

**EA02-025**

**TEXAS INSTRUMENTS,  
INC.'S 9/10/03  
ATTACHMENT**

**REQUEST NO. 7**

**BOX 8**

**PART A-U**

**PART A**



AG, 04 '92, 18/21, TEXAS INSTRUMENTS 618 2252209  
Ford Motor Company  
of Australia Limited  
A.C.N. 004 116 223

Rapifax Transmission

P.2/2

TO: DANNY O'DRISCOLL - TEXAS INSTRUMENTS  
cc: P.SAWYER/C.CAREY - BCIA AUST.  
FAX: cc: V.STOLOS/P.WILLIAMS - SUPPLY H/O  
DELIVERY:  NORMAL  RUSH-CALL FOR PICKUP cc:T. WORTHINGTON

OUR REF. ....

PAGE 1 OF 1

FROM JOHN PARK  
DEPT.: CAPRI ENGS.

RAPIFAX NO: (082) 788109  
SEE OVER

DC: DATE: 4/9/92

SUBJECT - SWITCH ASY: SPEED CONTROL DEACTIVATE (94JA-9F924-AB)

TESTS CONDUCTED ON THE EVACUATE AND FILL EQUIPMENT IN BROADMEADOWS PLANT 2, CONFIRMS THAT VACUUMS AS LOW AS 0.4 MILLIBARS ARE ACHIEVED, WITH A RANGE OF 0.4 - 1.0 MILLIBARS OVER THE VEHICLE POPULATION. THESE LOW PRESSURES HAVE BEEN ESTABLISHED TO MINIMISE THE 'SOFT PEDAL' CONCERN WHICH IS INHERENT IN CAPRI,

SINCE ABOVE SWITCH ASY IS DESIGNED TO COMPLY WITH ES-F2VG-9F924-AA, WHICH REQUIRES A VACUUM CAPABILITY OF 3-6 MM HG, THERE IS DOUBT AS TO THE ABILITY OF 94JA-9F924-AB TO COPE WITH OUR PROD EVAC AND FILL EQUIPMENT REQUIREMENTS.

WOULD YOU PLEASE INITIATE AND PREPARE A WORK PLAN TO

A) EVALUATE EFFECTIVENESS OF SOALS IN 94JA-9F924-AB AT VACUUM LEVELS OF 0.4 MILLIBARS.

B) IF NECESSARY, TO DETERMINE DESIGN CHANGES REQD TO SWITCH TO COPE WITH A VACUUM OF 0.4 MILLIBARS.

C) ESTABLISH COST AND TIMING IMPLICATIONS OF REQUIRED CHANGES (BOLING AND PIECE). OUR PROGRAM INTENT IS

FOR AN OTS SUBMISSION TO SUPPORT PCB BUILD NO 1 WITH STOCK IN B/M PLANT (MRD DATE) BY 1/14/92. LATEST POSSIBLE OTS SUBMISSION WOULD BE APRIL 19.

93 TO SUPPORT PCB BUILD.

WOULD YOU PLEASE TREAT THIS AS EXTREMELY URGENT, AND ADVISE YOUR TIMING FOR PREPARATION OF A) B) C) ABOVE BY 7/9/92.

REGARDS

TI-NHTSA 011208

**Board of Water Commissioners  
of Arkansas, L.L.C.**  
A.A.H. 0001 100 000

Digitized by srujanika@gmail.com

OUR REF.  
PAGE 1 OF 2  
FROM JOHN BEEC  
DEPT. CARRY WINGS  
FAX NO. (02) 788109  
MURKIN  
SIC: DATE: 6/1/94

TO: DANNY O'DOOLE - THERM INSTRUMENTS  
PATRICK SAWYER/CATHY CAREY - BCIA  
FROM: V. STAVIS/T. WORTHINGTON/P. ROBBETS/P. W.  
DELIVERY:  NORMAL  RUSH-CALL FOR PICKUP

SUBJECT - SWITCH ASY - SPEED CONTROL DEACTIVATED 940A-95124-AB

FOLLOWING DISCUSSIONS WITH DANNY O'DELL, RE REQUIREMENTS OUTLINED IN MY RAPPORT TO PATRICK SAWYER, DATED 3/10/72, IT IS APPARENT THAT THE [REDACTED]

REPRESENTS A MARGINAL PERFORMANCE REQUIREMENT, WHICH IS NOT ACCEPTABLE TO TEAMS INSTRUMENTS. REVIEW OF THE HMO IMPULSE TEST (H.M.T.) SHOWS THE REQUIREMENTS FOR PRESSURE PULSE (1440 PSI), AND TEST ROOM TEMPERATURE (135°C) TO BE FAR MORE SEVERE THAN CAAI FIELD REQUIREMENTS, AND OF THE REQUIREMENTS OF S84 DA-3140-A1 CYLINDRICAL ASY-SOURCE PLASTIC, AS APPLIED TO THE CAAI DURAY MASTER CYLINDER (1000 PSI AND AT ROOM TEMPERATURE, RESPECTIVELY).

I AM THEREFORE REQUESTING THAT THE DRAWING 943A-9F924-1A  
BE REVISED TO INCLUDE THE NOTE - "REQUIREMENTS OF  
ES-F2UC-9F924-1A REVERSED TO DELETE SECTIONS III.E  
(IMPULSE TEST) AND III.K (VACUUM TEST).

**ALL OTHER TEST CONDITIONS AND ACCEPTANCE CRITERIA INCLUDED IN MIL-STD-883 ARE TO BE APPLIED.**

INTENT IS THAT THE CURRENTLY RELEASED SWITCH ASY - 94JA-4F924-1B BE RETAINED, UNTIL ALL VENDOR RIG TESTING IS COMPLETED TO SUPPORT A FOTS SUBMISSION OF T-ASY, BRACE MASTIC CULTURE 94JA-2C247-1B, TO FORD, PRIOR

TH-NHTSA 011209

Approved Quality  
Control Department  
AIAA-93-0000

Flight Test Division

DATE REC.

PAGE 2 OF 2

FROM

DEPT:

REPORTER: R. Brown

NO. DATE

TICK  
FAX  
DELIVERY:  NORMAL  RUSH-CALL FOR PRIORITY

TO PCB MRD (14/4/93). SINCE APPENDIX IT IS  
ANTICIPATED THAT TESTING ALREADY CONDUCTED AND ACCEPTED  
FOR NADP PRESSURE SWITCH WILL BE CARRIED OVER INTO  
943A-2E924-AB, AND THAT THE ONLY UNIQUE PQA  
REQUIREMENTS WILL APPLY TO ADAPT TEST REQUIREMENTS, AND TO  
DIMENSIONAL CHARACTERISTICS OF INTERNAL (INTERNAL) THREAD AND  
TO .10-.RMS 3.403.

WORLD TRADE REQUIREMENTS PLEASE ADVISE CONCERNING WITH  
ADAPT TO ANGLES DRAWING TO BE REVISED ACCORDINGLY.  
REGARDS.



10 MAY 1993 BY R. BROWN

TINHHTSA 011210

## PRESSURE SWITCH DATA

Form 21605

TEST NO. 218-15-06

DEVICE	DATE REQUESTED	REQUESTED BY	REQUESTED COMPL. DATE
Bus Car 7785	9/06/24	Steve Officer	
Coffey, D. Deenice	9/06/25		
PROJECT TITLE: Fed Speed Control PS			

CUSTOMER: -

PURPOSE OF TEST: To test our Ford Australia Switch to a more general vacuum.

- PROCEDURE:
- 3 devices, 3 sensors
  - start devices
  - 5 cycles to meet attainable vacuum & hold position
  - stop devices
  - let heat off sensors and examine

Device 00 01	V	act 59							
Device 00 02	V	act 59							
Device 00 03	A	133 99							
Device 00 04	C	129 76							

The system is now being tested to be continued.

(OVER)

TH-NHTBA 011211

-MSG M#= 68675 FR=SB01 TO=ZARN SENT=06/29/92 02:00 PM  
R#=142 ST=C DIV=0050 CC=00101 BY=SB01 AT=06/29/92 02:00 PM

TO: Danny O'Driscoll DDD

CC: Peter Burner SAGT  
Dave Czarn ZARN  
Jeff DiDomenico DIDO  
Charlie Douglas CMP1

FR: Steve Offiler SB01

SJ: Ford of Australia Vacuum Spec.

We have attempted to test six 77PS devices on a vacuum manifold to the levels requested by Ford of A (0.4 to 1.1 mBar). After several minutes, we were able to attain a vacuum level which fluctuated between 1.3 and 1.45 mBar. I believe this represents the true limits of our equipment in its present form. We cycled five times between atmospheric and vacuum, as called out in the Ford of USA spec (ES-F2VC-9F924-AA). Three devices were characterized before and after, and showed no apparent performance degradation as a result of the vacuum exposure. The other three devices were disassembled in such a way as to allow inspection of the diaphragm/fluid seal in the assembled state. Unfortunately, the diaphragms showed obvious signs of being sucked significantly out of place on two of the three devices. I have a concern that this could lead to shortened diaphragm life. This has raised a number of questions:

- \* How is the diaphragm life affected, and at what vacuum level?
- \* Is five cycles needed, based on the factory-fill vacuum procedure, and will our device look better if only cycled once to these vacuum levels?
- \* Is the out-of-place diaphragm pushed back into place once the device is pressurized in fluid?

This is going to require additional effort and testing on my end to answer these questions, which presents some difficulty since both time and cycler capacity are in relatively short supply.

As far as Ford of A is concerned, can we get an answer on why they say they need a vacuum level (0.4 to 1.1 mBar) which exceeds the Ford of USA level by more than an order of magnitude? Is this realistic? What is their feeling on the number of vacuum cycles?

Regards,  
Steve O.

TI-NHTSA 011212

TO=371117 FR=S801 TO=ZARN SENT=07/10/92 09:24 AM  
Re=024 ST=C DCV=0050 CC=00101 BY=S801 AT=07/10/92 09:24 AM

TO: Danny O'Driscoll DCR

CC: Steve O'Riordan ZAPM  
Charlie Douglas IMP

FR: Steve O'Riordan S801

SJ: Vacuum x Samples

Danny, Glad to hear that Ford is proceeding with the measurements of the true vacuum levels in their system. Since we discovered that extreme vacuum tends to dislodge our diaphragm, I am maintaining my plan to better understand the effect (if any) of vacuums on diaphragm life. However, I'll be lowering the priority.

For the record, here are the vacuum unit conversions I've been using:

$$1 \text{ bar} = 14.504 \text{ psi} = 1000 \text{ mBar}$$

$$1 \text{ mm Hg} = 0.01934 \text{ psi}$$

$$1 \text{ mBar} = 0.750 \text{ mm Hg}$$

Ford (USA) spec of 3 to 6 mm Hg equals 4 to 9 mBar.

One other point: We have settled upon dark gray as the color for Cedar (77PSL6-1). In your message #2340878 you mention that the color change will take weeks, not days, to come through. We are intending to use dark gray for the 300 OTB samples, since this is indeed production-intent. We thought a better check to make sure that it's okay to ship parts in dark gray.

Regards,  
Steve O.

