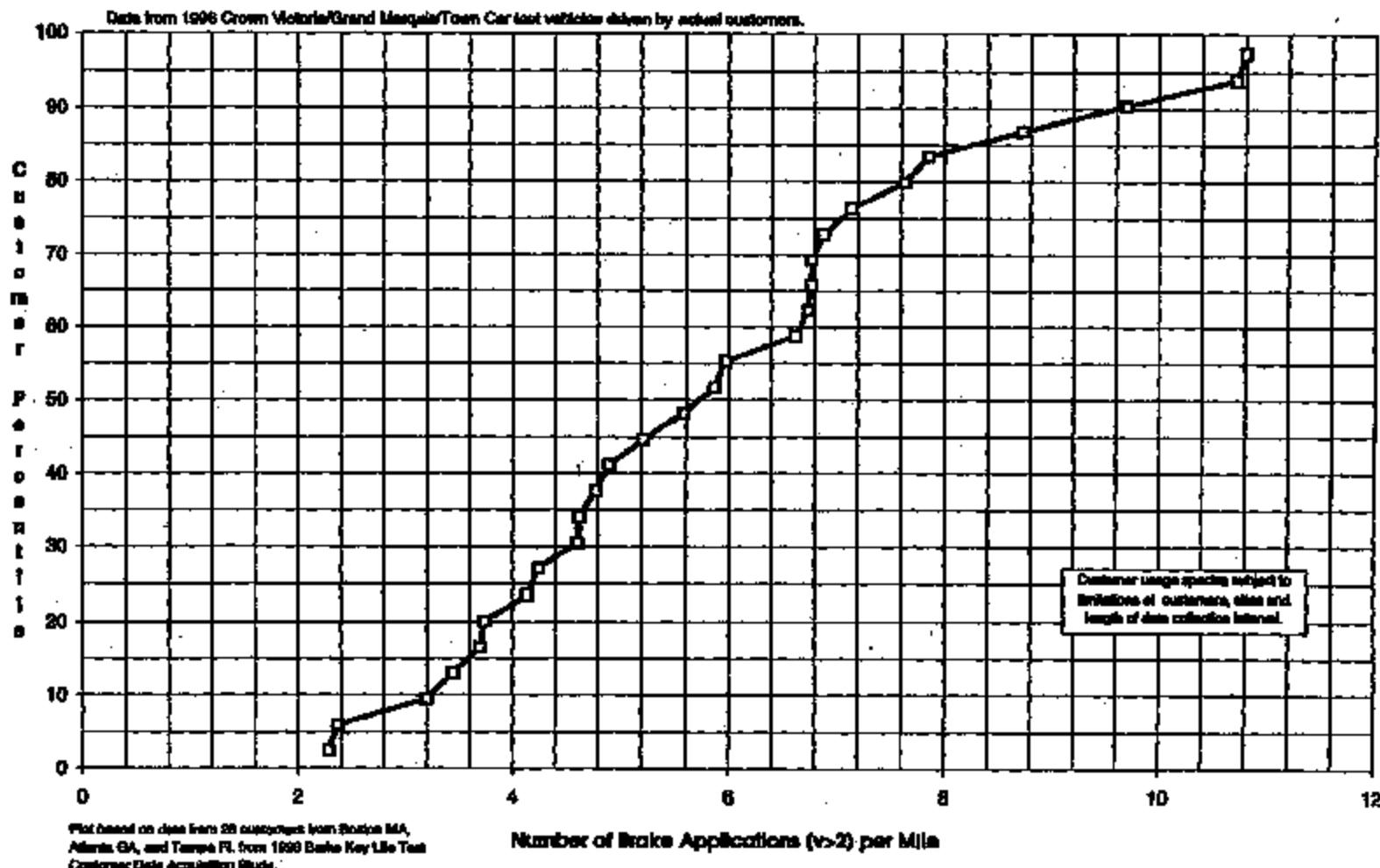


Figure 8 Customer Usage Spectrum for 150,000 Miles for Number of Brake Applications



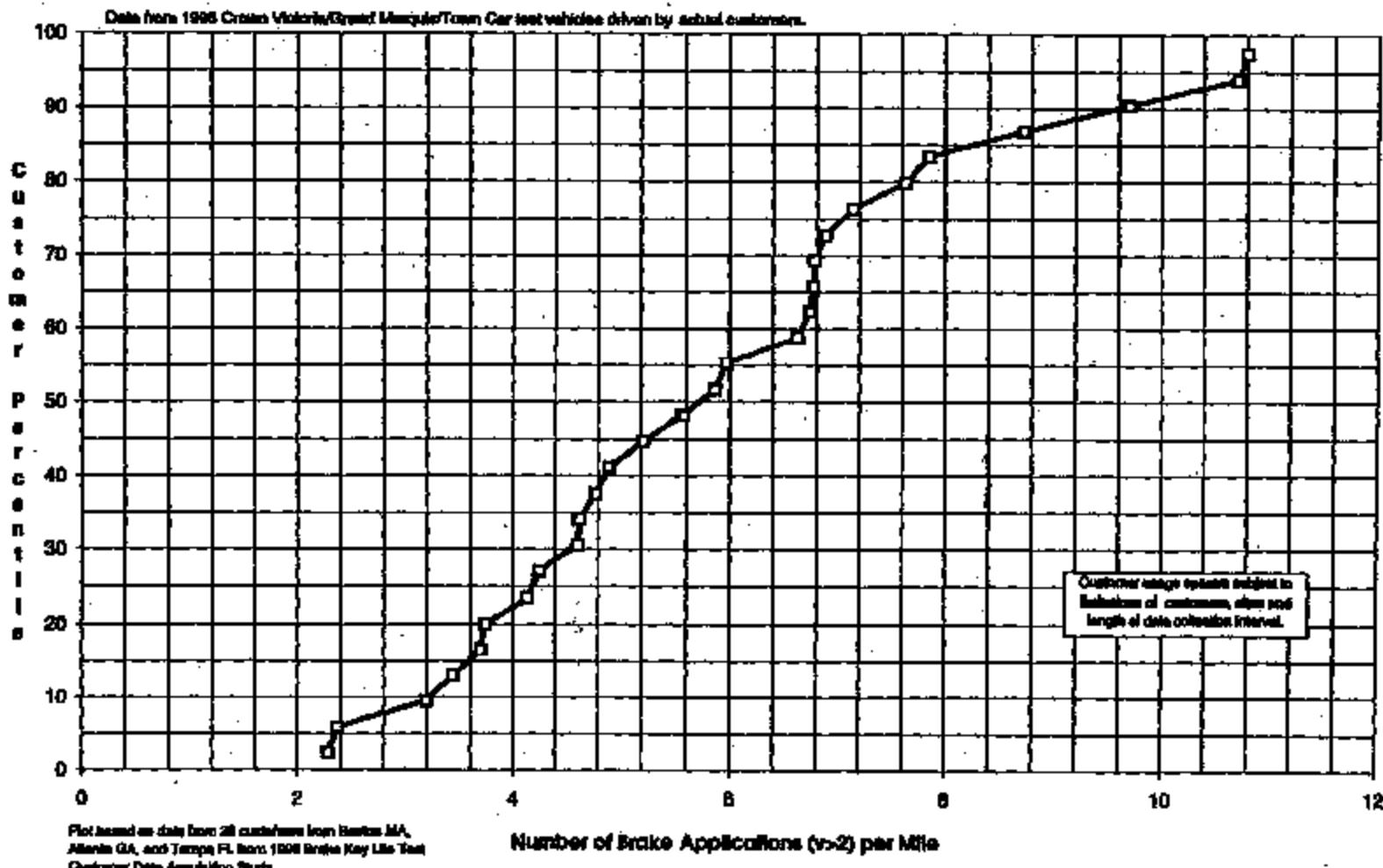
3713 7084

Figure 10 Customer Usage Spectrum for 150,000 Miles for Number of Brake Applications ($V>2$) per Mile



37137085

Figure 10 Customer Usage Spectrum for 150,000 Miles for Number of Brake Applications ($V>2$) per Mile



37137086

Figure 6 Customer Usage Spectrum for 150,000 Miles for Number of Brake Applications

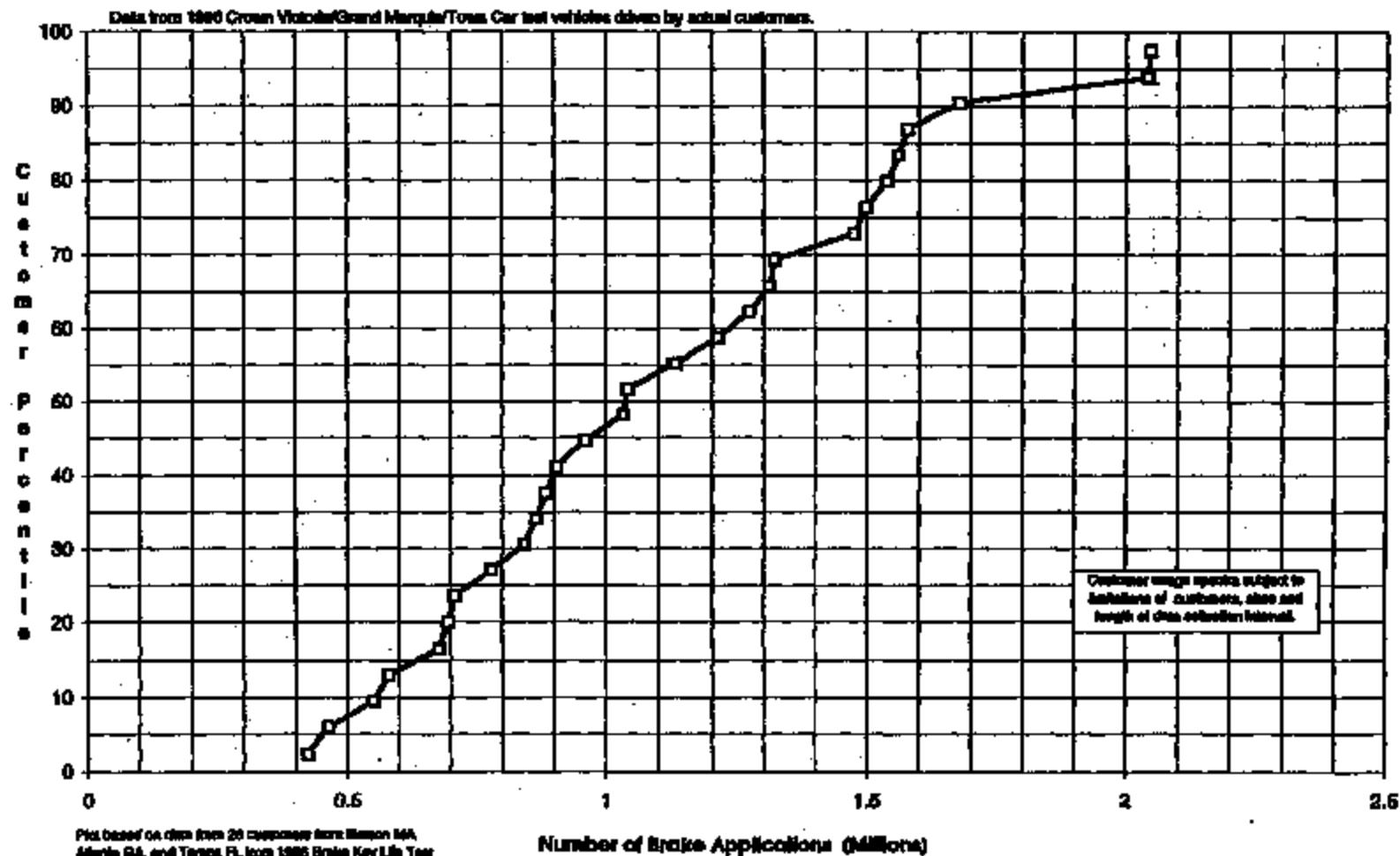
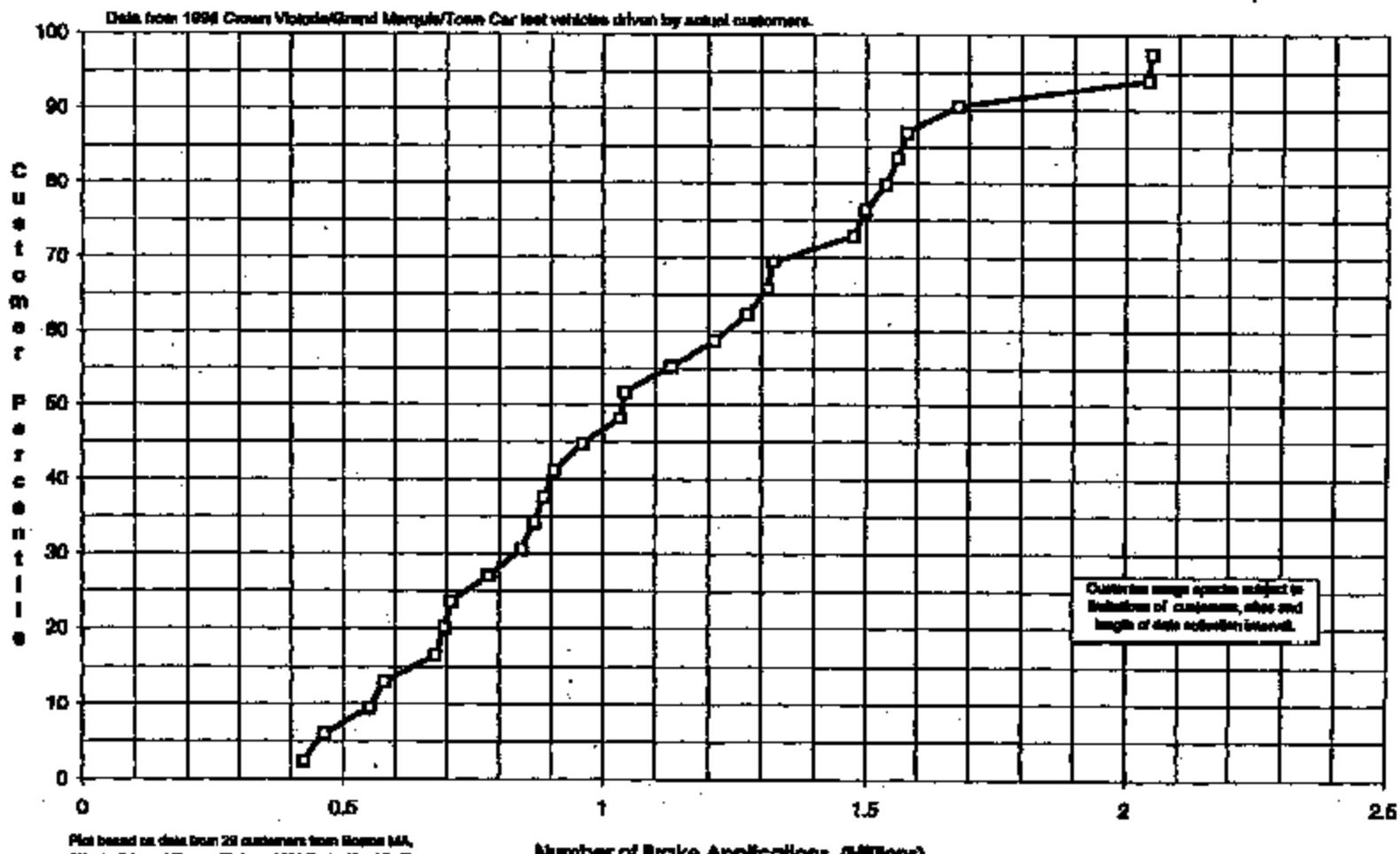


Figure 8 Customer Usage Spectrum for 150,000 Miles for Number of Brake Applications



5743 2766

2 and 3 parameter WEIBULL FAILURE ANALYSIS

CLICK here to recalculate if failure data
or "Minimum life" input value is altered

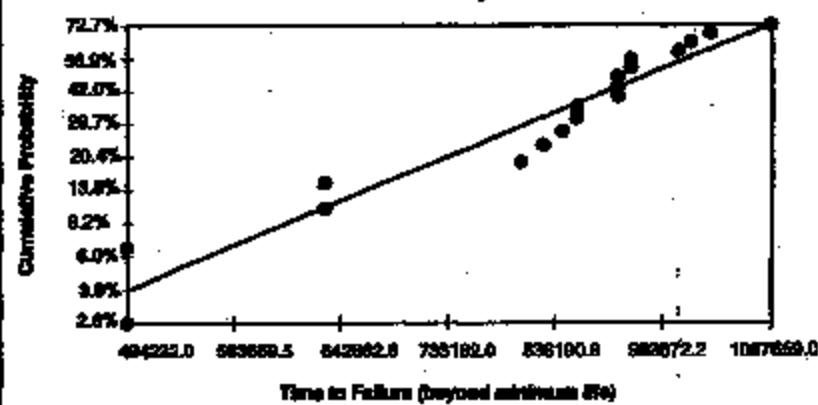
Failure No.	Time to Failure (beyond minimum life)
1	151100
2	150481
3	1525100
4	152472
5	152473
6	152474
7	152475
8	152476
9	152477
10	152478
11	152479
12	152480
13	152481
14	152482
15	152483
16	152484
17	152485
18	152486
19	152487
20	152488
21	152489
22	152490
23	152491
24	152492
25	152493
26	152494
27	152495
28	152496
29	152497
30	152498
31	152499
32	152500
33	152501
34	152502
35	152503
36	152504
37	152505
38	152506
39	152507
40	152508
41	152509
42	152510
43	152511
44	152512
45	152513
46	152514
47	152515
48	152516
49	152517
50	152518
51	152519
52	152520
53	152521
54	152522
55	152523
56	152524
57	152525
58	152526
59	152527
60	152528
61	152529
62	152530
63	152531
64	152532
65	152533
66	152534
67	152535
68	152536
69	152537
70	152538
71	152539
72	152540
73	152541
74	152542
75	152543
76	152544
77	152545
78	152546
79	152547
80	152548
81	152549
82	152550
83	152551
84	152552
85	152553
86	152554
87	152555
88	152556
89	152557
90	152558
91	152559
92	152560
93	152561
94	152562
95	152563
96	152564
97	152565
98	152566
99	152567
100	152568

Calculated Outputs

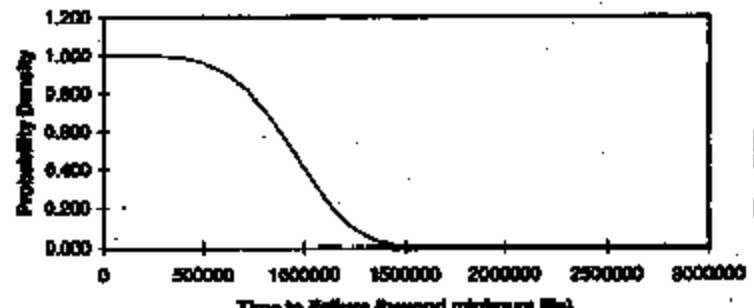
Calculated Outputs

Estimated Mean Life	661666.3561
Estimated Std Dev	151100.0000
Estimated Beta	1.8783
Estimated Alpha	1.0000

Weibull Probability Plot



Reliability Function



NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

CLICK here to recalculate if failure data
or "Minimum life" input value is altered

100.0000
200.0000
300.0000
400.0000
500.0000
600.0000
700.0000
800.0000
900.0000
1000.0000
1100.0000
1200.0000
1300.0000
1400.0000
1500.0000
1600.0000
1700.0000
1800.0000
1900.0000
2000.0000
2100.0000
2200.0000
2300.0000
2400.0000
2500.0000
2600.0000
2700.0000
2800.0000
2900.0000
3000.0000
3100.0000
3200.0000
3300.0000
3400.0000
3500.0000
3600.0000
3700.0000
3800.0000
3900.0000
4000.0000
4100.0000
4200.0000
4300.0000
4400.0000
4500.0000
4600.0000
4700.0000
4800.0000
4900.0000
5000.0000
5100.0000
5200.0000
5300.0000
5400.0000
5500.0000
5600.0000
5700.0000
5800.0000
5900.0000
6000.0000
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6200.0000
6300.0000
6400.0000
6500.0000
6600.0000
6700.0000
6800.0000
6900.0000
7000.0000
7100.0000
7200.0000
7300.0000
7400.0000
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10000.0000

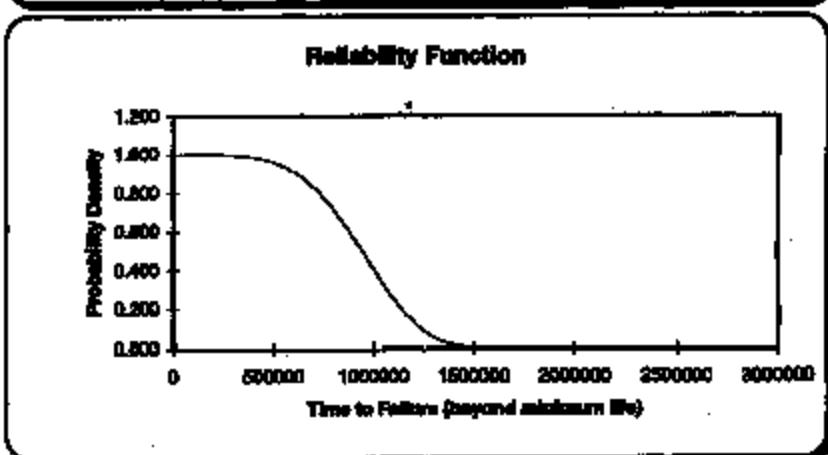
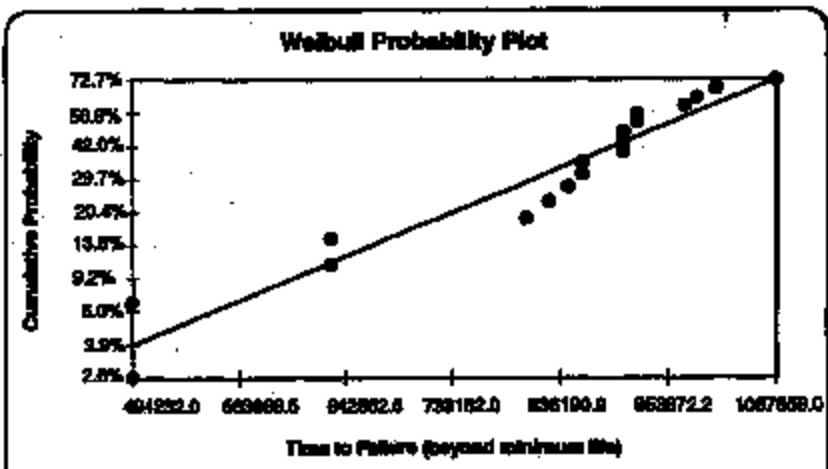
Calculated Outputs

[REDACTED]

Calculated Outputs

861666.3581

[REDACTED]



NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

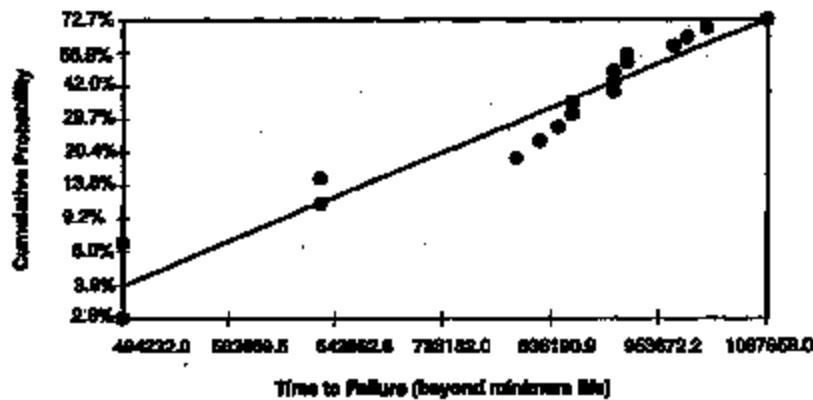
[CLICK here to recalculate if failure data or "Minimum life" input value is altered](#)