TMHTSA 011213

-MSG MH= 00504507 FR=8801 TD=8801 SENT=08/06/92 01:29 PM  
SH=194 ST=C DIV=0050 CC=00101 BY=8801 AT=08/06/92 01:29 PM

TO: Danny O'Driscoll DOD

CC: Dave Czarn ZARN  
Jeff DiDomenico D100  
Charlie Douglas CMP1

FR: Steve Offiler 8801

SJ: Ford of Australia Vacuum Spec

TI-NHTSA 011214

Thanks for the fax and MSG, Danny. (For future reference, my fax number is 508-699-3153.). We in Attleboro share your surprise that Ford is truly maintaining this rather extreme vacuum level, '0.4 mBar (=0.3 mm Hg or Torr, =0.006 psia). I've mentioned before that this is about one order of magnitude higher than the Ford US spec; although to be honest, we don't know where the US spec of 3-6 mm Hg came from and how accurate it is. In response to John Peck's request:

A) Evaluate the effectiveness of seals at vacuum level of 0.4 mBar.

We have found that the "diaphragm" (a polymer film material which serves to separate the fluid from the internal workings of the switch) is literally sucked out of place at a vacuum of about 1.3 mBar. I expect this condition to worsen down at 0.4 mBar. Although it still provides an effective seal initially, when sucked out of place the diaphragm becomes wrinkled, and these wrinkles give rise to stress concentrations during pressure cycling which will result in ruptures/tears and leakage long-term.

To evaluate the situation, I plan to first conduct a controlled experiment of diaphragm life versus vacuum. I'll need to come up with vacuum equipment capable of 0.4 mBar, but I don't think this will present a major problem. Eighteen test devices will be built. Twelve of these will be subject to vacuum, with 6 destined for life-cycling, 3 opened for visual inspection, and 3 subject to a 3000 psi proof pressure test then opened for visual inspection. (The proof parts will tell me what happens to the diaphragm with application of positive pressure). The remaining 6 of 18 will be the control group for life-cycling, not subject to vacuum. We'll run the cycle test at ES parameters of pressure and temperature, however all parts will be run to failure and analyzed using Weibull techniques to see if the vacuum causes a significant reduction in diaphragm life.

Scheduling the above is a bit touchy. This test must run in my brake-fluid

is writing now which will complete about 13/08/92, and I've got Capri validation which must be started approx. 04/09/92 in order to deliver the finished report to Australia as promised by end-September. I'll be running the vacuum test in the 3-week window between these two tests. So, results should be available roughly the first week of September.

B) Determine design changes required, if necessary.

This is purely speculation at this point. We really need to know the results of the testing in A) above before any direction can be determined. Initial ideas at this point include: an additional mechanical element to hold the diaphragm against the forces which displace it; more robust (thicker, stiffer) diaphragm materials; alternate diaphragm materials (such as EPDM rubber); features (such as knurling) to hold the diaphragm tighter; even wilder ideas like a check-valve which closes upon vacuum application, etc etc.

C) Cost and timing implications.

A 1/11/92 OTS submission for PPB build would mean a production-tooled solution by mid-October. The window for cycling testing to validate the proposed solution(s) starts with completion of the normal Capri validation in mid-September. It is flatly impossible to identify the solution, prototype it, prove it out, tool it, and validate using the production-tooled parts, in one month's time. Unless, of course, we hit upon something which requires no tooling (such as a material change). Therefore, it's likely that we'll be forced to work off of the 14/04/93 OTS submission for PCB build. (Question: are either of these builds producing cars for sale to customers?)

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A couple of other points, Danny. I'm updating the envelope drawing to include a note pointing to the O-ring. We've noticed an error in the overall length of the device, presently given as 57.15 mm. This number is correct on other devices which use a different hexport and was accidentally carried over on the Capri drawing. The O-ring hexport used on Capri is an off-the-shelf item also used on European power steering switches, and is longer in the hex-flat area by about 1 mm. Simple math says 58.15 would be correct, but we figured we'd take the opportunity to build in a little slack and would propose 58.5 mm.

Also, our EE testing strategy for Capri is to perform only the Impulse and Thermal Cycle tests, and claim similarity to 77PSL3-1 (the silent US Pass-Car device) for all the rest of the tests. To the best of my knowledge, this is an internally generated strategy, and really should be concurred by the customer. The rationale for running only these two tests is because the Capri device uses the same pressure-sensing components as L3-1, and the same electrical elements including the base material. The only real difference is the hydraulic seal, which is O-ring on Capri versus BAF J512 metal-to-metal on L3-1. The Impulse test exercises this seal at elevated temperature over a large number of cycles, and Thermal Cycle includes thermal expansion effects and rubber durometer changes with temperature. Please make sure our customer is aware that the EE test report will be structured as such.

Regards,  
Steve O.

-MSG Nbr= 2217727 FR=DOD TO=ZARN SENT=08/11/92 06:34 AM  
R#=034 ST=C DIV=0071 CC=09260 BY=DOD AT=08/11/92 03:37 AM

To: STEPHEN B. OFFILER 3801  
Cody, David Ciarm 2ARN  
CHARLIE DOUGLAS CMPA  
From: DANNY O'DRISCOLL BOD  
Subj: FORD CAPRI VACUUM SPEC

STEVE, THANKS FOR THE FAIRLY COMPREHENSIVE SUMMARY OF THE VAC. SPEC SITUATION. FORD HAVE REQUESTED THAT WE DEVELOP A PRESS SWITCH THAT CAN WITHSTAND THE VACUUM LEVEL AND SUPPLY PRODUCTION INTENT PROTOTYPES FOR THE FPD BUILD DATE OF 9/11/92.

WE DO NOT EXPECT THAT THE VALIDATION PROCESS WILL BE LENGTHY AND AS SUCH WE WOULD  
WANT TO COMMENCE AND ARRANGE FOR WHAT WE CAN PRACTICALLY DO IN OTHER  
TECHNIQUES AS WELL AS BEING CAPABLE OF SUMMARISING THE VALIDATION DATA  
AS IT COMES IN AS SOON AS POSSIBLE.

12.05.2014 10:36 AM (Page 107)

10.000 - 100.000 PAULI DUCK DVS 2-2024403 10.000 VACUUM THERMAL BIAS SYSTEM  
10.000 - 100.000 PAULI DVS 2-2024403 10.000 VACUUM THERMAL BIAS SYSTEM

10. THE FIVE HOUR SCHEDULE FOR THE 1962-1963 SCHOOL YEAR IS AS FOLLOWS:  
11. 7:00 AM - 7:30 AM ALL STUDENTS ARE LOVED IN THE INTERDIAZ COMPANY PLAZA  
12. 7:30 AM - 8:00 AM THE PLAZA AND 4TH AVENUE.  
13. 8:00 AM - 8:30 AM THE PLAZA AND 4TH AVENUE MAY BE USED BY THE  
14. 8:30 AM - 9:00 AM THE PLAZA AND 4TH AVENUE MAY BE USED BY THE  
15. 9:00 AM - 9:30 AM THE PLAZA AND 4TH AVENUE.

REBELLION IN THE GULF AND TURKISH AND TERRITORIAL TRADES CONTINUE ON. KING JOHN  
OF ENGLAND HAS ALREADY BEEN A FUGITIVE. THE KING'S BODY GUARD IS IN CONSPIRACY  
WITH THE TURKS. THE TURKISH EMPIRE IS GOING DOWN AND NAZAR HAS A SECRET VENTURE IN  
GOLD AND SILVER. THE COST OF THE WAR HAS BEEN A TERRIBLE BURDEN FOR NEARLY A DECADE. A LOT OF  
MONEY HAS BEEN SPENT ON THE WAR AGAINST THE TURKS.

2017.5.3.

TINH TRẠM 011216

MSO MM# 16934 FR=SB01 TO=ZARN SENT=08/11/92 03:47 PM  
R#:061 ST=C DIV=0050 CC=00101 BY=SB01 AT=08/11/92 03:47 PM

TO: Danny O'Driscoll DOD

CC: Jeff DiDomenico ZARN  
Charlie Douglas D1DO  
Jim Watt CMPI  
JW02

FR: Steve Officer SB01

SBJ: Vacuum - Capri 77PSL6-1

TH-NHTSA 011217

Danny: I'm trying to lay out a schedule that will allow us to meet the NRD of 1/11/92. Thanks for clarifying that these can be production-intent PROTOTYPES. Previously, I was under the impression that "off-tool" (OTS) PRODUCTION parts were needed.

I have a key question that significantly impacts my schedule: do we still need to submit a complete ISR package by the end of September as previously planned? I am assuming the answer is no. Instead, one possibility is to submit a partial ISR in the late September timeframe, which contains no ES testing report. It makes no sense to test a soon-to-be-modified design (assuming we do indeed have a problem with the present design), and I'll need my cycle for development testing and evaluation of the high-vacuum design modifications instead of performing ES testing on the "old" 16-in. A complete ES report on the new design will be planned for late December/early '93 per your inputs. Please close with Ford on this. Do they even need/want the partial in Sept, or do we just send the complete package in Dec?

I'm presently reworking my plans for diaphragm life testing of the present design under high vacuum conditions. Instead of spending time simply evaluating the present design, instead I'll be including several prototype ideas in this same test. Regarding your question on the number of cycles of vacuum: This is a good point. We'll structure the evaluation to include checks of the present design at one and five vacuum cycles to characterize this. Maybe we'll discover the problem can be handled by specmanship. All other proto's will be at five cycles.

Danny, I'd like to make sure you understand a seeming inconsistency in the ES. My evaluation work is to characterize diaphragm life after application of vacuum. This is because I've noted visually that, after vacuum, the diaphragm looks like there might be a long-term life problem. The inconsistency is that nowhere in the ES does any device undergo a diaphragm life test after vacuum! So it may seem that, purely from a spec standpoint, I am performing superfluous work - when in fact this is a very fundamental test. On the other hand, we don't necessarily wish to rewrite the spec to include this type of test either, because it adds a lot of extra work. I'm trying to sensitize you to this because I suspect it may come up with Ford when we discuss our ES testing plans for validation with them. If I were Ford, I'd insist upon this addition.

Do we allow air to leak into the brake system? I really don't know, but I doubt it. If we can plug up a nice tight vacuum test system, that doesn't leak when dead-ended, then we can try a leakdown test on our device.

I think your comments on the soft pedal issue are pretty interesting - no one has been able to solve it in a decade! So does super vacuum actually make a difference? And if so, what happens later on when the brake system is serviced?

What did you think my question regard the change in the overall length  
from 37.75m, now to 38.5 mm. back? I'm looking ahead with the print  
changes I'm hoping we don't blind-side Ford with this.

Regards,  
Steve O.

TI-NHTSA 011218

-M80 MS= 00018P34 PR=BB01 TO=BB01 SENT=08/11/92 09:47 PM  
SU=194 ST=C DIV=0050 CC=00101 BY=BB01 AT=08/11/92 09:47 PM

TO: Danny O'Driscoll D00

CC: Dave Czarniak ZAPM  
Jeff DiDomenico DIDO  
Charlie Douglas CRP1  
Jim Watt JWC2

FR: Steve Offiler BB01

Bcc: Vacuum - Capri 77PBL6-1

TI-NHTSA

Danny: I'm trying to lay out a schedule that will allow us to meet the MRD of 1/11/92. Thanks for clarifying that these can be production-intent PROTOTYPES. Previously, I was under the impression that "off-tool" (OTS) PRODUCTION parts were needed.