11201000
14873200
15151002
11211002
13026801
13231000
12927879
12858023
13038190
13160022
13162022
13162023
13162024
13162025
13162026
13162027
13162028
13162029
13162030
13162031
13162032
13162033
13162034
13162035
13162036
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13162038
13162039
13162040
13162041
13162042
13162043
13162044
13162045
13162046
13162047
13162048
13162049
13162050
13162051
13162052
13162053
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13162088
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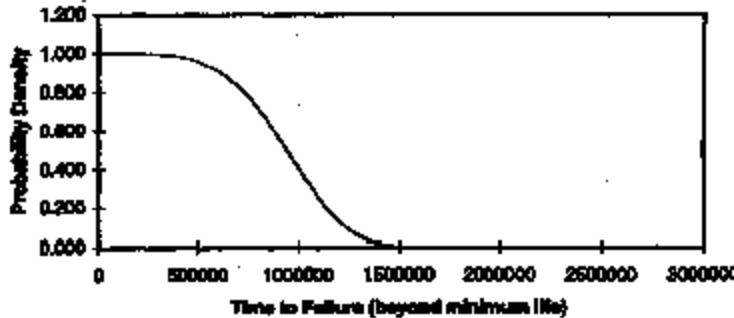
Calculated Outputs



Weibull Probability Plot



Reliability Function



NOTE:

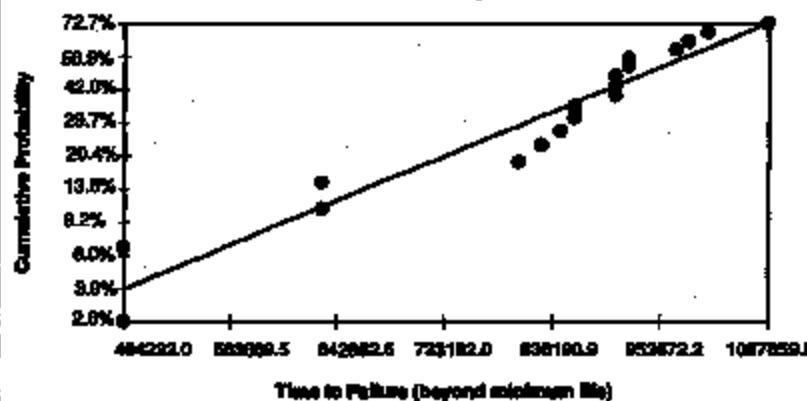
2 and 3 parameter WEIBULL FAILURE ANALYSIS

[CLICK here to recalculate if failure data or "Minimum life" input value is altered](#)

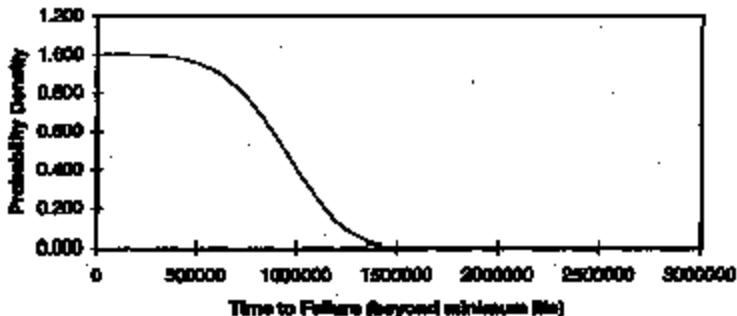
1151102	41%
1151103	
1151104	
1151105	
1151106	
1151107	
1151108	
1151109	
1151110	
1151111	
1151112	
1151113	
1151114	
1151115	
1151116	
1151117	
1151118	
1151119	
1151120	
1151121	
1151122	
1151123	
1151124	
1151125	
1151126	
1151127	
1151128	
1151129	
1151130	
1151131	
1151132	
1151133	
1151134	
1151135	
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1151137	
1151138	
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1151141	
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1151177	
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1151191	
1151192	
1151193	
1151194	
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1151196	
1151197	
1151198	
1151199	
1151100	

Calculated Outputs

Weibull Probability Plot



Reliability Function



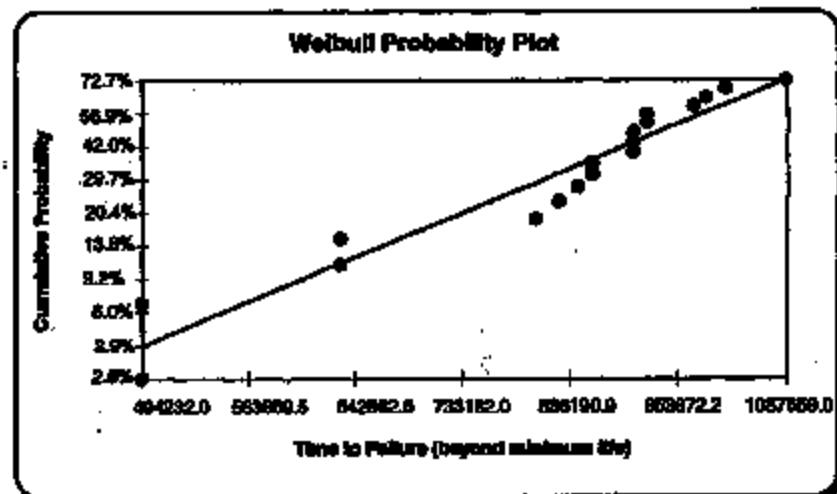
NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

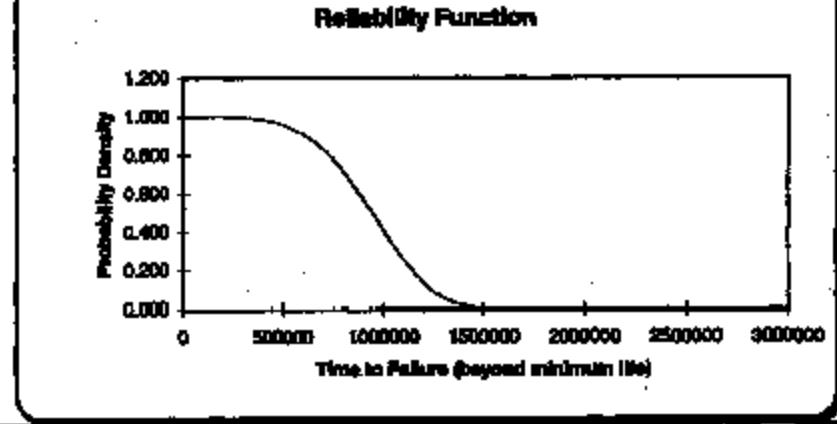
CLICK here to recalculate if failure data
or "Minimum life" input value is altered

Failure Time	1400000.0
Failure Time	1500000.0
Failure Time	1600000.0
Failure Time	1700000.0
Failure Time	1800000.0
Failure Time	1900000.0
Failure Time	2000000.0
Failure Time	2100000.0
Failure Time	2200000.0
Failure Time	2300000.0
Failure Time	2400000.0
Failure Time	2500000.0
Failure Time	2600000.0
Failure Time	2700000.0
Failure Time	2800000.0
Failure Time	2900000.0
Failure Time	3000000.0
Failure Time	3100000.0
Failure Time	3200000.0
Failure Time	3300000.0
Failure Time	3400000.0
Failure Time	3500000.0
Failure Time	3600000.0
Failure Time	3700000.0
Failure Time	3800000.0
Failure Time	3900000.0
Failure Time	4000000.0
Failure Time	4100000.0
Failure Time	4200000.0
Failure Time	4300000.0
Failure Time	4400000.0
Failure Time	4500000.0
Failure Time	4600000.0
Failure Time	4700000.0
Failure Time	4800000.0
Failure Time	4900000.0
Failure Time	5000000.0
Failure Time	5100000.0
Failure Time	5200000.0
Failure Time	5300000.0
Failure Time	5400000.0
Failure Time	5500000.0
Failure Time	5600000.0
Failure Time	5700000.0
Failure Time	5800000.0
Failure Time	5900000.0
Failure Time	6000000.0
Failure Time	6100000.0
Failure Time	6200000.0
Failure Time	6300000.0
Failure Time	6400000.0
Failure Time	6500000.0
Failure Time	6600000.0
Failure Time	6700000.0
Failure Time	6800000.0
Failure Time	6900000.0
Failure Time	7000000.0
Failure Time	7100000.0
Failure Time	7200000.0
Failure Time	7300000.0
Failure Time	7400000.0
Failure Time	7500000.0
Failure Time	7600000.0
Failure Time	7700000.0
Failure Time	7800000.0
Failure Time	7900000.0
Failure Time	8000000.0
Failure Time	8100000.0
Failure Time	8200000.0
Failure Time	8300000.0
Failure Time	8400000.0
Failure Time	8500000.0
Failure Time	8600000.0
Failure Time	8700000.0
Failure Time	8800000.0
Failure Time	8900000.0
Failure Time	9000000.0
Failure Time	9100000.0
Failure Time	9200000.0
Failure Time	9300000.0
Failure Time	9400000.0
Failure Time	9500000.0
Failure Time	9600000.0
Failure Time	9700000.0
Failure Time	9800000.0
Failure Time	9900000.0
Failure Time	10000000.0

Calculated Outputs



Reliability Function



NOTES:

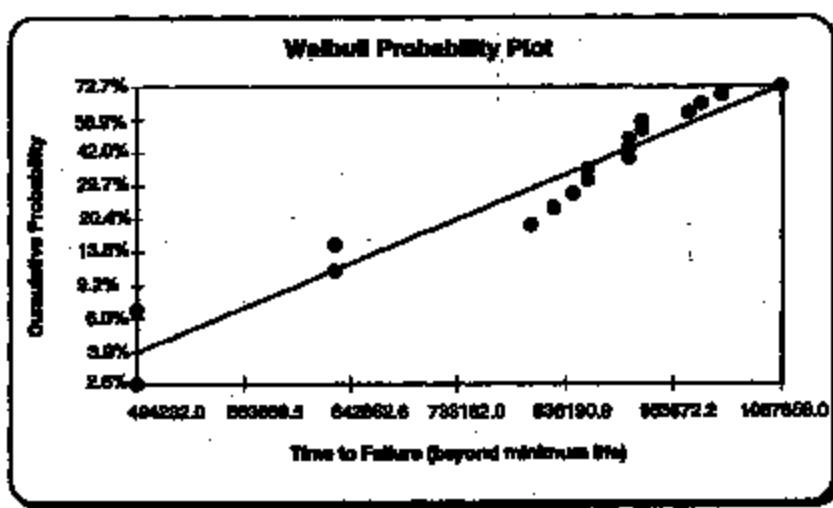
3713 7084

2 and 3 parameter WEIBULL FAILURE ANALYSIS

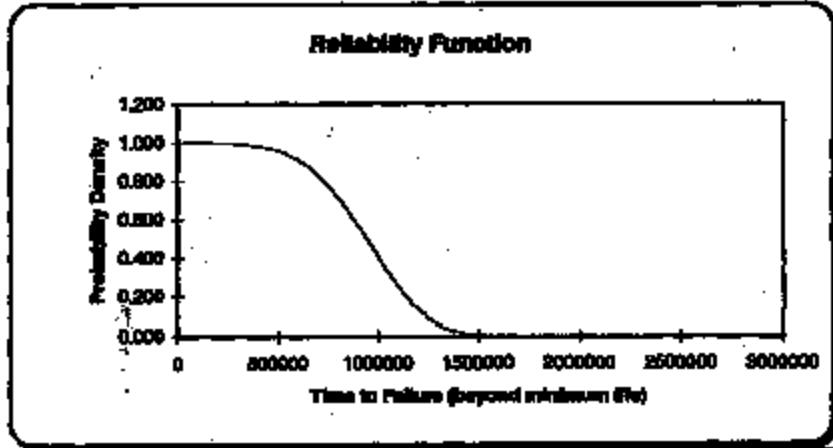
[CLICK here to recalculate if failure data
or "minimum life" input value is altered](#)

1000000
1131104
1181102
1200000
1225100
1244173
1250000
1266000
1273000
1280000
1286000
1292000
1298000
1304000
1310000
1316000
1322000
1328000
1334000
1340000
1346000
1352000
1358000
1364000
1370000
1376000
1382000
1388000
1394000
1400000
1406000
1412000
1418000
1424000
1430000
1436000
1442000
1448000
1454000
1460000
1466000
1472000
1478000
1484000
1490000
1496000
1502000
1508000
1514000
1520000
1526000
1532000
1538000
1544000
1550000
1556000
1562000
1568000
1574000
1580000
1586000
1592000
1598000
1604000
1610000
1616000
1622000
1628000
1634000
1640000
1646000
1652000
1658000
1664000
1670000
1676000
1682000
1688000
1694000
1700000
1706000
1712000
1718000
1724000
1730000
1736000
1742000
1748000
1754000
1760000
1766000
1772000
1778000
1784000
1790000
1796000
1802000
1808000
1814000
1820000
1826000
1832000
1838000
1844000
1850000
1856000
1862000
1868000
1874000
1880000
1886000
1892000
1898000
1904000
1910000
1916000
1922000
1928000
1934000
1940000
1946000
1952000
1958000
1964000
1970000
1976000
1982000
1988000
1994000
2000000

Calculated Outputs



Reliability Function



NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

[CLICK here to recalculate if failure data
or "Minimum life" input value is altered](#)

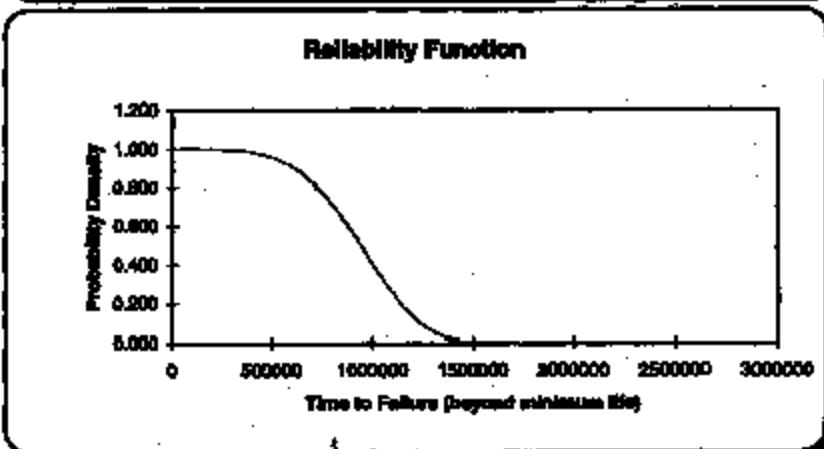
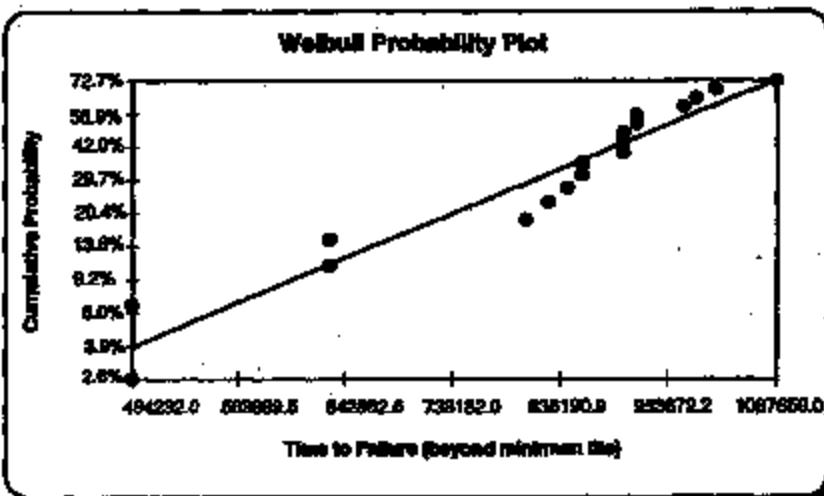
Failure Time	Probability
494232.0	0.00%
528889.5	100%
542882.6	100%
738182.0	100%
836190.9	100%
928672.2	100%
1087866.0	100%

Calculated Outputs

Estimated Mean:
1,531,032

Estimated Std Dev:
100%

Estimated Beta:
1022000.407



NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

CLICK here to reanalyze 2 failure data
or "Minimum life" input value is altered

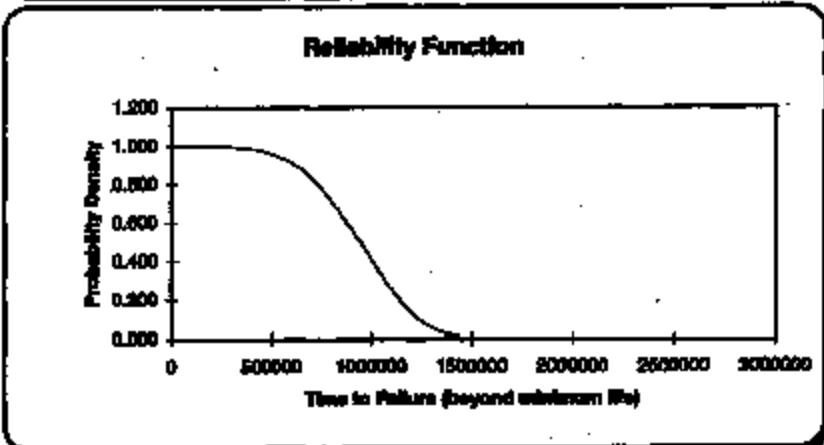
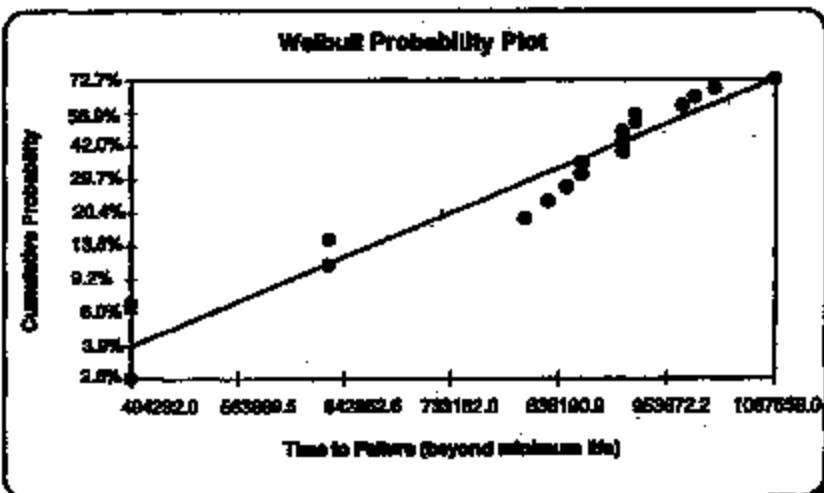
Failure No.	Time to Failure (hrs)	Failure Type
1	404282.0	Normal
2	553899.5	Normal
3	642828.6	Normal
4	735182.0	Normal
5	836190.0	Normal
6	953872.2	Normal
7	1067658.0	Normal

Calculated Outputs

Estimated Parameters

Estimated Parameters

Estimated Parameters



NOTES:

2 and 3 parameter WEIBULL FAILURE ANALYSIS

[CLICK here to recalculate if failure data
or "Minimum life" input value is altered](#)

Failure Data	Time to Failure (days)	Cumulative Probability
1	494232.0	0.00%
2	503899.5	10.00%
3	642862.4	20.00%
4	733182.0	30.00%
5	838190.9	40.00%
6	953672.2	50.00%
7	1057059.0	60.00%
8	1115347.613	70.00%
9	1183636.226	77.78%
10	1251924.839	85.56%
11	1319213.452	93.33%
12	1386502.065	100.00%

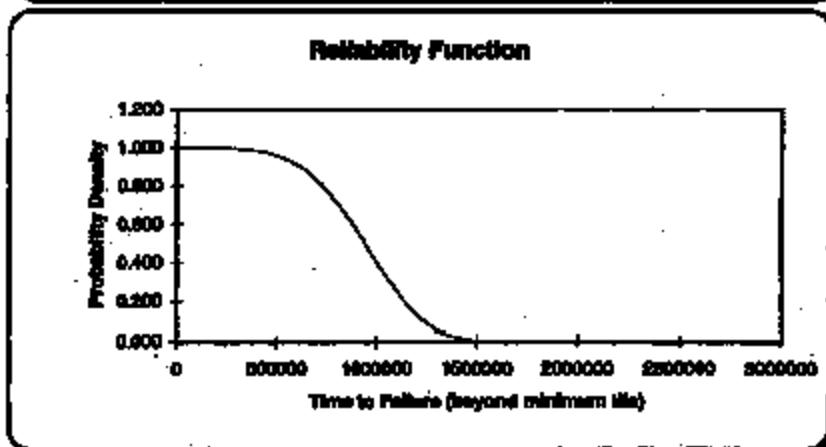
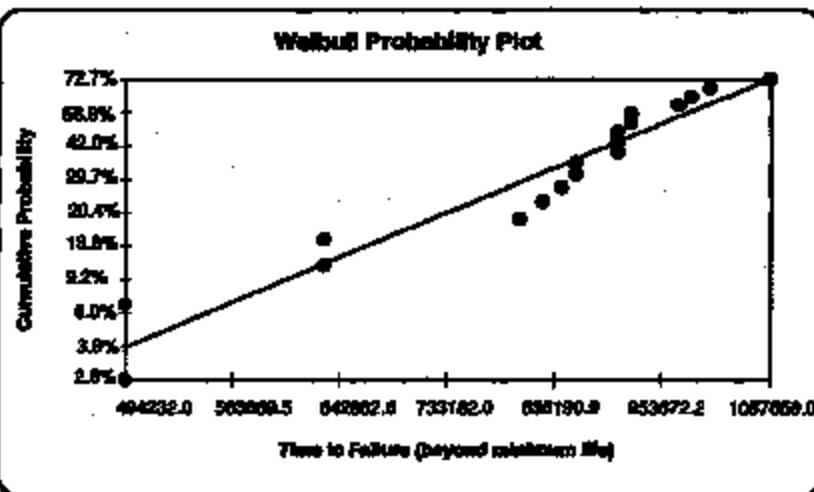
Calculated Outputs