I have a key question that significantly impacts my schedule: do we still need to submit a complete IER package by the end of September as previously planned? I am assuming the answer is no. Instead, one possibility is to submit a partial IER in the late September timeframe, which contains no EE testing report. It makes no sense to test a soon-to-be-modified design (assuming we do indeed have a problem with the present design), and I'll need my cycler for development testing and evaluation of the high-vacuum design modifications instead of performing EE testing on the "old" LA-1. A complete EE report on the new design will be planned for late December/early '92 per your input. Please close with Ford on this. Do they even need/want the partial in Sept, or do we just send the complete package in Dec?

I'm presently revamping my plans for diaphragm life testing of the present design under high vacuum conditions. Instead of spending time simply evaluating the present design, instead I'll be including several prototype ideas in this same test. Regarding your question on the number of cycles of vacuum: This is a good point. We'll structure the evaluation to include checks of the present design at one and five vacuum cycles to characterize this. Maybe we'll discover the problem can be handled by spec'manship. All other proto's will be at five cycles.

Danny, I'd like to make sure you understand a seeming inconsistency in the EE. My evaluation work is to characterize diaphragm life after application of vacuum. This is because I've noted visually that, after vacuum, the diaphragm looks like there might be a long-term life problem. The inconsistency is that nowhere in the EE does any device undergo a diaphragm life test after vacuum! So it may seem that, purely from a spec standpoint, I am performing superfluous work - when in fact this is a very fundamental test. On the other hand, we don't necessarily wish to rewrite the spec to include this type of test either, because it adds a lot of extra work. I'm trying to sensitize you to this because I suspect it may come up with Ford when we discuss our EE testing plans for validation with them. If I were Ford, I'd insist upon this addition.

Do we allow air to leak into the brake system? I really don't know, but I doubt it. If we can plumb up a nice tight vacuum test system, that doesn't leak when de-pressured, then we can try a leakdown test on our device.

I think your comments on the soft pedal issue are pretty interesting - no one has been able to solve it in a decade! Got some super vacuum schmees?

TI-NHTSA 011219

Finally, did you note my question regarding a change in the overall length spec from 57.1844 max to 58.5 mm max? I'm going ahead with the print changes; I'm hoping we don't blind-side Ford with it.

Regards,  
Steve O.

TI-NHTSA 011220

-MSG NM# 2237069 FR=DOD TO=ZARN SENT=08/12/92 03:14 AM  
R#=062 ST=C DIV=0071 CC=00260 BY=DOD AT=08/12/92 02:46 AM

To: STEPHEN B. OFFILER 8B01

Copy: JEFFREY A. DIDOMENICO D1D0  
CHARLIE DOUGLAS CMP1  
JIM WATT JW02  
DAVID CZARN ZARN

From: DANNY O'DRISCOLL DOD

Subj: CAPRI VACUUM

I AM WORKING ON GETTING SOME ANSWERS TO YR QUESTIONS AT BCIA AND FORD  
BUT IT MAY TAKE A FEW DAYS.

YES - I DID COMPREHEND THE LENGTH CHANGE BUT AGAIN I AM WAITING ON BCIA.

WITH REGARD TO YR ES TEST STRATEGY: HOW SUCCESFULL DO YOU THINK WE CAN BE  
IN OUR EFFORTS TO CLAIM SIMILARITY TO THE L3-1 ? MY VIEW IS THAT IT MAY  
BE DIFFICULT TO CLAIM SIMILARITY BETWEEN A MODIFIED "LOW VACUUM" P/S AND  
THE L3-1.

CHARLIE - THE CAPRI WAS DESIGNED AND BUILT ON THE A PRE EXISTING FORD  
MODEL CALLED A "LASER". NOW, THE LASER IS A BADGED MAZDA 323. BY BADGED  
I MEAN THAT FORD ASSEMBLE THE MAZDA 323 (WITH SOME MINOR VARIATIONS)  
UNDER A LICENSING AGREEMENT WITH MAZDA JAPAN.

FORD HAVE ALWAYS HAD A SOFT PEDAL PROBLEM WITH THE LASER AND SINCE IT USES  
THE SAME BRAKE SYSTEM AND COMPONENTRY - SO DOES THE CAPRI.  
FORD HAVE DETERMINED THAT IT WILL COST FAR TOO MUCH TO REDESIGN THE MASTER  
BRAKE CYLINDER TO ELIMINATE THIS PROBLEM SO THEY USE A LOW VACUUM EVACUATION  
AND FILL PROCESS TO OVERCOME THIS.

THE MAIN REASON BEHIND THE LOW VACUUM LEVEL IS TO ELIMINATE THE SOFT PEDAL  
BUT AT THE SAME TIME FILL THE BRAKE SYSTEM WITHIN THE ALLOTTED MANUFACTURING  
ASSEMBLY CYCLE TIME.

WHEN THE BRAKE SYSTEM IS SERVICED IN THE FIELD A DIFFERENT METHOD OF FILLING  
IS USED WHICH IS LONGER (TAKES ABOUT 15 MINUTES COMPARED TO 2 MINUTES IN THE  
FACTORY) AND CONSIDERED BY THE FORD GUYS TO BE A SUPERIOR METHOD THAN THE  
EVAC & FILL PROCESS.

I HOPE THIS ANSWERS YR QUESTIONS.

RGDS, DANNY

TMNFR04041221

TITLE: Ford-Australia High Vacuum Testing

PURPOSE: Characterization of diaphragm life vs. vacuum and prototyping work on possible solutions if a problem is indicated.

SCOPE: This test will be conducted with emphasis on controlling variables. All devices shall be built at the same time, by the same individual, and tested simultaneously in the same cycler. Previous diaphragm life testing has produced widely varied results, so it should be recognized that the results of this test are valid only in a relative manner. This emphasizes the importance of controlling as many variables as possible and including as many test lots as possible.

Control lots and prototype lots will be constructed, and indicated devices subject to high vacuum (0.006 psia) per the attached chart. Diverging from the ES, we will subject the devices to only one cycle of vacuum, to be held for 60 seconds per the ES. Certain devices will be cycled to failure using the standard Impulse parameters; others will be cut open for diaphragm visual inspection. Weibull analysis will be employed to compare the diaphragm life in each lot. Test lots include: control lot, standard production configuration, no vacuum; standard production with high vacuum; knurled hexports to increase grip on diaphragm; full-round diaphragms; rubber diaphragms; and Tefzel (fluid-layer) diaphragms. A total of 48 parts will be built; 42 of these must be subject to vacuum; and a total of 12 will be opened.

In a separate procedure, a leakdown test will be conducted. This requires initial leak characterization of the system, dead-ended. Then, a standard production device is installed and the characterization is repeated. Initial vacuum level should be 0.006 psia; final vacuum level should be checked at intervals of 1, 5, and 10 minutes. The vacuum source is to be isolated from the test once the indicated vacuum is attained.

580/920819/64 : VACUUM

TI-NHTSA 011222

	1 DRIVE	2 VAL	3 TST	4	5	6	7	8	9	10
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TINHTSA 011223

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*

SHAPE(BETA) PARAMETER : 4.554  
LOWER LIMIT : 2.251 —  
UPPER LIMIT : 9.213

HIGH VACUUM LOT  
6 PIECES

SCALE(THETA) PARAMETER: 1597.764  
LOWER LIMIT : 1362.310 —  
UPPER LIMIT : 1873.912

DATA

846 K  
1219 K  
1556 K  
1619 K  
1691 K  
1783 K

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\* WEIBULL SLOPE : 2.25  
\* CHARACTERISTIC LIFE : 1362.30

NO.	RELIABILITY(%)	TIME
1	90	<u>501.3031 K</u>

C LO - 301 K  
C MED - 975 K  
C HIGH - 1480 K

(ALL AND 90% ~~90~~ 501.3031 K IS LESS THAN 1373 K)

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*

SHAPE(BETA) PARAMETER : 14.025  
LOWER LIMIT : 6.933 —  
UPPER LIMIT : 28.372

CONTROL LOT

6 PIECES

DATA

1284 K  
1352 K  
1377 K  
1572 K  
1571 K  
1639 K

SCALE(THETA) PARAMETER: 1522.074  
LOWER LIMIT : 1445.282 —  
UPPER LIMIT : 1602.945

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\* WEIBULL SLOPE : 6.93  
\* CHARACTERISTIC LIFE : 1445.28

NO.	RELIABILITY(%)	TIME
1	90	<u>1044.6863 K</u>

C LO - 1044 K  
C MED - 1296 K  
C HI - 1461 K

TI-NHTSA 011224

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*  
SHAPE(BETA) PARAMETER : 4.554  
LOWER LIMIT : 2.251 ←  
UPPER LIMIT : 9.213

HIGH VACUUM LOT  
6 PIECES

SCALE(THETA) PARAMETER: 1597.764  
LOWER LIMIT : 1362.310 ←  
UPPER LIMIT : 1873.912

DATA  
846 K  
1219 K  
1555 K  
1619 K  
1681 K  
1792 K

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\*\*\*\*\*  
\* WEIBULL SLOPE : 2.25  
\* CHARACTERISTIC LIFE : 1362.30

NO.	RELIABILITY(Z)	TIME
1	90	<u>501.3031 K</u>

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*  
SHAPE(BETA) PARAMETER : 14.025  
LOWER LIMIT : 6.933 ←  
UPPER LIMIT : 28.372

CONTINENTAL LOT

6 PIECES

SCALE(THETA) PARAMETER: 1522.074  
LOWER LIMIT : 1445.282 ←  
UPPER LIMIT : 1602.945

DATA  
1324 K  
1352 K  
1377 K  
1512 K  
1571 K  
1659 K

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\*\*\*\*\*  
\* WEIBULL SLOPE : 6.93  
\* CHARACTERISTIC LIFE : 1445.28

NO.	RELIABILITY(Z)	TIME
1	90	<u>1044.6863 K</u>

TINHTSA 011225

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*  
SHAPE(BETA) PARAMETER : 4.554 HIGH VACUUM LOT  
LOWER LIMIT : 2.251 ← 6 PIECES  
UPPER LIMIT : 9.213

SCALE(THETA) PARAMETER: 1597.764 DATA  
LOWER LIMIT : 1362.310 ← 844 K  
UPPER LIMIT : 1873.912 1214 K  
1555 K  
1619 K  
1681 K  
1782 K

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\*\*\*\*\*  
\* WEIBULL SLOPE : 2.25  
\* CHARACTERISTIC LIFE : 1362.30

NO.	RELIABILITY(%)	TIME
1	90	<u>501.3031 K</u>

ESTIMATE AND TWO SIDED 90 % CONFIDENCE  
INTERVALS FOR DISTRIBUTION PARAMETERS

\*\*\*\*\*  
SHAPE(BETA) PARAMETER : 14.025 CONTROL LOT  
LOWER LIMIT : 6.933 ← 6 PIECES  
UPPER LIMIT : 28.372

SCALE(THETA) PARAMETER: 1522.074 DATA  
LOWER LIMIT : 1445.282 ← 1324 K  
UPPER LIMIT : 1602.945 1352 K  
1377 K  
1512 K  
1571 K  
1659 K

TIME VALUES FOR SPECIFIED LEVELS OF RELIABILITY

\*\*\*\*\*  
\* WEIBULL SLOPE : 6.93  
\* CHARACTERISTIC LIFE : 1445.28

NO.	RELIABILITY(%)	TIME
1	90	<u>1044.6863 K</u>

TI-NHTSA 011226

-MSG M#- 17608 FR-SB01 TO-JW02 SENT=09/02/92 01:35 PM  
RF-148 ST=C DIV=0050 CC=00101 BY-SB01 AT=09/02/92 01:35 PM

TO: Danny O'Driscoll DOD Jeff DiDomenico DIDO  
CC: Mark Bissell AEB3 Jim Watt JW02  
Dave Csarn ZARN Paul Westerlind B511  
Chris Wagner CDW3 Capri  
PR: Steve Offiler SB01 (7795) 16-1 K-8/ANSR!  
SJ: High Vacuum CCPS for Ford Australia 943A-9P934-R6

Danny has pointed out that we neglected to comprehend the Tier-1 supplier in meeting the PPB date of Nov. 1. BCIA (Brake and Clutch Industries of Australia) would like to receive our parts by Oct 16. I have pulled two weeks out of the previous schedule by shortening the total time available for the prototype experiment from 3 weeks to 2 weeks and eliminating the "sanity check" Impulse test (thus relying solely on the results of the experiment to give confidence in the chosen design). Also, I'm back to assuming use of black O-rings with a subsequent changeover to colored O-rings at a later date. I have not shifted the date of ES submission.