Calculated Outputs

Calculated Outputs

Calculated Outputs

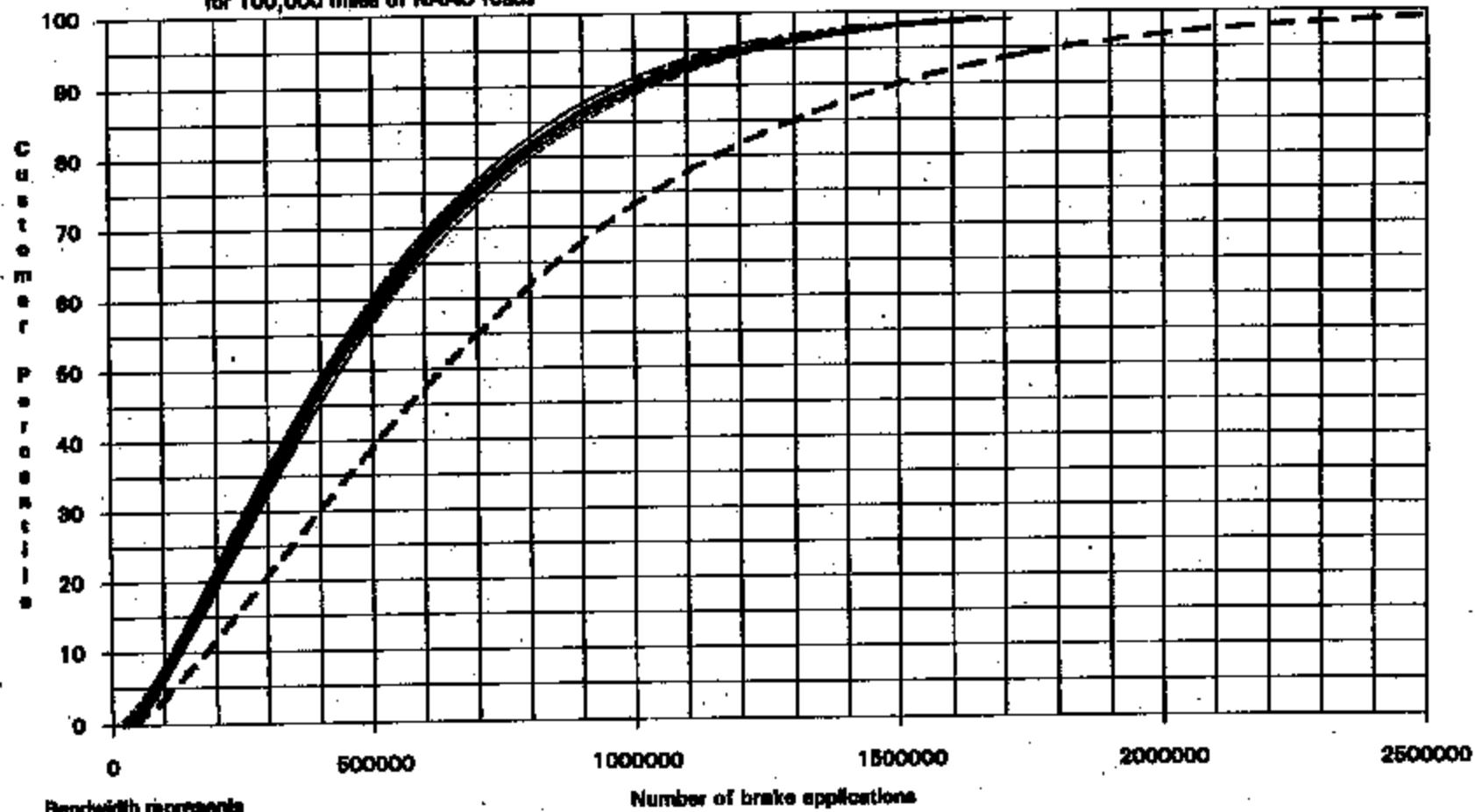


NOTES:

1991 Thunderbird 5.0L, AOD
NAAO Customer Usage Spectrum

Heavy solid curve represents the
average of 9 Monte Carlo projections
for 100,000 miles of NAAO roads

Heavy dashed curve represents
the 160,000 mile projection



Bandwidth represents
variability of 9 Monte Carlo
projections for 744
Thunderbird owners

AGP/TWG W97/XLC 05/94

3713 7069

* Note printed by NLAPPOINT on 19 Apr 1999 at 15:15:08 *

From: FPORTER --DEBN007
To: SLAROUCH--FORDMAIL
cc: NLAPPOINT--DEBN005
PELAAS --FORDMAIL

Date and time 04/16/99 20:15:11
SREIMERS--DEBN007

FROM: F. J. Porter
Subject: Brake Switches

USANT(UTC -04:00)

For clarification:

When you are done with the testing, do the switches from EAA show signs of a cell being formed?

Is there, or could there be, evidence of corrosion occurring on the EAA switches similar to the junkyard switches.

Regards,

Fred Porter OV - fporter fporter@ford.com
Chassis E/E Systems Applications (313) 845-3722
Bldg 5 - Mail Drop 5030 - Cubicle 3E004 fax: 390-4145
*** Forwarding note from SLAROUCH--FORDMAIL 04/16/99 17:14 ***
To: FPORTER --FORDMAIL Porter, Fred (F.J.
cc: NLAPPOINT--FORDMAIL LaPointe, Norman (PELAAS --FORDMAIL Klaas, Pete (P.P.)
SLAROUCH--FORDMAIL LaRouche, Steve (S

From: LaRouche, Steve (S.)
Subject: Brake Switches

Fred: I had a meeting this afternoon with two gentlemen (one was Ed Carter) from Scientific Research Laboratories to discuss brake switches. I showed them the results I have so far, and they felt that our testing was pretty comprehensive and that there was really not any more that they could contribute in terms of testing. We tried to come up with possible scenarios that would tie our findings to a cause of fire, but couldn't come up with anything. Here is a quick summary of our findings at Central Lab:

There appear to be two modes of failure occurring: One involving leakage of brake fluid through the Kapton seal and an apparent cell being set up between the contacts and steel cup. The other involving ingressation of water into the switch cavity, with no brake fluid leakage, and no evidence of a cell.

The Memphis switch and all of the leakers analyzed so far show a leak path through the Kapton seals. The cup faces show transfer of the brass contact material to them which suggests that a cell has occurred between the hot contacts and the grounded cup. In addition to brake fluid, the Memphis switch shows evidence (desiccification of the brass contact) that some moisture may have also been present. We found no evidence that road salt had entered the switches.

The completely burned switches also show probable transfer of contact material to the cups, indicating a possible cell. No evidence of road salt detected. Could not determine if brake fluid leakage occurred.

Three of the junk yard switches (including the one analyzed by SRL) showed corrosion of the cups suggesting ingressation of water into the

switch cavities. This appears to be a different mode of failure in that there was no evidence of a gall occurring between the contacts and cup. Again, there was no evidence of road salt in the switch cavities. Although there is some damage to the Kapton seals, there appears to have been no leak path or leakage of brake fluid.

I have received two switches from the OASIS which were both leakers. Testing is pretty much complete on these switches and so far we have found nothing different from the other leakers we analyzed.

I have also received three switches from HAA which were completely burned. These switches are in various stages of analysis, but so far do not appear to be different from the completely burned switches we analyzed previously.

The brake fluids in the Memphis switch and all the leakers (including those from the OASIS) contain oxalates. The brake fluid from the car you have out at MPG does not. The guys from GM suggested that we analyse brake fluids from old vehicles for oxalates and other contaminants, as well as measure conductivity. So far, we have received no brake fluid samples for this.

This is what we have so far: We have found several conditions which may have contributed to a fire, but have not been able to link any of them to a definite cause. I don't think that additional testing on switches is going to be beneficial. I would like to complete the testing that is in progress and wrap this up. Let me know how you want us to proceed.

Steve LaRouche (SLAROUCH)
Metallurgy Section, Central Laboratory, Room W410
(313) 845-4876 (313) 322-1614 FAX

* Note printed by NLAPPOINT on 16 Apr 1999 at 09:02:14 *

From: SLAROUCH--FORDMAIL Date and time 03/18/99 16:59:33
To: NLAPPOINT--FORDMAIL LaPointe, Norman (

From: LaRoucha, Steve (S.)
Subject: FW: Brake Switch Analysis - Junk Yard Part

Steve LaRoucha (SLAROUCH)
Metallurgy Section, Central Laboratory, Room N410
(313) 845-4576 (313) 322-1614 FAX

-----Original Message-----

From: Shaun McCarthy mailto:smccart3@gw.ford.com
Sent: Thursday, March 18, 1999 3:47 PM
To: fporter@gw.ford.com
Cc: smccart3@gw.ford.com; slarouch@mail.ford.com;
rcarter@pobox.srl.ford.com
Subject: Brake Switch Analysis - Junk Yard Part

We have disassembled the switch you gave me that showed corrosion inside the switch area. Inorganic analysis of the material has been completed. The reddish material is FeO(OH) and the whitish material is 3ZnCO₃-3Zn(OH)₂. Extensive photos have been taken through out the disassembly. Our conclusion by examining the path or location of corrosion is that fluid entered through the connector area for this sample. We would be glad to meet with you and show you the photos and discuss our conclusions.

Analysis is still underway regarding the material from the "Memphis" switch.

Regards,
SHAUN McCarthy SRL Room 1339 Mail Stop 1170
32-21358 FAX 32-31129

Data Log
Brake Pressure Switch

Last Updated 03/09/08

Inv ID	Inv Date/Code	Vehicle	VIN	Event	Mileage	Turn-Hesitate Resistance	Lastest Kapton #1	Kapton #2	Kapton #3	Preset Status	Parts Received
Memph	2009	Town Car	PY	Svc. Fls				crack	crack	crack	Analyze Complete
A	2001	Town Car	PY	Underhood Fls				no info	no info	no info	Analyze Complete
B	2114	Town Car	NY	Underhood Fls							Inv. not available
C	2003	Town Car	NY	Underhood Fls							Analyze In Progress
D		Crown Vic Police Car	VX	Crash Imp		4.0MEGAohm	yes	crack	crack	crack	Analyze Complete
E	2137	Town Car	NY	Reference		OPEN	no	worn	worn	worn	Analyze Complete
F	2185	Town Car	NY	Crash Imp		4.0MEGAohm	yes	crack	worn, no crack	worn, no crack	Analyze Complete
1	2000	Town Car	NY	Reference	79184	OPEN					Analyze Complete
2	2016	Town Car	PY	Reference	71307	OPEN					Analyze Complete
3	2040	Town Car	PY	Reference	68037	OPEN					Analyze Complete
4	2004	Town Car	PY	Reference	68046						Analyze In Progress
5	2005	Town Car	PY	Reference	47305						Analyze In Progress
6	2009	Town Car	NY	Reference	68002	OPEN					Analyze In Progress
7	2005	Town Car	PY	Reference	68014	OPEN					
8	2005	Town Car	PY	Reference	77						
9	2000	Town Car	PY	Reference	68024	OPEN					
10	2001	Town Car	PY	Reference	61388	OPEN					
11	2008	Town Car	PY	Reference	68038	OPEN					
12	2000	Town Car	NY	Reference	68037	OPEN					
13	2071	Town Car	PY	Reference	64145	OPEN					
14	2001	Town Car	PY	Reference	77	OPEN					
15	77	Town Car	NY	Reference	67169	OPEN					
16	77	Town Car	PY	Reference	72114	OPEN					
17	2008	Town Car	PY	Reference	68046	OPEN					
18	77	Town Car	PY	Reference	67649	OPEN					
19	77	Town Car	PY	Reference	48021	OPEN					
20	77	Town Car	PY	Reference	48021	OPEN					
21	2074	??	??	Reference	77						
22	77	Town Car	PY	Reference	68062	OPEN					
23	77	Town Car	PY	Reference	77						
From TX trip of 2/19 to 2/12, John McIntyre Group											
1	2008	Town Car	PY	Reference							
2	2002	Crown Vic Police Car	PY	Reference							
3	2006	Grand Marquis	PY	Reference	77	100000					
4	2005	Crown Vic	PY	Reference		48042					
5	2003	Town Car	PY	Reference		73115					
6	77	Town Car	NY	Underhood Fls	77						
7	2001	Town Car	PY	Reference	77						

Steve Rollins, 313-39 03286,
steve@ford.com,
http://fordtrucks.com

Page 1 of 2
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revised 2/16/09

3713 7133

Data Log
Brake Pressure Switch

8	2046	Town Car	NY	Underhood Firs	106810						
9	3068	Town Car	PY	Reference	??						
10	2272	Crown Vic	PY	Reference	72614						
11	2115	Town Car	NY	Reference	??						
12	3065	Town Car	PY	Reference	??						
13	3069	Town Car	PY	Reference	106845						
OADS											
Baton Rouge, LA	2062	Town Car	PY	Dealership Return	85005						
Memphis, TN	2128	Town Car	PY	Dealership Return	106865	yes	crack	crack	crack	Analysis In Progress	Switch
EAA											
Davenport, FL	2060	Town Car	NY	Underhood Firs	100000+						
Aurora, IL	2069	Town Car	PY	Underhood Firs	?	?					
Naples, FL	?	Lincoln	PY	Underhood Firs	75005	?	crack	crack	crack	Analysis In Progress	Switch
											Switch, #C2.15A Plus Brake Fluid Only

Steve Polkner, 313-55 02660,
steve@fordcars.com,
See SwitchLog.xls

page 2 of 2
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revised 2/16/00

87137104

C - COMPLETED
NA - NOT APPLICABLE
TSP - TO BE PERFORMED

Brake Switch Test

NP = INFINITY (OPEN)
NP = NOT PERFORMED
NRDL = NOT REC'D AT CEN. LAB.