Prepare vac/dia life test proto lots (obtain rubber diaphragms - PURCH) (obtain model shop parts)	Now - Sep 06 Sep 4 Sep 4
Build prototype lots	Sep 08 - Sep 09
Complete vacuum work (Druck, MARK B.)	Sep 09 - Sep 10
Run dia. life proto experiment	* Sep 11 - Sep 25 (was 3wk)
Impulse on prod. intent (sanity check)	* N/A
Pigmented or striped O-rings (Obtain - CHRIS / PURCH)	* T-B-D (was Oct 14)
Build parts (PPB samples & validation)	Sep 29 - Sep 30
PPB Samples leave US	Oct 02
Val: Init char	Oct 05 - Oct 07
Val: Impulse & FC	Oct 16 - Oct 28
PPB Samples arrive Australia - BCIA	Oct 16
Val: Compl. of final char	Nov 06
Val: Draft ES report	Nov 09 - Nov 16
ES report leave USA	Nov 20
ES report arrive Australia	Nov 30

Regards, Steve O.

TI-NHTSA 011227

## PRESSURE SWITCH DATA

FORM 21605

TEST NO. 324-15-48

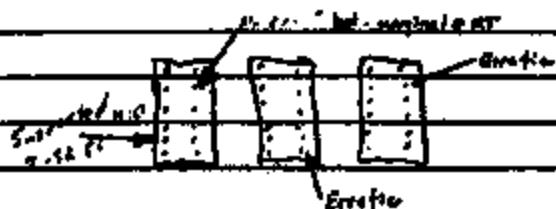
DEVICE CCPS	DATE REQUESTED 9/1/93	REQUESTED BY Steve Officier	REQUESTED COMPL. DATE
PERFORMED BY Testax Diagnostics	DATE STARTED 9/1/93	DATE COMPLETED	APPROVED BY
PROJECT TITLE: Speed Control PS			

CUSTOMER:

PURPOSE OF TEST: characterization of diaphragm life vs. Vacuum  
and prototyping various possible solutions if a problem is indicated.

PROCEDURE: Next pages

NOTE: This is the first lifetest to use the automatic end-of-test feature on frigidaire  
# Counter has turned over. Add 8KK to Count!



Device #	Vacuum	TEST	application	avg. to Failure	Remarks			
324-15-01	AIR	Cyclic	constant	1377K				
-62			100%	1659K	gentle			
-73				1352K	VERY SLOW LEAK			
-74				1324K				
-85				1512K				
-96				1571K				
324-15-67	yes	cyclic	none	1619K	Gentle Switch			
-76				EVER				
-89				1774K				
-10				1219K	T-diff?			
-11				1681K				
-12	▼	▼	▼	1555K				
324-15-17	yes	cyclic	none					
-74	▼	▼	▼					
324-15-18	yes	burst/over none						
-18	▼	▼	▼					
324-15-19	yes	cyclic	variable	970K				
-19			Max	944K				
-19			Min	952K				
-20				1324K				
-21				1210K				
-22	▼	▼	▼	1380K	60Hz/1			

TH-NHTSA 011228

### fact sheet

Play      Practice

Time 1955-08 0955-08

John 246 016

*Scans* .033 .033

~~Music~~ .453 .046

\_\_\_\_\_

\_\_\_\_\_

—  
—  
—

TI-NHT9A 011229

## PRESSURE SWITCH DATA

Page 21605

TEST NO. 324-15-48

DEVICE CC15	DATE REQUESTED 8/1/93	REQUESTED BY SFC, OFFICER	REQUESTED COMPL. DATE
PERFORMED BY Jeffrey A. Rosenzweig	DATE STARTED 8/1/93	DATE COMPLETED	APPROVED BY
PROJECT TITLE: Speed Control PS			

CUSTOMER:

PURPOSE OF TEST: Characterization of displacement vs. Vacuum  
and prototyping various switch solutions if a problem is indicated.

PROCEDURE: Next pages

# Counter has turned over. A 1kk to Count

next



Device #	Vacuum	Test	Condition	Failure	Remarks		
324-15-01	100	cycle	open	1377			
-02			open	1619 K			100% low
-03				1552 K	2. DEAD ST 1388 K	B = 140	4.937
-04				1329 K	very slow	B = 162	1445.3
-05				1512 K			
-06				1571		B = 190	1044.7 K
324-15-07	yes	cycle	open	1619 K	crash		
-08				936 K			100% low
-09				1702 K	2. DEAD ST 846 K	B = 4.6	2.871
-10				1219 K	T-24	B = 170	1348.3
-11				1691 K			
-12				1555		B = 180	501.2 K
324-15-13	yes	cycle	open				
-14							
324-15-15	yes	multiple	open				
-16	*	*					
324-15-17	yes	cycle	closed	870 K			
-18			open	944 K			86.5
-19			closed	953 K	5. DEAD ST 870 K B = 147		
-20				1324 K			
-21				1210 K			
-22				1580	Cryost		

TIA-NHTSA 011230

Device #	Version	Test	multifunction failure	Comments			
324-15-32	v65	open	Engel				
-34		+					
324-15-35	v65	cycle	Miller				
-36		+	Engel				
-37		+					
-38		+			① DEAD		
-39		+					
324-15-31	v65	open	Miller				
-30		+	Engel				
324-15-36	v65	cycle	Engel				
-39		+	Miller				
-40		+			① DEAD		
-41		+					
324-15-39	v65	open	Full load				
-40		+	Engel				
324-15-41	v65	cycle	Miller	199K			
-43		+		1226K			
-47		+		1334K	5-1690 ST ② 644K ③ 1209		GE 5.5
-48		+		1447K			
-49		+		1985K			
-51		+		199K			
324-15-43	v65	open	Miller				
-44		+					
199K	data	test					
199K	series						
199K	parallel						
0	.00675A	.00175A					
199K	.006	.016					
599K	.037	.072					
19.99K	.053	.091					

TI-NHTSA 011231

-MSG MN= 59390 FR=VAGG TD=PCME SENT=09/03/92 02:48 PM  
RM=186 ST=C DIV=00050 CC=00134 BY=VAGG AT=09/03/92 02:48 PM

TO: DICK GARIETY MPPC

CC: ED SMITH PCQA BILL SWEET PCME  
STEVE OFFILER SBO1 ELAINE ROSE GAMY  
DENNIS NATALE PCME DAVE CZARN ZARN  
ANDY MCQUIRK PCQA CHARLIE DOUGLAS CPPC  
STAN HOMOL SHZ GARY SNYDER GJS1  
TOM CHARBONEAU TC JIM WATT PCQA  
NORM FREDA WHLZ TOM BURKE MPPC  
BOB BARTOSH PCQA

FR: MATT SELLERS PCME

SJ: IMPULSE TESTING ON 77PS

DICK,

THERE ARE OTHER THINGS BESIDES DISC ENDURANCE THAT ARE EVALUATED DURING THE IMPULSE CYCLING OF THE 77PS DEVICES IN PARTICULAR WHICH SHOULD HAVE BEEN EVALUATED PRIOR TO MAKING THE DETERMINATION THAT IMPULSE TEST FREQUENCY COULD BE REDUCED FROM EVERY LOT TO ONCE EACH WEEK. I REFER MOST STRONGLY TO SPRING LIFE ON THE 77PS PRODUCT. WE HAVE DETECTED IN THE PAST INSTANCE WHERE AN IMPROPERLY BENT SPRING ARM, WHICH PASSES ALL SPC REQUIREMENTS, WILL BREAK DURING IMPULSE CYCLING. FOR THIS REASON ALONE I WOULD ADVISE NOT REDUCING THE IMPULSE TEST CYCLING FREQUENCY ON 77/87PS AT THIS TIME. PERHAPS THERE IS A COMPROMISE FREQUENCY THAT WOULD BE MUTUALLY AGREEABLE TO THE ENTIRE MRB. ADDITIONALLY, THERE ARE OTHER CONCERN AS FOLLOWS:

- 1.) THIS MAY CONSTITUTE A CHANGE TO THE MANUFACTURING CONTROL PLAN WHICH WOULD REQUIRE CUSTOMER APPROVAL.
- 2.) D&P FMEA'S REFER TO THESE CYCLING AUDITS AS ONE OF SEVERAL MANUFACTURING CONTROLS FOR SEVERAL PROCESSES.
- 3.) THE MRB TEAM MAY NEED TO REVIEW OTHER AREAS OF THE D&P FMEAS TO DETERMINE IF ANY OTHER ADVERSE IMPACTS MAY RESULT.

PLEASE CONSIDER THESE ITEMS AND RESPOND URGENTLY TO MYSELF AND THE REST OF THE MRB.

REGARDS...MATT  
X1245

TI-NHTSA 011232

-MSG MH= 77135 FR=VAGB TO=COPY SENT=09/04/92 07:38 AM  
ST=C DIV=0050 CC=00134 BY=VAGB AT=09/04/92 07:38 AM

TO: 1ST SHIFT SMWT MPPC 04 SEPT, 1992  
2ND SHIFT SMWT MPPC

CC: CLAIRE BALTHAZAR PCME  
DICK DARIEPY MPPC  
TOM BURKE MPPC  
BILL SWEET PCME  
STEVE OFFILER SBOI  
DAVE CZARN ZARN

FR: MATT [REDACTED] PCME

REI 77PS AND 87PS IMPULSE TESTING

=====

77/87PS SMWT'S.

INFORMATION THAT MAY HAVE BEEN PREVIOUSLY PASSED ON TO YOU CONCERNING THE REDUCTION OF IMPULSE TESTING FROM "EVERY DISC LOT" TO "ONE LOT EACH WEEK" HAS BEEN RESCINDED. YOU HAVE BEEN, AND ARE STILL REQUIRED TO, IMPULSE TEST EVERY DISC LOT PRIOR TO RELEASING THE LOT FOR BUILD/SHIP. AT TIMES, WHEN PRODUCTION DEMAND REQUIRES, YOU MAY BE AUTHORIZED TO BUILD THE SENSOR LOT PRIOR TO RECEIVING FORMAL IMPULSE TEST ACCEPTANCE FROM Q.C. HOWEVER, NO LOT MUST EVER BE PACKED WITHOUT FIRST RECEIVING FORMAL IMPULSE TEST RESULTS FROM Q.C. AUTHORIZATION TO BUILD SENSOR LOTS PRIOR TO COMPLETION OF IMPULSE TEST CAN ONLY COME FROM ENGINEERING OR MANUFACTURING MANAGEMENT. BUILDING SENSOR LOTS WITHOUT IMPULSE TEST RESULTS IS CONSIDERED A SPECIAL SITUATION AND DOING SO WITHOUT FIRST RECEIVING AUTHORIZATION FROM ONE OF THESE SOURCES IS PROHIBITED.

REGARDS...MATT  
X1245

TH-NHTSA 011233

-MSG MH= 77526 FR=RWG3 TO=~~RWG2~~ SENT=09/04/92 08:01 AM  
RH=030 ST=C DIV=0050 CC=00127 BY=RWG3 AT=09/04/92 07148 AM

To: CCPS SELF-DIRECTED TEAM MFPC

Copy: ALAN SADLER AMS2 GARY J. SNYDER DJS1  
DAVID CZARN ZARN

From: DICK GARIEPY RWG3

Subj: IMPULSE TESTING

MATT SELLERS HAS RAISED A VERY IMPORTANT ISSUE RELATIVE TO THE IMPULSE TESTING DONE ON 77PS DEVICES. THAT IS, IN ADDITION TO TESTING FOR DISC PROBLEMS THE IMPULSE ALSO IS OUR ONLY METHOD OF INSURING THE CONTACT ARM WAS FORMED PROPERLY.

THEREFORE WE WILL CONTINUE THE IMPULSE TESTING ON 77PS DEVICES ACCORDING TO PREVIOUS PROCEDURE. A NEW DEFINITION OF TEST FREQUENCY "LOT SIZE" IS BEING CONSIDERED.

DICK

TI-NHTSA 011234

-MSG NO= 00451891 TO=SBC01 FROM=SBC01 SENT=09/18/92 08:09 AM  
SI=033 ST=C DIV=0050 CC=00101 BY=SBC01 AT=09/18/92 08:09 AM

TO:	Jeff DiDominico	DIDC	Danny O'Driscoll	DOO
	Charlie Douglas	CMDP1	Matt Sallars	PCME
	Born Freda	WELA	Bill Sweet	WS4
	Dick Gariepy	MFPC	Chris Wagner	CDW3
	Dennis Natale	DJNL	Jim Watt	PCQA
CC:	Dave Czarn	(delivered separately)		
FR:	Steve Offiler	SBC01		

SJ: Vacuum/Diaphragm Life Status Report - 52/57/77/87PS Sensor

Six lots of 6 devices each began cycling last week. The devices are 77PSL2-1's, and the lots are composed of:

- \* Standard diaphragm configuration, no vacuum
- \* Standard diaphragm configuration, vacuum (0.4 mbar)
- \* Roughened hexport flange, vacuum
- \* Rubber diaphragm, vacuum
- \* Full-round Kapton, vacuum
- \* Tefzel (fluid layer) plus two Kapton, vacuum

The first two lots will be directly compared to determine the impact of high vacuum spec'd by Ford Australia on diaphragm life (if any). The other four are experiments which can be compared with the second lot to determine a direction for design modifications (if required).

To-date, the test has progressed to about 1230K cycles. There have been NO failures in the lot of standard config/no vacuum, versus two in the vacuum lot (846 and 1219K). This data, while preliminary, is supporting the hypothesis that high vacuum ultimately results in shortened diaphragm life due to Kapton displacement and wrinkles. There have been four failures in the roughened hexport flange lot (870, 944, 952, and 1210K) and four failures in the Tefzel lot (844, 846, 992, and 1226K). There have been NO failures yet in the rubber diaphragm lot, or the full-round Kapton lot.

Extra test devices were built for visual analysis of the diaphragm in the as-built condition, which is done by cutting off the box while retaining the crimp. As expected, the standard configuration parts w/ vacuum show general looseness and wrinkles across the Kapton. The roughened hexport flange lot shows some specific localized wrinkling. The rubber diaphragms, with and without vacuum, look identical - the rubber is protruding upward in a dome shape, which seems to be caused by the compression at the edges of the diaphragm. This is apparently not due to vacuum displacement. The full-round Kapton parts look quite good - no displacement from exposure to vacuum. The Tefzel parts look similar to the standard config. parts; their relatively early failure is likely to be caused by Tefzel's highly directional properties with sharply reduced elongation in the transverse direction.

Regards, Steve O.

TI-NHTSA 011235

## PRESSURE SWITCH DATA

FORM 21605

TEST NO. 324-15-41

DEVICE CCP5	DATE REQUESTED 9/4/93	REQUESTED BY Steve OFF:lev	REQUESTED COMPL. DATE
PERFORMED BY Testsys Div. Inc.	DATE STARTED 9/8/93	DATE COMPLETED	APPROVED BY
PROJECT TITLE: Speed Control PS			

CUSTOMER:

PURPOSE OF TEST: characterization of diaphragm 170 vs. Vacuum  
and prototyping various possible solutions if a problem is  
indicated.

PROCEDURE: Next pages

# Counter has turned over. A 1kk to Count!



Device #	Vacuum	Test condition	Failure	Remarks		
324-15-01	110	Cycle	Normal	1377		
-02			NO F	11574		100% LOW
-03				1352 K	2 DEAD ST	1324 K P=160 6.933
-04				1324 K		0 = 1132 - 1405.3
-05				1512 K	very slow	
-06	↓	↓	↓	1571		P=90 = 1044.7 K
324-15-07	yes	Cycle	Normal	1619 K	run fine	
-08				876 K		100% low
-09				1782 K	2 DEAD ST	846 K P= 4.6 2.251
-10				1219 K	T-54	0 = 1198 - 1362.3
-11				1601 K		
-12	↓	↓	↓	1555		P=90 = 10 - 501.3 K
324-15-13	yes	open	open			
-14	↓	↓	↓			
324-15-15	yes	burst mode	normal			
-16	↓	↓	↓			
324-15-17	yes	Cycle	Normal	830 K		
-18			NO F	944 K		0=6.5
-19			Range	952 K	6 DEAD ST	870 K P=103
-20				1529 K		
-21				1210 K		
-22	↓	↓	↓	1380	6x90°	

TINHTSA 011238

Device #	Vision	Test	Condition	Notes	Comments
324-15-37	yes	open	closed		
-34	no				
324-15-35	yes	cycle	either		
-31	-		display		
-32					
-33					O. DEAD
-34					
-35					
324-15-34	yes	open	either		
-36	no		display		
-37					
-38					O. DEAD
324-15-39	yes	open	full and		
-39	no		negative		
324-15-40	yes	cycle	either	✓ 8168	
-40	-			✓ 122646	Bc 5.5
-41				✓ 1324-46	✓ 10410 ST 10 044K Q=12.09
-42				✓ 1442E	
-43				✓ 4948	
-44				✓ 3494K	
324-15-47	yes	open	both		
-45	no				
109E alarm test					
Time	0000	00:00			
0	0000000	0000000			
1000	.010	.010			
5000	.037	.037			
10000	.073	.073			

-MSO M#= 506728 FR=SB01 TO=MJ82 SENT=09/21/92 10:46 AM  
R#=051 ST=C DIV=0050 CC=00101 BY=SB01 AT=09/21/92 10:46 AM

TO: Dave Czarn ZARN Matt Sellers MJ82  
Jeff DiDomenico DIDO Bill Sweet WB4  
Dick Barriepy RW03 Chris Wagner CDW3  
Dennis Natale DJN1  
  
CC: Charlie Douglas CMPI Danny O'Driscoll DOD  
Mike Downey MIKD  
  
FR: Steve Offiler SB01  
  
SBJ: Vacuum/Diaphragm Life

The characterization test of high vacuum (for Ford Australia) combined with diaphragm life testing is reaching a point where some conclusions can begin to be made. It is imperative that we give this matter due attention ASAP, because the data suggests a diaphragm design change is in order to increase the device's resistance to high vacuum. There will, of course, be manufacturing ramifications. Furthermore, samples which MUST use the production-intent configuration will be built and shipped next week. Thus, we'll be locking ourselves into a particular design configuration very soon.

\*\*\*\*\*  
A meeting to discuss the interim results of ongoing testing, and the design direction, will be held in the cafeteria on Tuesday 92-09-22 at 4:00 pm.  
\*\*\*\*\*  
Regards, Steve O.

- \* Base volume vs. volume which needs vacuum resist.  
Cipriani + Falcon.
- \* When were we made aware of this spec.  
It is written into my documents for our review.
- \* Replace Keptron or Gasket or both.
  - Gasket → Yes,
  - Keptron → maybe 1 or 2 eliminated short term.  
→ maybe 3 eliminated if we receive cloth backed
- \* Performance = Temp extremes.
  - Drift.
  - Initial
- \* Volume : 23k annualy. Hard line?
- \* Gary Smidler input.
- \* Ship 300 OCT 1<sup>st</sup> Prod Intent.

\* Cost! Open  
Check book or the  
there constraints.

TINHTSA 011238

27713-1 Cup Modification 9-22-92

Production

Problem: In-sufficient preload placed on Quiet Pass Car devices. Problem manifest itself AS A loose, rattling disc after sensor crimping on the AMI.

Affect: Yield loss for low pin on F.A.M.  
27 to 52.

Root

Cause: General Stack-up error between Cup, Converter, and disc. Disc envelope is too large (height) by appx .004" to .005"

Solutions

Evaluated:

Stepped washer, bump on converter, tapered converter, spacer between washer + converter, increasing bump height on Cup, lowering washer shelf on Cup Increase in O.P. @ SAM, increase in crimp pressure @ SAM

Solution Selected:

Lowering washer shelf on Cup  
\$5K tooling upgrade from Valentine.

# Rubber Diaphragm

9-23-92

## Preliminary Cost Analysis

Components : Add

Delete

→ Rubber diap.

→ Internal Gasket

25 ft

2 ft

Operations : Add

→ Manual Sensor ASSY

4 x 1.8 (7.2)

→ AMI ASSY

1 x 1.0

\$10.81/hr + \$7.02/k

Yields : Due to different methods of ~~screwing~~  
the keypad, it is questionable what  
the impact will be on lot size and  
~~x~~ repeatability.

600 each.

5% Loss of Func Test = \$ 30/k

Volume : 23k annually

Cost Summary      Material .005

                        Labor .07

                        Yields .03

.105

TO: Tom Charboneau TC Dennis Natale DJN1  
[REDACTED] Danny O'Driscoll DOD  
Jeff DiDomenico DIDO Matt Gellers MJS2  
Charlie Douglas CMP1 Gary Snyder GJS1  
Mike Downey MIKD Bill Sweet WS4  
Dick Gerlipy RWG3 Chris Wagner CDW3  
Andy McKenna SPRT

(File)

FROM: Steve D'Filler SBC01

SUBJ: Vacuum/Diaphragm Life Testing 57/77PS Brake Sensors

A meeting was held to discuss the interim results of the combined high-vacuum / diaphragm life test, which is being run specifically to address Ford Australia's need for high vacuum withstand, and generally to explore options to improve diaphragm life.