Part No.	Description	PTC	Monitoring						
			A	B	C	D	E	F	G
	01. Bag Pre-tensioner pretensioning logic	C	C	C	C	C	C	C	C
	02. Ignition	P/RE	P/RE	P/RE	P/RE	NP LEAK	NO PNL/LEAK	NP LEAK	NO PNL/LEAK
	03. Photographic switch	C	C	C	C	C	C	C	NP
	04. Front/rear seat belt alarm interconnection	C	C	C	C	C	C	C	C
	05. Check for Occupant engagement	C	NA	NA	NA	NA	NA	NA	NA
	06. Safety belt reminder	C	NA	NA	NA	NA	NA	NA	NA
Switch + Connector Assembly	07. Brake Master Pressure Monitoring	NA	NA	NA	NA	NA	NA	NA	NA
	08. Safety belt reminder function	NA	NA	NA	NA	NA	NA	NA	NA
	09. Accelerator Pedal Position	NA	NA	NA	NA	NA	NA	NA	NA
	10. Accelerator Pedal Position Performance	NA	NA	NA	NA	NA	NA	NA	NA
	11. Accelerator Pedal Position Function	NA	NA	NA	NA	NA	NA	NA	NA
Connector Only	12. Accelerator Pedal Position Function	C	NA	NA	C	NA	NA	NA	NA
	13. Accelerator Pedal Position Function	C	NA	NA	C	NA	NA	NA	NA
	14. Accelerator Pedal Position Function	NA	NA	NA	NA	NA	NA	NA	NA
	15. Accelerator Pedal Position Function	NA	NA	NA	NA	NA	NA	NA	NA
Switch Controlled by Computer	16. Accelerator Pedal to Headlight Blend	NA	NA	NA	NA	NA	NA	NA	NA
	17. Safety Beam to Headlight Headlight Function	NA	NA	NA	NA	0.4	0.8	NP	0.3
	18. Safety Beam to Headlight Function	NA	NA	NA	NA	4.0M	NP	2M	NP
	19. Safety Beam to Headlight Function	NA	NA	NA	NA	NP	NP	NP	NP
	20. Safety Beam to Headlight Function	NA	NA	NA	NA	1.1	0.0M	NP	0.0M
Switch Controlled by Computer	21. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NA
	22. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NA
	23. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NA
	24. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NA
Switch	25. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NP
	26. Brake Master Pressure	NA	NA	NA	NA	NA	NA	NA	NP
	27. Front Seat for Lockout	NA	NA	NA	NA	NA	NA	NA	NP LEAK
	28. Required Items 17. Summary of TSP only	NA	NA	NA	NA	NA	NA	NA	C
		NA	NA	NA	NA	NA	NA	NA	NP
		NA	NA	NA	NA	NA	NA	NA	NP
		NA	NA	NA	NA	NA	NA	NA	NP
		NA	NA	NA	NA	NA	NA	NA	NP
Switch	29. Emergency Stop	C	C	NP	C	C	C	C	NP
	30. Emergency Stop, Photogate	C	C	NP	C	C	C	C	NP
	31. Photo eye	C	C	NP	C	C	C	C	NP
	32. Emergency Stop, Photogate	C	C	NP	C	C	C	C	NP
Tachograde	33. Accelerator踏板踏板, Accelerator踏板踏板	C	C	NP	C	C	C	C	NP
	34. Accelerator踏板踏板, Accelerator踏板踏板	C	C	C	C	C	C	C	NP
	35. Accelerator踏板踏板, Accelerator踏板踏板	C	C	C	C	C	C	C	NP
	36. Accelerator踏板踏板, Accelerator踏板踏板	C	NA	NP	C	C	C	C	NP
	37. Accelerator踏板踏板, Accelerator踏板踏板	C	NA	NP	C	C	C	C	NP

C = COMPLETE
NA = NOT APPLICABLE
TIP = TO BE PERFORMED

Brake Switch Test Checklist

INF = INFINITY OPEN
NP = NOT PERFORMED
NRCLL = NOT READ AT OEM LAL.

	3	4	5	6	7	8	9	10
	PY 2005	PY 2006	PY 2007	PY 2008	PY 2009	PY 2010	PY 2011	PY 2012
Tool Info								
	1. Any tools with low system logics	O	O	O	O	O	O	O
	2. Oxygen	NO FIRE/LEAK						
	3. Hydrogen Sulfide	NP						
	4. Ethanol or Ethylene Propylene	C	C	C	O	O	O	G
	4. Ethanol or Ethylene Propylene	NA						
	5. Safety T-T System	NA						
Switch + Connector								
	6. Power to Military Personnel Protection	NA						
	7. Power Transfer to Hospital Personnel	NA						
	8. Power Transfer to Hospital Personnel	NA						
	9. Power Transfer to Hospital Personnel	NA						
Connector Only								
	10. Power Transfer to Hospital Personnel	NA						
	11. Power Transfer to Hospital Personnel	NA						
	12. Check for full engagement of connector	NA						
	13. Check seal requirement	NA						
	14. Check integrity seals	NA						
	15. Full seal requirement for check for sealability	NA						
Switch Controlled Unplugged								
	16. Reassembly straight to Challenge Board	O	D	NA	NA	C	C	C
	17. Power Transfer to Military Personnel Protection	NA						
	18. Power Transfer to Hospital Personnel	INF	INF	NA	NA	INF	INF	INF
	19. Power Transfer to Hospital Personnel	INF	INF	NA	NA	INF	INF	INF
	20. Power Transfer to Hospital Personnel	11.4	1.0M	NA	NA	7.0M	7.0M	1.0M
Switch Controlled Plugged								
	21. Double Grounding Protection	134	180	NA	NA	147	120	140
	22. Double Cleaning Protection	30	60	NA	NA	70	60	112
	23. Seal Test for Leakage	NO LEAK	NO LEAK	NA	NA	NO LEAK	NO LEAK	NO LEAK
	24. Sealed Plugs 17.5mm length 200 at 1000psi	O	O	NA	NA	O	O	C
Switch								
	25. Double Cleaning when dry	INF	INF	NA	NA	INF	INF	INF
	26. Double Cleaning surfaces. Photograph	INF	INF	NA	NA	INF	INF	INF
	27. Double Cleaning	INF	INF	O	O	INF	INF	INF
	28. Double Cleaning surfaces. Photograph	INF	INF	O	O	INF	INF	INF
Techniques								
	31. Visual-Inspect visual check	INF	INF	C	C	INF	INF	INF
	32. Visual-Inspect surfaces. Photograph	INF	INF	C	C	INF	INF	INF
	33. Visual-Inspect surfaces. Photograph	INF	INF	O	O	INF	INF	INF
	34. Visual-Inspect visual check	INF	INF	O	O	INF	INF	INF
	35. Visual-Inspect visual check. Report, written note, etc.	INF	INF	NA	NA	INF	INF	INF
	36. Visual-Inspect visual check. Report, written note, etc.	INF	INF	NA	NA	INF	INF	INF
	37. Visual-Inspect visual check.	INF	INF	NA	NA	INF	INF	INF
	38. Inspect condition of equipment or setting	INF	INF	NA	NA	INF	INF	INF

C = COMPLETE
NA = NOT APPLICABLE
TBP = TO BE PERFORMED

Brake Switch Test Procedure

INF = INFINITY (OPEN)
NP = NOT PERFORMED
NRCLD = NOT READ AT GEN. LBL.

	11	12	13	14	15	16	17	18
	PV	P	NP	P	PV	NP	PV	P
Part Info	1. Log P/N# into Initial Logbook	C						
	2. Drain	NO FRIE/LEAK						
	3. Perform leak test.	NP						
	4. Thread any loose threads down after test.	O						
	5. Check for O-ring/gasket configurations	NA						
	6. Check for O-ring/gasket configurations	NA						
	7. Log O-Ring/gasket configurations	NA						
	8. Log O-Ring/gasket configurations	NA						
Switch + Connector Assembly	9. Wire 12VDC to 12VDC Power Source	NA						
	10. Wire 12VDC to Heater Resistor	NA						
	11. Wire 12VDC to Heater Resistor	NA						
	12. Separate Resistor from Heater	NA						
Connector Only	13. Verify Connector Seal	NA						
	14. Wire 12VDC to Wire 12VDC Power	NA						
	15. Check for seal engagement of connector	NA						
	16. Check wire insulation	NA						
	17. Check wire gauge	NA						
	18. Cut wire insulation to check for connector	NA						
Switch Operational Implementation	19. Insertable Switch to Operational State	O						
	20. Bring Ventilator System to Operational System	NP						
	21. Verify Ventilator is Proper Readiness	NP						
	22. Verify Ventilator is Proper Readiness	NP						
	23. Start at Max Fan Performance	NP						
Switch	24. Switch/Opening Pressure	NP						
Controlled	25. Switch/Closing Pressure	T1						
Pressurized	26. Press Test for Leakage	NO LEAK						
	27. Report Status 17 through 20 w/ initial info	C						
		NP						
		NP						
		NP						
		NP						
Switch	28. Perform physical testing	NP						
	29. Inspect control surface, Plugs/leads	NP						
	30. Perform op.	NP						
	31. Inspect control surface, Plugs/leads	NP						
Test Sequence	32. Perform physical testing	NP						
	33. Inspect control surface, Plugs/leads	NP						
	34. Perform physical testing, Plugs/leads	NP						
	35. Inspect control surface, Plugs/leads	NP						
	36. Log all findings of inspection or testing	NP						

O = OBTAINABLE
NA = NOT APPLICABLE
TIP = TO BE PERFORMED

Brake Switch Test Checklist

INF = INFINITY (OPEN)
NP = NOT PERFORMED
NRCLB = NOT READ AT CRNL LBL

Test Info	19	20	21	22	23	1	2	3
	P1	P2	P2A	P2	P2A	P1	P2	P2A
1. Eng Field into Min Brake Length						C	C	C
2. Connect						NO FRIELEAK	NO FRIELEAK	NO FRIELEAK
3. Throughput Brake						C	D	C
4. Ensure no unusual external visual phenomena						C	C	C
5. Check for Overdrive engagement						C	C	C
6. Verify 2nd gear position						NA	C	NA
Switch + Connector Assembly	8. After 10.000km When INGRANDE Position					9.5	8.4	0.2
	9. After 10.000km to Normal Position					INF	INF	INF
	10. After 10.000km to Handbrake Position					INF	INF	INF
	11. After 10.000km to Shift Lever Position					C	C	C
Connector Only	12. Verify Neutral Lever Gated					O	C	C
	13. Verify Overdrive Gear					O	C	C
	15. After 10.000km to Shift Lever Position					INF	INF	INF
	16. Check for full engagement of overdrive					O	C	C
	18. Check when Brake Applied					O	C	C
	19. Check when gears applied					C	C	C
	20. Check when handbrake applied					TIP	TIP	TIP
Switch External Powerfeed	10. Verify Neutral by Overdrive Lever					O	C	C
	12. Verify Neutral in preliminary Standard Position					9.5	8.1	0.2
	18. Verify Standard in Manual Position					INF	INF	INF
	19. Verify Standard to Handbrake Position					INF	INF	INF
	20. Check in Handbrake Position					9.5	INF	17.7K
	21. Check in Neutral Position							
	22. Check in Standard Position							
	23. Check in Handbrake Position							
	24. Check for Leakage							
	25. Check after 100 km trip							
	26. Check after 1000 km trip							
	27. Check after 10000 km trip							
	28. Check after 100000 km trip							
Technique	29. Visual inspection along day					156	157	158
	30. Inspect external surfaces, photographs					94	95	94
	31. Inspection					NO LEAK	NO LEAK	NO LEAK
	32. Inspect internal surfaces, photographs					C	C	C
	33. Check for internal leakage					INF	0.1	INF
	34. Check for external leakage					INF	INF	INF
	35. Check for internal noise					INF	INF	INF
	36. Check for external noise					INF	INF	INF
	37. Check for internal vibration					INF	0	INF
	38. Check for external vibration					INF	0	INF
	39. Check for internal pressure or testing					INF	0	INF
	40. Check for external pressure or testing					INF	0	INF

C = COMPLETE
NA = NOT APPLICABLE
TIP = TO BE PERFORMED

Braze Switch Test Checklist

INF = INFINITY (OPEN)
NP = NOT PERFORMED
NRCL = NOT RECD AT OEM, LAB.

	4	5	6	7	8	9	10	11
	PA	PI	N	PY	SI	PY	PA	NY
Initial Info	Test PGM into Switch Logic	C	C	C	C	0	0	0
	Position	NO FIRELEAK	NO FIRELEAK	FIRE	NO FIRELEAK	FIRE	NO FIRELEAK	NO FIRELEAK
	2000degC Burn	C	0	0	0	0	0	0
	2000degC visual external visual observation	C	0	C	0	C	C	C
	4 Check for Cracks or Pinholes	C	0	NA	C	0	NA	0
	5 Any TV appearance	0	NA	NA	NA	C	NA	NA
Vehicle + Connector Assembly	Water 100psi Visual (Oversize) Pinholes	NP	0.2	NA	INF	NA	2	0.4
	Water 100psi in Height Pinholes	NP	INF	NA	INF	NA	0.0M	NP
	Water 100psi in Height Pinholes	NP	INF	NA	INF	NA	0.0M	NP
	Temperature Change from 0°C	C	0	NA	C	0	0	C
Connector Only	80psi Connector Seal	C	0	NA	C	NA	C	C
	Temperature Change from 0°C	NP		NA	INF	NA	NP	NP
	13 Check for TV appearance at connector	C	C	NA	C	NA	0	C
	15 Check for TV appearance	0	0	NA	0	NA	C	C
	14 Check after 500 cycles	0	0	NA	C	NA	C	C
	16 Check after braze operation	TIP	TIP	NA	TIP	NA	TIP	TIP
Switch Delivery Documentation	19 Braze Switch to Calibration Board	C	C	NA	C	NA	C	C
	27 Braze Tested in Assembly Specified Temperature	0.1	0.2	NA	0.3	NA	0.4	0.1
	28 Braze Tested in Height Pinholes	INF	INF	NA	INF	NA	0.2M	INF
	29 Braze Tested in Height resistance	INF	INF	NA	INF	NA	0.4M	INF
	30 Braze Height Resistance	INF	100K	NA	400K	NA	0.4	0.5
Delivery Information	34 Braze Operating Pressure	TIP	0.01	NA	100	NA	100	100
	35 Braze Cleaning Pressure	0.0	0.0	NA	0.0	NA	0.0	0.0
	36 Braze Seal for Leakage	NO LEAK	NO LEAK	NA	NO LEAK	NA	NO LEAK	NO LEAK
	37 Braze Stage 1 Through 20 or Above Only	C	D	NA	C	NA	C	C
		INF	INF	NA	INF	NA	0.1	INF
		INF	INF	NA	INF	NA	-200K	INF
		INF	INF	NA	INF	NA	-200K	INF
		INF	INF	NA	INF	NA	INF	INF
Test	38 Braze Shutdown after 500	NP	NP	NP	NP	NP	C	NP
	39 Braze Preheat Visual Photograph	NP	NP	NP	NP	NP	0	NP
	40 Braze Key	NP	NP	NP	NP	NP	0	NP
	41 Braze visual condition Photograph	NP	NP	NP	NP	NP	0	NP
Test 2000psi	35 Braze 2000psi, height, resistance	NP	NP	NP	NP	NP	C	NP
	36 Braze 2000psi, height, resistance, etc.	NP	NP	10	NP	NP	C	NP
	38 Braze 2000psi of resistance	NP	NP	NP	NP	NP	NP	NP
	39 Braze 2000psi of sealing	NP	NP	NP	NP	NP	C	NP

04007100

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04007100

37137100

C = COMPLETE
NA = NOT APPLICABLE
TIP = TO BE PERFORMED

Brake Switch Test Results

INF = INFINITY (CROSS)
NPF = NOT PERFORMED
NRCISL = NOT READ AT CBL LAB.

	12	13	OASIS	OASIS	EAA	EAA	EAA	EAA
	P	P	P	P	P	P	N	N
Field Info								
	1 Long Radiator Intake Duct Leakage	C	O	C	C	C	C	C
	2 NO PRESSURE	NO PRESSURE	NO PRESSURE	INF LEAK	INF LEAK	FIRE	FIRE	FIRE
	3 Radiator Hatch	C	C	C	C	O	O	NA
	4 Pressurized air vented around sheet metal areas	O	C	O	C	C	NA	NA
	5 Check for Corrosion/Debris/Contamination	C	C	NA	NA	NA	NA	NA
	6 Check for Corrosion/Debris/Contamination	C	C	NA	NA	NA	NA	NA
	7 Liquid Fugitive Emissions	C	NA	NA	NA	NA	NA	NA
Health +	8 Check for Acid/Corrosive Residues	0.3	0.5	NA	NA	NA	NA	NA
Corrosion	9 Check 10.5% Acidity Residues	20.2M	INF	NA	NA	NA	NA	NA
Leakage	10 Check 10.5% Acidity Residues	21.5M	INF	NA	NA	NA	NA	NA
	11 Check for Corrosion/Debris/Contamination	C	C	NA	NA	NA	NA	NA
Chemical	12 Check for Corrosion/Debris/Contamination	C	C	NA	NA	NA	NA	NA
Only	13 Check 10.5% Acidity Residues	INF	INF	NA	NA	NA	NA	NA
	14 Check for full engagement of connectors	O	O	NA	NA	NA	NA	NA
	15 Check wire insulation	C	C	NA	NA	NA	NA	NA
	16 Check wire gauge width	O	C	NA	NA	NA	NA	NA
	17 Check wire insulation thickness	INF	INF	NA	NA	NA	NA	NA
Quality	18 Check for Corrosion/Debris/Contamination	C	C	O	C	NA	NA	NA
Control	19 Radiator Hatch in Classification Sheet	INF	INF	NA	NA	NA	NA	NA
	20 Check Threaded to Structure Threaded to Structure	0.2	0.2	>20K	>20K	NA	NA	NA
	21 Check Threaded to Structure Threaded to Structure	0.2M	INF	>20K	>20K	NA	NA	NA
	22 Check Threaded to Structure Threaded to Structure	0.2M	INF	>20K	>20K	NA	NA	NA
	23 Check Threaded to Structure Threaded to Structure	0.2M	INF	>20K	>20K	NA	NA	NA
	24 Check Threaded to Structure Threaded to Structure	0.2M	INF	>20K	>20K	NA	NA	NA
Tool	25 Check Quality Pressure	140	180	180	NO SOUND	NA	NA	NA
Control	26 Check Quality Pressure	01	20	20	NO SOUND	NA	NA	NA
Prepared	27 Check Thread to Structure	NO LEAK	NO LEAK	NO LEAK	NO LEAK	NA	NA	NA
	28 Check Gauge 1/2 length 1000 psig	C	O	C	O	NA	NA	NA
	29 Check Quality Pressure	4.1	5E	>10K	TRP	NA	NA	NA
	30 Check Quality Pressure	INF	INF	>10K	>10K	NA	NA	NA
	31 Check Quality Pressure	INF	INF	>10K	>10K	NA	NA	NA
	32 Check Quality Pressure	INF	INF	INF	INF	NA	NA	NA
Visual	33 Check Radiator Hatch	SCI LAB	NP	O	TRP	TRP	NA	NA
	34 Check Radiator Hatch	SCI LAB	NP	O	TRP	TRP	NA	NA
	35 Check Radiator Hatch	SCI LAB	NP	O	TRP	TRP	NA	NA
	36 Check Radiator Hatch	SCI LAB	NP	O	TRP	TRP	NA	NA
Verification	37 Check Radiator Hatch, Radiator, Hatchback	SCI LAB	NP	TRP	TRP	TRP	NA	NA
	38 Check Radiator Hatch, Radiator, Hatchback	SCI LAB	NP	TRP	TRP	TRP	NA	NA
	39 Check Radiator Hatch	SCI LAB	NP	TRP	TRP	TRP	NA	NA
	40 Check Radiator Hatch	SCI LAB	NP	TRP	TRP	TRP	NA	NA
	41 Check Radiator Hatch	SCI LAB	NP	TRP	TRP	TRP	NA	NA

04071900

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04071900

3713 7110

C = COMPLETE
NA = NOT APPLICABLE
TBP = TO BE PERFORMED

Baby Switch Test Results

INF = INFINITY (OPEN)
NP = NOT PERFORMED
NRGL = NOT READ AT GEN LAB.

Test Info	1 Any Fluid Intake Switch Response 2 Photograph Results 3 Report any reported customer visual observations 4 Measure Customer Weight 5 Any V/S response 6 Any Mammal Test Response								
Switch + Customer Assembly	Switch 100% to Hospital Readiness Switch 100% to Hospital Readiness								
Connector Only	10 Verify Connector Seal 11 Drive Mating In Without External Assistance 12 Check and seal by hand without external assistance 13 Check seal 14 Check seal again 15 Seal tape Readiness to seal after connection								
Switch Control Unspecified	16 Unsealable Switches in Calibration Blend 17 Sealing Threaded to Sealless Snap-fit Readiness 18 Sealing Threaded to Sealless Snap-fit 19 Sealless Threaded to Sealless Snap-fit 20 Sealless to Sealless Readiness								
Switch Control Prescribed	21 Switch Control Preset 22 Switch Control Preset 23 Seal Vector Location 24 Return Stage 17 through 20 at test only								
Switch	25 Connect electrical wiring day								
Techniques	26 Connect electrical wiring, Photograph 27 Connect electrical wiring 28 Connect electrical wiring 29 Connect electrical wiring, Photograph 30 Connect electrical wiring 31 Connect electrical wiring, Photograph 32 100% 100% PT/FT tests, customer, template 33 100% 100% vsg. hospital, vendor tools, etc. 34 100% 100% readability of responses 35 100% 100% readability of responses 36 100% 100% readability of responses								

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04/09/1998

3713 7111

Brake Pressure Switch Plan Updated 2/16/09
Evaluation Plan for Field Returns.