Two lots use the standard (orake) Kapton diaphragm configuration, one exposed to 0.4 mBar vacuum, the other no vacuum. It is known that the vacuum causes diaphragm displacement and gives rise to wrinkles, which have previously been shown to have an adverse effect on life. Also previously shown is the fact that some unknown combination of design and/or process variables causes brake diaphragm life to be quite erratic, with some test lots performing very poorly (significant failures well below the customer spec. of 300K cycles) while other test lots built at a different time may show no failures until over 1KK cycles.

Two facts come out of this: one, results of a given test are only valid relative to other lots built with the same parts at the same time, and two, in general we need to have a big-picture goal of finding a more robust diaphragm configuration.

Interim results show the standard-config dia. no vac. has 5/6 failures, but please note the significant fact that the failures are tightly grouped between 1324K and 1571K cycles. Once completed, a Weibull analysis will show a high beta (steep slope) which is desirable. The vacuum parts have only 3/6 failures, but these are widely spaced from 846K to 1555K, which will produce a low beta. These results interpreted casually may suggest the vacuum-exposed parts are not all that bad. Nonetheless, given the visual observations of the wrinkles in the vacuum parts, combined with a much poorer beta, plus the fact that they began to die much sooner, confirms my hypothesis that we have a problem with vacuum exposure which is significant enough to require immediate attention. Furthermore, bear in mind that Danny has been preparing the customer for this as well, opening a window of opportunity for us to make and validate design/process changes, and gain manufacturing experience with it due to the low volumes of the Australia application.

-REC ID= 00575252 FR=SM01 TO=SM01 SENT=09/23/92 08:32 AM  
SJ=040 ST=C DIV=0050 CC=00101 ST=SM01 AT=09/23/92 08:32 AM

TO:	Tom Charbonneau	TC	Dennis Natale	DJW1
	Dave Czaja	ZANN	Danny O'Driscoll	DOD
	Jeff DiBenedetto	DIDO	Matt Sellera	MWS2
	Charlie Douglas	CMP1	Gary Snyder	GJS1
	Mike Downey	MIRD	Bill Sweet	BS4
	Dick Gariety	MRG3	Chris Wagner	CDW3
	Andy McKenna	SPRE		
FR:	Steve Offiler	SM01		

SJ: Vacuum/Diaphragm Life Testing 57/7799 Brake Sensors

A meeting was held to discuss the interim results of the combined high-vacuum / diaphragm life test, which is being run specifically to address Ford Australia's need for high vacuum withstand, and generally to explore options to improve diaphragm life.

Two lots use the standard (brake) Kapton diaphragm configuration, one exposed to 0.4 mbar vacuum, the other no vacuum. It is known that the vacuum causes diaphragm displacement and gives rise to wrinkles, which have previously been shown to have an adverse effect on life. Also previously shown is the fact that some unknown combination of design and/or process variables causes brake diaphragm life to be quite erratic, with some test lots performing very poorly (significant failures well below the customer spec. of 500K cycles) while other test lots built at a different time may show no failures until over 1MK cycles.

Two facts come out of this: one, results of a given test are only valid relative to other lots built with the same parts at the same time; and two, in general we need to have a big-picture goal of finding a more robust diaphragm configuration.

Interim results show the standard-config dia. no vac. has 5/6 failures, but please note the significant fact that the failures are tightly grouped between 1324K and 1371K cycles. Once completed, a Weibull analysis will show a high beta (steep slope) which is desirable. The vacuum parts have only 3/6 failures, but these are widely spaced from 546K to 1555K, which will produce a low beta. These results interpreted casually may suggest the vacuum-exposed parts are not all that bad. Nonetheless, given the visual observations of the wrinkles in the vacuum parts, combined with a much poorer beta, plus the fact that they began to die much sooner, confirms my hypothesis that we have a problem with vacuum exposure which is significant enough to require immediate attention. Furthermore, bear in mind that Danny has been preparing the customer for this as well, opening a window of opportunity for us to make and validate design/process changes, and gain manufacturing experience with it due to the low volumes of the Australia application.

Of the four experimental lots, we have 6/6 failures in both the Tefzel lot and the roughened-harport-flange (increased friction w/ dia.) lot, all between roughly 550K and 1450K. No further efforts are planned for these lots. In comparison, we have NO failures in either the rubber diaphragm lot or the full-round Kapton lot with the test at roughly 1600K presently. These two lots will immediately be evaluated further.

Engineering concerns include:

\* calibration shift at temperature extremes - 40°C / 175°C

TI-NHTSA 011244

- \* Thermal cycling performance
- \* calibration drift over life
- \* high-temperature aging effects on calibration

Manufacturing concerns include:

- \* effects on lot piloting and sigma's
- \* cost
- \* visibility of hand-line vs AMT assembly; req'd tooling mod's

We plan to build ASAP fairly sizable quantities of rubber diaphragm, full-round Kapton, and control lots, using the 57PS hand line. All parts will be checked using the 77PS automatic pressure tester, for calculation of piloting and sigma info. These parts will be subdivided into test lots to undergo each of the four items listed under engineering concerns above. Mfg. Eng., Mkt., and Field Sales will need to put some thought into the cost issues.

From a schedule/program standpoint, we must ship 300 production-intent parts on Oct 2 in order to be received in Australia at the Tier 1 (SCIA) in time for them to deliver completed brake systems to Ford for a pre-production build Nov 1. These dates are cast in stone; furthermore, the requirement that these be production-intent will serve to cast in stone whatever diaphragm configuration we choose to send.

Regards, Steve C.

\* BUILD TEST LOTS — USING SILENT DISCS, NYRYL BASES, CAR 369003

\* RUN ON AUTO P.T. — DATA FOR X 5

\* TESTS :

— 1) CAL SHIFT & TEMP (DAYS OUT) 1C + STAGES

CONVENTIONAL AT ~6 OR SO X 3 LOTS

RT CHAR ; LOW -40 HI +175°C

2) RE-CHAR ~6 X 3 LOTS

0.1% H<sub>2</sub>

DAMS TO -40, SOAK ~ 30-60 min

UP TO +121°C SOAK ~ 30-60 MIN

RT-CHAR

3) DRIFT OVER LIFE PROG CYCLE

PRE-CHAR 0 X 3 LOTS

RUN 250K 1 1/2 LOTS ; CHAR

COMPL. 500K, CHAR

RUN 250K OTHER 1 1/2 LOTS ; CHAR

COMPL. 500K, CHAR

4) HI TEMP AGING — MANIFOLD, PRE-FILL w/ FLUID(DIRT)  
SIT @ +150°C FOR X DAYS

VACUUM / DIA LIFE STATUS

TIRE 9-22 @ ~ 1.5 KEL

COTTER

- 1 # RUBBER DIA } 0/6
- 2 # FULL-AND RAYTON }
- 3 # STD NO VAC ? 4/6 @ 1.32, 1.33, 1.38, 1.51 KEL
- 4 # STD w/ VAC ? 2/6 @ 0.0, 1.3 KEL
- 5 # TEFZEL } 6/6 @ 0.0 - 1.4 KEL
- 6 # ROUGH NBR PLANGE }

→ DIAING ~ 20/04  
 → HANDBR  
 → PULL AND  
 → COTTER  
 → CHARACTERISTIC KEL 200  
 → 101/02L, &  
 → → TRY CHARGE TEMP  
 → → TRY TC  
 → → TRY DRIFT  
 → → AAGING TEST (41 TEMP)

RUBBER DIA

- MPG CONCERN'S : 1. PICE-AND-PLACE  
 2. SMALL LIP FROM CENTER OF BEARING  
 3. EFFECT ON PILOTTING + D

- PERF CONCERN'S : 1. EFFECT ON CHARGE @ TEMP DATA.  
 2. TC LIFE  
 3. EFFECT ON DRIFT

FULL- AND

- MPG CONCERN'S : 1. NEW RADIAL TOOLING SWING-CHG  
 -OR-

2. NEW RADIAL TOOLING APPROX ON ALL 62- }  
 57-77 }  
 3. NEW PRESENCE - CHG  
 4. EFFECT ON PILOTTING + D } HAND-  
 LINE

- PERF CONCERN'S : 1. EFFECT ON CHARGE @ TEMP DATA.  
 2. DRIFT UNKNOWN

-MSG NO- 16807 FR-SBO1 TO-GAMY SENT-09/24/92 08:17 AM  
RF-006 ST-C DIV-0050 CC-00101 BY-SBO1 AT-09/24/92 08:17 AM

TO: Dave Csarn SARN Dennis Natale DJW1  
Jeff DiDomenico DIDO Danny O'Driscoll DOD  
Charlie Douglas CMPI Elaine Rose GAMY  
Mike Downey MIND Matt Sellers MJ52  
Dick Garispy RWC9 Chris Wagner CDW3  
Stan Hosol SH2 Jim Watt JW02

FR: Steve Officer SBO1

SU: High-Vacuum CCPS for Ford Australia 7766-1 Capri

New information received from Danny this morning indicates we've got some additional time before we need to ship the PPS samples to SCIA. We'll use this time to complete a series of tests to confirm the chosen design, and to work out many details relating to manufacturing and cost. Results of the high-vacuum experiment have pointed definitively at either the full-round Kapton, or the rubber diaphragm, both of these lots having experienced NO failures thru about 1.7 million cycles; versus the standard diaphragm configuration, with or without high vacuum exposure, where all devices have died. Below is an updated schedule for sample builds and validation.

Prepare vac/dia life test proto lots

COMPL

Build prototype lots

COMPL

Complete vacuum work (Druck, MARK B.)

COMPL

Run dia. life proto experiment

Sep 11 - Sep xx ONGOING

In-house design confirmation

Sep xx - Oct 16

Cal shift at temp extremes

Thermal cycling

Drift over life

Hi temp aging

Pigmented or striped O-rings

Oct 16

Build parts (val; PPS optional)

Oct 19 - Oct 20

Val: Init char

Oct 21 - Oct 23

Val: Impulse & TC

Oct 26 - Nov 09

Build parts (PPS latest)

Oct 27 - Oct 28

PPS Samples leave US

Oct 30

Val: Compl. of final char

Nov 10 - Nov 13

PPS Samples arrive Australia - SCIA

Nov 13

Val: Draft ES report

Nov 16 - Nov 18

ES report leave USA

Nov 20

ES report arrive Australia

Nov 30

Regards, Steve O.

TI-NHTSA 011247

MSG ID= 20895 FR=9501 TO=MJS2 SENT=09/24/92 11:04 AM  
R#-060 ST=C DIV=00050 CC=00101 BY=9501 AT=09/24/92 11:04 AM

TO: Dave Czarn ZARN Dennis Natale DJN1  
Jeff DiDomenico DIDO Danny O'Driscoll DOD  
Charlie Douglas CMP1 Matt Sellers MJS2  
Andy McKenna SPRT Chris Wagner CDW3

FR: Steve Offiler 8801

SJ: Vacuum/Diaphragm Life Test Update

At this point, ALL devices have failed in four of the test lots. These are the standard dia. config with vacuum, standard config. without vacuum, the Tefzel experimental lot, and the roughened hexport flange experimental lot. THE REMAINING TWO EXPERIMENTAL LOTS (rubber diaphragm and full-round Kapton) HAVE EXPERIENCED NO FAILURES YET. The test is at 1.8KK cycles presently. A quick Weibull analysis shows the following Beta and Theta (slope and characteristic life):

Dia config	Vac	Beta	Theta
Standard	no	14.0	1522
Standard	yes	4.6	1598
Rough flange	yes	6.5	1197
Tefzel	yes	5.3	1209

\* 1866 Full round failure

Note both the rough flange and the Tefzel performed poorly, comparing characteristic life with the standard config (control) lots. Also note that, comparing the two standard config's, vacuum has the effect of lowering Beta (slope) which is a measure of the predictability of the failures; the failure of the vacuum parts is much less predictable. We are continuing the test, waiting for a failure to occur in either the rubber dia. or the full-round Kapton. Previously, we've seen about 30% better characteristic life for full-round vs. standard config. We have no previous data for rubber dia.

Regards, Steve O.

\* Round kapton. 3 layers.

\* Rubber dia. 3 layers. MAY be able to reduce the # of layers. Needs to be scheduled into the test plan

TI-NHTSA 011248

-MSG ID= 00025895 FR=SB01 TO=SB01 SENT=09/24/92 11:04 AM  
SI=049 ST=C DIV=0050 CC=00101 BY=SB01 AT=09/24/92 11:04 AM

TO: Dave Czarniak ZARM Dennis Matale DOW1  
Jeff DiDomenico DIDO Danny O'Driscoll DOD  
Charlie Douglas CMPI Matt Sellars MJE2  
Andy McKenna SPRT Chris Wagner COWS

NR: Steve Offiler SB01

SJ: Vacuum/Diaphragm Life Test Update

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Regards, Steve O.

TI-NHTSA 011249

## PRESSURE SWITCH DATA

Form 21605

TEST NO. 332-15-66

DEVICE <u>77PS</u>	DATE REQUESTED	REQUESTED BY <u>Steve Offley</u>	REQUESTED COMPL. DATE
PERFORMED BY <u>Jeffrey DiDomenico</u>	DATE STARTED <u>7/20/925</u>	DATE COMPLETED	APPROVED BY
<u>PROJECT TITLE: Speed Control Reactivate PS</u>			

## CUSTOMER:

PURPOSE OF TEST: Evaluation of Rubber Diaphragms/Round Kapton

## PROCEDURE:

Color Code: Black = control

Blue = Rubber Diaphragm

Red = Round Kapton

L = .004" offset

R = .009" offset

All devices subjected to .0006 PSIA vacuum for 60 sec.

Device #	Test	Size	Offset	INIT	Test Vacuum				Final	Remarks
					Avg	Avg	Avg	Avg		
332-15-0	Thermal control		.004"	126 103	122	102	102	102	121	111°C
	-01	Cycle		127 97	128	104	102	97	124	111°C
	-03			125 97	125	98	97	97	124	111°C
332-15-05			.007"	158 142	160	142	139	140		
	-05			150 103	159	104	101	101		
	-06			177 152	177	156	147	147		
332-15-07	Rubber		.004"	148 105	146	104	103	102	147	114°C
	-08	Diaphragm		145 105	137	106	102	97	141	113°C
	-09			139 103	135	104	109	96	134	112°C
332-15-10			.009"	126 142	126	142	136	140		
	-11			130 156	128	148	142	145		
	-12			139 142	136	141	138	142		
332-15-11	Round		.004"	156 135	158	136	125	109		
	-12	Kapton		162 129	163	130	126	109		
	-13			154 139	157	127	-	-		Work: N. Brantley 34/05/92
332-15-11			.009"	282 260	206	279	191	154		
	-11			211 189	217	182	150	119		
	-12			246 239	227	242	179	114		

(OVER)

TI-NHTSA 011250

## PRESSURE SWITCH DATA

Form 21605

TEST NO.

DEVICE	DATE REQUESTED	REQUESTED BY	REQUESTED COMPL. DATE																																																																																																								
PERFORMED BY	DATE STARTED	DATE COMPLETED	APPROVED BY																																																																																																								
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<p>High temp aging (150°C) started @ 11:00 AM 9/21/02            Test ended 9/21/02 @ 11:00AM. Total hrs: 456 hrs.</p> <table border="1"> <thead> <tr> <th>Device #</th> <th>Test Seal</th> <th>0.025"</th> <th>2A-1</th> <th>PST-100</th> <th>PST-1000</th> <th></th> <th></th> </tr> </thead> <tbody> <tr> <td>200-15-55</td> <td>High Temp Control</td> <td>#1</td> <td>131 104 132 99</td> <td>157 104</td> <td></td> <td></td> <td></td> </tr> <tr> <td></td> <td>- CL Agnt</td> <td>#</td> <td>126 102 127 99</td> <td>150 107</td> <td></td> <td></td> <td></td> </tr> <tr> <td>200-15-57</td> <td></td> <td>.007"</td> <td>156 138 161 135</td> <td>180 140</td> <td></td> <td></td> <td></td> </tr> <tr> <td>- 51</td> <td></td> <td>#</td> <td>191 173 193 179</td> <td>198 152</td> <td></td> <td></td> <td></td> </tr> <tr> <td>200-15-59</td> <td>Gaffer</td> <td>.007"</td> <td>145 109 145 H2 N.G.</td> <td></td> <td></td> <td></td> <td></td> </tr> <tr> <td>- 69</td> <td>Padlock</td> <td>#</td> <td>148 111 142 108</td> <td>132 112</td> <td></td> <td></td> <td></td> </tr> <tr> <td>200-15-61</td> <td></td> <td>.005"</td> <td>193 150 201 168</td> <td>195 151</td> <td></td> <td></td> <td></td> </tr> <tr> <td>- 62</td> <td></td> <td>#</td> <td>172 154 178 154</td> <td>190 146</td> <td></td> <td></td> <td></td> </tr> <tr> <td>200-15-63</td> <td></td> <td>.007"</td> <td>157 120 158 130</td> <td>170 137</td> <td></td> <td></td> <td></td> </tr> <tr> <td>- 64</td> <td>Steel Box</td> <td>#</td> <td>157 120 157 112</td> <td>171 139</td> <td></td> <td></td> <td></td> </tr> <tr> <td>200-15-65</td> <td></td> <td>.007"</td> <td>131 108 106 107</td> <td>187 161</td> <td></td> <td></td> <td></td> </tr> <tr> <td>- 66</td> <td></td> <td>#</td> <td>233 203 240 210</td> <td>244 192</td> <td></td> <td></td> <td></td> </tr> </tbody> </table>				Device #	Test Seal	0.025"	2A-1	PST-100	PST-1000			200-15-55	High Temp Control	#1	131 104 132 99	157 104					- CL Agnt	#	126 102 127 99	150 107				200-15-57		.007"	156 138 161 135	180 140				- 51		#	191 173 193 179	198 152				200-15-59	Gaffer	.007"	145 109 145 H2 N.G.					- 69	Padlock	#	148 111 142 108	132 112				200-15-61		.005"	193 150 201 168	195 151				- 62		#	172 154 178 154	190 146				200-15-63		.007"	157 120 158 130	170 137				- 64	Steel Box	#	157 120 157 112	171 139				200-15-65		.007"	131 108 106 107	187 161				- 66		#	233 203 240 210	244 192			
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TI-NHTSA 011292

## DIAPHRAGM REDESIGN TESTING

Results from initial testing of high-vacuum exposure combined with diaphragm cycling have shown two candidates for improving the design of the diaphragm: the rubber diaphragm (plus 3 pc. standard square Kapton) and the full-round Kapton diaphragm (3 pc.). This test compares various performance parameters of these two designs plus a control group.

Engineering tests include: thermal cycling, characterization at temp extremes, drift over life, and aging. Manufacturing tests include: statistical determination of act/rel average and standard deviation.

For engineering evaluations, build 1/2 at low preload extreme (4 mils) and the other 1/2 at high preload extreme (9 mils). To choose pin on each of these devices, measure sensor and base individually, but assume no variability in pin, and no crimp shift. For statistical evaluation, build all devices using normal production tooling and techniques, with the standard quiet passenger car preload value of 6.5 ~ 7.0 mils.

Perform a quick check on the eng. parts using the auto. pressure tester, to ensure functionality. Rebuild bad devices as necessary. All devices for eng. tests will ultimately be characterized using the normal manual lab technique. All devices for mfg. statistics should be checked on the auto pressure tester, collecting data for each device.

### EXPERIMENT SPECIFICS:

#### Thermal Cycle

- \* 3 low pin/3 high pin per lot x 3 lots (~18)
- \* Preferred, Cycler B (auto temp ctrl); or CCPS cycler
- \* Char act/rel
- \* Cycler at approx. 0.1 Hz continuous
- \* Temp from -40 C to +175 C, with soaks of 30-60 min (T-B-D, flexible) at each extreme
- \* Test completion point is T-B-D (based on failures ?)
- \* Final char act/rel, compare data for pre-post shifts, discrepancies related to preload, failures if any
- \* Disassemble, analyze (compr set, diaphragm condition)

SBO/920928/file DIAPHR.001

TI-NHTSA 011254

#### Calibration Shift at Temperature Extremes

- \* 3 low pin/3 high pin per lot x 3 lots (~18)
- \* Measure act/rel @ RT, -40 C, +175 C
- \* Run in any convenient oven - Delta (?)
- \* Compare data for shift magnitude relative to control lot, and discrepancies related to preload

#### Drift over life

- \* 3 low pin/3 high pin per lot x 3 lots (~18)
- \* Characterize act/rel
- \* Run in production cycler
- \* Run 1.5 lots (9) to 250K cycles, remove and char.
- \* Complete 500K cycles, char.
- \* Repeat on remaining 1.5 lots
- \* Draw graphs of act/rel vs. # cycles

#### High Temperature Aging

- \* 2 low pin/2 high pin per lot x 3 lots (~12)
- \* Run in Despatch oven
- \* Characterize act/rel
- \* Prefill devices, install on manifold and fill (optional, use fluid reservoir T-B-D, purposely entrap air, which will expand to pressurize devices slightly when heated)
- \* Soak at 150 C for as long as feasible
- \* Post-char, look for shifts
- \* Disassemble, examine diaphragm condition

#### Statistics

- \* Build remainder (approx. 75) from each lot, on 77 line, using all normal techniques
- \* Run each lot separately thru automatic pressure tester
- \* Collect data for each device individually
- \* Bad devices should be analyzed for failure mode; if not representative of production, toss data
- \* Calc. avg & std, histogram for each lot
- \* compare lots for avg (piloting concerns) and std (device variability)

Sensor Division

Control

Outer Diaphragm

Rand Kaptan

1	.0473	.0491	.0473
2	.0477	.0490	.0479
3	.0476	.0477	.0472
4	.0478	.0478	.0490
5	.0477	.0480	.0481
6	.0479	.0475	.0481
7	.0478	.0479	.0470
8	.0477	.0476	.0475
9	.0474	.0475	.0477
10	.0478	.0478	.0479
11	.0470	.0477	.0470
12	.0479	.0477	.0470
13	.0471	.0470	.0478
14	.0475	.0472	.0426
15	.0470	.0471	.0470
16	.0477	.0473	.0477
17	.0474	.0476	.0470
18	.0470	.0476	.0470
19	.0475	.0471	.0470
20	.0470	.0476	.0479
21	.0478	.0475	.0479
22	.0472	.0472	.0471