Evaluation Process Brake Pressure switch / Harness

Box #		Date of update
Message		Box Date Code
Box PN		

Category	Step #	Action	Notes/Date	Comments
Field Info	1	Log Field Info Into Switch Logbook		
	2	Photograph Switch		
	3	Record any unusual external visual observations		
	4	Check for Connector engagement		If not correct conduct X-Ray to determine fit-up between base lip and seal
Switch + Connector Assembly	5	Wire 1(L/G/R)to Wire 2(ORANGE) Resistance		
	6	Wire 1(L/G/R) to Harness Resistance		
	7	Wire 2(ORANGE) to Harness Resistance		
	8	Separate Harness from Switch		
Connector Only	9	Verify Connector Seal		Visual check of Red Seal, Dirt lines, Indentation mark... Indentation mark must be 360 degrees.
	10	Wire 1(L/G/R) to Wire 2(ORANGE) resistance		
	11	Current Leakage Wire 1(L/G/R) to Wire 2(ORANGE)		
	12	Check for full engagement of connector		Visual check of dirt lines on treated switch base
	13	Check wire insulation		
	14	Check wire gray seals		
	15	Cut wire insulation to check for corrosion		Cut insulation longitudinally to check for wicking along wires. If signs of corrosion, identify color, save samples for chem lab.
Switch External: Unpressurized	16	Assemble Switch to Calibration Stand		
	17	Spring Terminal to Stationary Terminal Resistance		
	18	Spring Terminal to Harness Resistance		
	19	Stationary Terminal to Harness resistance		
	20	Base to Harness Resistance		
	21	Current Leakage Spring Terminal to Harness		Stationary Terminal is closest to the outside connector keying tab.

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original 2/16/09
revised 2/16/09

Evaluation Process

Brake Pressure switch / Harness

22 Current Leakage Stationary Terminal to Harness
 23 Voltage drop at 750 mA

Switch External Presurized	24 Switch Opening Pressure 25 Switch Closing Pressure 26 Proof Test for Leakage 27 Repeat Steps 17 through 26 at 180 psig	Do not perform on parts from undamaged tires, as may disturb diaphragm/oil condition Do not perform on parts from undamaged tires, as may disturb diaphragm/oil condition Do not perform on parts from undamaged tires, as may disturb diaphragm/oil condition
Switch Internal	Procedure to remove aluminum crimp ring Use aluminum foil (or plastic if Ford prefers) to mask the analysis surfaces. Also create a paper/tape shield to further reduce chance of contamination during cutting of crimp ring. Place a piece of tape over the area to be cut. Cut crimp ring using jeweler's saw or Dremel cutoff wheel Cut corners of ring at 180 degree orientation Unfold crimp ring Optically examine revealed surfaces. Take optical photographs (Digital camera with macro lens plus instant microphotography) and document observations where appropriate. Inside surface of crimp ring. Base area and underside of base Top of cup Assess Need for Analytical Techniques Start SEM-EDX (Scanning Electron Microscope with Energy Dispersive Analysis of X-rays) analysis on the inside of the ring and on various surfaces of the plastic base. Reprotect the top surface and remove the cup Optically document all revealed surfaces starting with cap. Meanwhile, start SEM-EDX analysis on top side of cup. Particularly look for evidence of corrosion or arcing Particularly focus in on the edges of the contact pin guides and on the indented ring that lines up with interior wall of the switch cavity Particularly look for evidence of corrosion or arcing Decide if we should try to file off any of the overlying debris to try to examine the underlying metal surface. Proceed to perform SEM-EDX analysis on other component surfaces revealed by removal of cup.	
Date Entry	Log All data from this sheet into Switch Log Photographs, Elemental maps etc must be retained and referenced by Switch #	

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Evaluation Process
Brake Pressure switch / Harness

Brake Pressure Switch Plan Updated 2/18/00
 Evaluation Pts for Field Returns

Br #		Date of update	
Mileage		Sv Date Code	
Sv P/N			

Category	Step #	Action	Notes/Date	Comments
Field Info	1	Log Field Info into Switch Log.xls		
	2	Photograph Switch		
	3	Record any unusual external visual observations		
	4	Check for Connector engagement		If not correct conduct X-Ray to determine fit-up between base lip and ret ears.
Switch + Connector Assembly	5	Wire 1(LGR)to Wire 2(ORANGE) Resistance		
	6	Wire 1(LGR) to Harness Resistance		
	7	Wire 2(ORANGE) to Harness Resistance		
	8	Separate Harness from Switch		
Connector Only	9	Verify Connector Seal		Visual check of Red Seal, O-Ring, Indentation mark. Indentation mark must be 360 degrees.
	10	Wire 1(LGR) to Wire 2(ORANGE) resistance		
	11	Current Leakage Wire 1(LGR) to Wire 2(ORANGE)		
	12	Check for full engagement of connector		Visual check of dim lines on mated switch base
	13	Check wire insulation		
	14	Check wire gray seals		
	15	Cut wire insulation to check for corrosion		Cut insulation longitudinally to check for sticking along wires. If signs of corrosion, identify color, save samples for chem I.d.
Switch External Unprotected	16	Assemble Switch to Calibration Stand		
	17	Spring Terminal to Stationary Terminal Resistance		
	18	Spring Terminal to Harness Resistance		
	19	Stationary Terminal to Harness resistance		
	20	Base to Harness Resistance		
	21	Current Leakage Spring Terminal to Harness		

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Evaluation Process

Brake Pressure switch / Harness

22 Current Leakage Stationary Terminal to Harness
23 Voltage drop at 750 mA

Switch External Pressurized	24 Switch Opening Pressure 25 Switch Closing Pressure 26 Proof Test for Leakage 27 Repeat Steps 17 through 28 at 180 psig	Do not perform on parts from underhood fires, as may disturb diaphragm/oil condition Do not perform on parts from underhood fires, as may disturb diaphragm/oil condition Do not perform on parts from underhood fires, as may disturb diaphragm/oil condition
Switch Disassembled	Procedure to remove aluminum crimp ring Use aluminum foil (or plastic if Ford perfect) to mask the analyze surfaces. Also create a paper/tape shield to further reduce chance of contamination during cutting of crimp ring. Place a piece of tape over the area to be cut. Cut crimp dog using jeweler's saw or Dremel cutoff wheel. Cut corners of ring at 180 degree orientation Untold crimp ring Optionally examine revealed surfaces. Take optical photographs (Digital camera with macro lens plus instant microphotography) and document observations where appropriate. Inside surface of crimp ring. Base area and underside of base Top of cup Access Need for Analytical Techniques Start SEM-EDX (Scanning Electron Microscope with Energy Dispersive Analysis of X-rays) analysis on the inside of the ring and on various surfaces of the plastic base. Reposition the top surface and remove the cup Optionally document all revealed surfaces starting with cup. Meanwhile, start SEM-EDX analysis on top side of cup. Particularly look for evidence of corrosion or scaling Particularly focus in on the edges of the ceramic pin guide and on the indented ring that lines up with interior wall of the switch body Particularly look for evidence of corrosion or scaling Decide if we should try to take off any of the overlying debris to try to examine the underlying metal surface. Proceed to perform SEM-EDX analysis on other component surfaces revealed by removal of cup.	
Data Entry	Log All data from this sheet into Switch Log Photographs, Elemental maps etc must be retained and referenced by Switch #	

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Evaluation Process

Brake Pressure switch / Harness

Brake Pressure Switch Plan Updated 2/16/98
 Evaluation Plan for Field Returns

Sw # Mileage Or P/N	Date of update Sw Data Code			
Category	Step #	Action	Notes/Date	Comments
Field Info	1 Log Field Info into Switch Log.xls 2 Photograph Switch 3 Record any unusual external visual observations 4 Check for Connector engagement			If not correct conduct X-Ray to determine fit-up between base tip and red seal
Switch + Connector Assembly	5 Wire 1(LG/R) to Wire 2(ORANGE) Resistance 6 Wire 1(LG/R) to Harness Resistance 7 Wire 2(ORANGE) to Harness Resistance			
	8 Separate Harness from Switch			
Connector Only	9 Verify Connector Seal 10 Wire 1(LG/R) to Wire 2(ORANGE) resistance 11 Current Leakage Wire 1(LG/R) to Wire 2(ORANGE) 12 Check for full engagement of connector 13 Check wire insulation 14 Check wire grey seals 15 Cut wire insulation to check for corrosion		Visual check of Red Seal, Dot lines, Indentation mark. Indentation mark must be 90 degrees.	Visual check of dot lines on mated switch base
				Cut insulation longitudinally to check for wrinkling along wires. If signs of corrosion, identify color, save samples for chem I.d.
Switch External Unpressurized	16 Assembly Switch to Calibration Block 17 Spring Terminal to Stationary Terminal Resistance 18 Spring Terminal to Harness Resistance 19 Stationary Terminal to Harness resistance 20 Base to Harness Resistance 21 Current Leakage Spring Terminal to Harness			Stationary Terminal is closest to the outside connector keying tab.

Evaluation Process Brake Pressure switch / Harness

22 Current Leakage Stationary Terminal to Harness
23 Voltage drop at 750 mA

Switch External Pressurized	24 Switch Opening Pressure 25 Switch Closing Pressure 26 Proof Test for Leakage 27 Repeat Steps 17 through 26 at 180 psig	Do not perform on parts from underhood fires, as may distract diaphragm/oil condition Do not perform on parts from underhood fires, as may distract diaphragm/oil condition Do not perform on parts from underhood fires, as may distract diaphragm/oil condition
Switch Internal	Procedures to remove aluminum crimp ring Use aluminum foil (or plastic if Ford prefers) to mask the analysis surfaces. Also create a paper tape shield to further reduce chance of contamination during cutting of crimp ring. Place a piece of tape over the area to be cut. Cut crimp ring using jeweler's saw or Dremel cutoff wheel Cut corners of ring at 180 degree orientation Unfold crimp ring Optically examine revealed surfaces. Take optical photographs (Digital camera with macro lens plus instant microphotography) and document observations where appropriate. Inside surface of crimp ring. Base area and underside of base Top of cap	
Analytical Techniques SEM/EDXAR	Answer Need for Analytical Techniques Start SEM-EDX (Scanning Electron Microscope with Energy Dispersive Analysis of X-rays) analysis on the inside of the ring and on various surfaces of the plastic base. Reprotect the top surface and remove the cap Optically document all revealed surfaces starting with cap. Meanwhile, start SEM-EDX analysis on top side of cap. Particularly look for evidence of corrosion or arcing Particularly focus in on the edges of the ceramic pin guide and on the Indented ring that lines up with interior wall of the switch cavity Particularly look for evidence of corrosion or arcing Decide if we should try to clean off any of the overlying debris to try to examine the underlying metal surface. Proceed to perform SEM-EDX analysis on other component surfaces revealed by removal of cap.	
Data Entry:	Log All data from this sheet into Switch Log Photographs, Elemental maps etc must be retained and referenced by Switch 4	

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Spare Bearing house
Brian

Impulse Test

0 - 1450 lpm. 2 Hz.

135°C 500,000-N_f

91-92
Crimp

90 psi ~ ripple?

Issue could be brake fluid & water
 $i \rightarrow$ corrosion (multiple modes).

process change

4 levels of Kapton \rightarrow unknown
solid state devices for high cycles

- Nov 91 \rightarrow Nov 92 Recall Test T.I.C
150,000 initial sw. 60% o/H before recall C.V
replace sw w/ replacement of sw G.M.
& connector shell w/ silicone seal instead
of foam seal.
Campaign start ~ May 30, 1999
Cost \rightarrow FC 50 \sim \$

UTA \rightarrow Shell
T.I. \rightarrow 3-Switches

Options 3-30-04

1. Blotch Pad - no sw. yet

2. Shroud to protect fire in sw.

3. ~~trip~~ relay \Rightarrow limit current to 200 ma from 192
not eliminate sw. corrosion F6AB-14B192-1A

4. trip loop / ground isolation - no plastic/rubber
device yet avail

To shut off the Start Control:

1. Drivers off

2. Brake to ~~Start~~ A10 mph

3. B " " " = 25 mph

B - In Base material action existing fuel is the same

Action Plan

1. Not starting the in-relay fire @ 200 ma @ 1amp C2
amp w/o 1amp fuel

- Towing package on Veb?
- fuse change (15A)?
type of fuse (slow-blow max)?
bad
- Engine Clean up from floods, etc.
Ammonium
- Oxyalate is NOT part of brake fluid.
- no Sulphur in Brake fluid
possible Rose (rust) causing

Dow

- SAE report of Brake fluid corrosion
- Captain & Captain rep't.
- ~~Brake fluid~~ (not) fluid

T.I.



508 - 2

401 - 2

930
9a.mee

NN 734

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