TEST NUMBER (GIANT CAR) 6.5 MILS

TIME 3.2.5 MILE 1.2 TDA

TIME ARE 4 AND 91 MILS

# T-Cycle

Cold

- \* approx. 0.1 Hz continuous cycling
- \* heat -40°C to +175°C, soak 30-60 min(flexible)

HOT

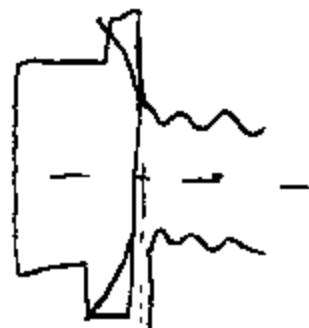
	1	2	3	4	5	6	7	8	9	10	11	12	13
	Date, Time	Temp	Count	Date, Time	Temp	Count		Date, Time	Temp	Count	Date, Time	Temp	Count
Cycle													
1	1/1 2:00P	35°C	0	1/1 4:05	-41°C	526		1/2 7:00A	30°C	526	1/2 8:35	175°C	1060
2	1/1 9:35A	175°C	10,60	1/1 10:10	-41°C	1810		1/3 11:10A	-41°C	1190	1/3 12:55	175°C	2970
3	1/1 12:55	175°C	2970	1/1 3:35	-41°C	3300		1/4 8:35A	-41°C	2700	1/4 4:50P	175°C	3774
4	1/1 4:50P	175°C	3774	1/1 9:40P	-43°C	5354		1/5 6:30A	23°C	5354	1/5 8:25A	175°C	6890
5	1/1 9:45	175°C	5940	1/1 10:05A	-41°C	6670		1/6 10:05A	-7°C	6690	1/6 10:55	175°C	7270
6	1/1 10:45	175°C	7270	1/1 3:55P	-45°C	9200		1/7 3:35P	-46°C	8200	1/7 6:50P	175°C	9200
7	1/1 7:00A	23°C	9200	1/1 8:35A	-71°C	9930		1/8 8:35A	-41°C	9930	1/8 10:30A	175°C	10,375
8	1/1 10:30A	175°C	10,375	1/1 10:55P	-41°C	11,390		1/9 10:55P	-1°C	11,290	1/9 7:25P	175°C	11,715
9	1/1 12:55	175°C	11,715	1/1 4:50P	-41°C	11,564		1/10 8:00A	20°C	12,014	1/10 8:00A	175°C	12,990
10	1/1 4:50A	175°C	11,990	1/1 10:20P	-40°C	13,780		1/11 10:20A	-30°C	13,780	1/11 10:55A	175°C	14,500
11	1/1 12:35	175°C	14,500	1/1 5:00P	-72°C	15,295		1/12 5:00P	-32°C	15,295	1/12 7:25P	175°C	15,715
12	1/1 7:25A	23°C	15,715	1/1 7:55A	-41°C	16,395		1/13 9:00A	-41°C	16,395	1/13 10:55P	175°C	17,890
13	1/1 1:45P	175°C	16,395	1/1 4:05P	-40°C	18,577		1/14 9:00P	21°C	18,077	1/14 9:00A	175°C	18,470
14	1/1 4:10A	175°C	18,470	1/1 10:55A	-45°C	20,165		1/15 9:00A	-7°C	20,265	1/15 12:55	175°C	20,391
15	1/1 12:35	175°C	20,591	1/1 3:00P	-40°C	21,296		1/16 3:00P	-40°C	21,296	1/16 5:15P	175°C	21,962
16	1/1 9:10P	23°C	21,297	1/1 10:10P	-41°C	22,555		1/17 9:55	-41°C	22,555	1/18 8:35A	175°C	23,075
17	1/1 10:30A	175°C	23,075	1/1 2:55P	-40°C	23,195		1/18 3:55	-48°C	24,775	1/18 9:20P	175°C	24,667
18	1/1 7:10A	20°C	24,667	1/1 3:00P	-40°C	25,950		1/19 2:00A	-40°C	24,750	1/19 10:55	175°C	25,930
19	1/1 1:00P	175°C	26,430	1/1 3:50P	-40	27,370		1/20 3:50P	-12.5	27,370	1/20 5:00P	175°C	26,972
20	1/1 7:55	21°C	27,372	1/1 4:50A	-41	28,974		1/21 10:00A	-41°C	28,974	1/21 0:25P	175°C	28,535
21	1/1 8:10	175°C	28,535	1/1 3:00P	-41	30,340		1/22 3:00P	-41°C	30,340	1/22 4:25P	175°C	30,761
22	1/1 7:15	156°C	30,341										
23													
24													
25													
26	T-S8-f1: Light	8,18,11,13,14,16	Note? These are connector numbers and do not match device number										
27													
28	Erroneous	8 < 143°C	15 = 16°C	FL Temp									
29													
30													
31													

25 sec on t 100 sec Hot cyc. #31  
25 sec on cold cyc. #31

COLD → HOT	0:45 / 0:50
HOT → COLD	11:45 / 12:00
RT → HOT	0:35
RT → COLD	11:15 / 11:30

## GROUP: 1100000 DIAPHRAGMS

DEVICE	ACT	REL	DIF	ACTCR	RELCR
65	0.0	127.8	-127.8		
60	0.0	130.6	-130.6		
11	119.3	97.7	21.6		
19	130.8	106.5	24.3		
51	131.9	105.4	26.5		
4	134.7	111.6	23.1		
28	136.0	103.8	32.2		
12	136.4	111.4	25.0		
24	137.9	106.7	31.2		
3	138.5	110.6	27.9		
18	139.3	105.5	33.8		
21	141.0	110.4	30.6		
10	142.5	113.6	28.9		
7	143.3	109.8	33.7		
6	143.5	109.8	33.7		
25	143.5	111.1	32.4		
8	144.7	113.1	31.6		
20	145.5	115.1	30.4		
14	145.6	113.4	32.2		
23	146.2	105.4	40.8		
26	146.3	116.2	30.1		
17	147.5	117.9	29.6		
15	147.8	116.5	31.3		
13	148.0	113.3	34.7		
9	149.5	113.6	35.9		
5	149.7	111.5	38.2		
22	150.1	112.5	37.6		
27	151.2	117.4	33.8		
43	152.0	121.0	31.0		
16	152.1	117.9	34.2		
29	152.8	114.1	38.7		
42	152.9	121.0	31.9		
2	153.3	118.2	35.1		
1	153.3	121.2	32.1		
44	153.9	120.0	33.9		
32	154.0	120.1	33.9		
36	154.0	121.0	39.0		
47	154.2	120.0	34.2		
39	154.4	118.7	35.7		
35	154.6	116.4	38.2		
33	154.6	117.2	37.4		
31	154.6	120.1	34.5		
40	154.6	122.7	31.9		
46	155.4	125.4	30.0		
49	156.9	122.7	34.2		
37	157.0	120.4	36.6		
41	157.2	122.9	34.3		
34	157.7	120.8	36.9		
48	157.7	123.2	34.5		
53	157.7	127.0	30.7		
63	157.8	120.8	37.0		
64	157.8	122.5	35.3		



55	158.1	119.4	36.7
59	158.8	120.7	38.1
62	158.8	121.7	37.1
66	159.3	124.5	34.8
56	159.5	119.9	39.6
52	159.9	120.7	39.2
38	159.9	121.1	38.8
58	160.1	127.9	32.2
30	160.4	119.5	41.0
54	161.0	124.2	36.8
45	161.3	129.7	31.6
57	161.4	117.6	43.8
30	162.0	127.2	34.8
61	166.7	130.1	36.6

avg. 150.9 117.2 33.7 Devices 60 and 63 not included  
std. 9.1 6.6 4.3

Rubber Diaphragm

<u>Device</u>	<u>Act. Vol.</u>	<u>Spec. 1</u>	<u>Spec. 2</u>
11	103.3	.027	.026
19	120.7	.057	.057
51	136.9	.050	.051
7	134.7	.059	.058
28	136.0	.051	.050
5	177.9	.101	.100
24	150.1	.102	.103
27	152.2	.102	.107
93	152.0	.105	.105
4	152.1	.097	.099
54	161.0	.105	.105
45	161.3	.106	.105
57	161.4	.120	.121
50	170.0	.103	.102
51	166.7	.116	.116

**Rubber Diaphragm, Round Kapton Evaluation**

Device #	Test	Diaphragm	Offset	Initial		Post Vac		Final		Remarks
				act	rel	act	rel	act	rel	
322-15-01	Thermal cycle	control	0.004	126	108	126	107	119	99	T-shift @ 161 deg. C.
322-15-02	Thermal cycle	control	0.004	127	97	126	100	112	91	T-shift @ 143 deg. C.
322-15-03	Thermal cycle	control	0.004	126	97	126	98	97	74	T-shift @ 143 deg. C.
322-15-04	Thermal cycle	control	0.008	139	142	160	149	139	110	
322-15-05	Thermal cycle	control	0.008	232	205	234	208	161	136	
322-15-06	Thermal cycle	control	0.008	177	152	179	165	127	162	
322-15-07	Thermal cycle	rubber	0.004	148	108	146	104	133	102	T-shift @ 174 deg. C.
322-15-08	Thermal cycle	rubber	0.004	169	109	197	106	112	87	T-shift @ 143 deg. C.
322-15-09	Thermal cycle	rubber	0.004	184	103	120	104	104	98	T-shift @ 177 deg. C.
322-15-10	Thermal cycle	rubber	0.008	178	148	176	165	136	110	
322-15-11	Thermal cycle	rubber	0.008	180	106	168	162	148	116	
322-15-12	Thermal cycle	rubber	0.008	184	163	188	181	158	106	
322-15-13	Thermal cycle	round	0.004	198	135	189	136	126	100	
322-15-14	Thermal cycle	round	0.004	184	129	162	130	126	104	
322-15-15	Thermal cycle	round	0.004	184	129	187	158	—	—	Lack: 21 T-cyc, 30,785 Imp.
322-15-16	Thermal cycle	round	0.008	206	200	206	205	181	181	
322-15-17	Thermal cycle	round	0.008	211	184	217	189	180	129	
322-15-18	Thermal cycle	round	0.008	206	204	276	242	179	134	
Device #	Test	Diaphragm	Offset	Initial		Post Vac		-40 deg.C	+175 deg C	Remarks
				act	rel	act	rel	act	rel	
322-15-19	Cel. shift @ temp	control	0.004	149	110	146	128	106	135	111 83
322-15-20	Cel. shift @ temp	control	0.004	126	108	126	107	137	108	98 72
322-15-21	Cel. shift @ temp	control	0.004	131	103	133	106	145	122	89 88
322-15-22	Cel. shift @ temp	control	0.008	156	137	163	155	168	176	116 92
322-15-23	Cel. shift @ temp	control	0.008	159	154	165	168	202	183	117 89
322-15-24	Cel. shift @ temp	control	0.008	174	140	176	146	229	202	119 89
322-15-25	Cel. shift @ temp	rubber	0.004	148	87	147	106	178	149	no continuity at 175 deg. C
322-15-26	Cel. shift @ temp	rubber	0.004	148	112	146	112	169	127	97 71
322-15-27	Cel. shift @ temp	rubber	0.004	129	106	181	107	146	116	70 84
322-15-28	Cel. shift @ temp	rubber	0.008	212	186	218	191	216	204	130 98
322-15-29	Cel. shift @ temp	rubber	0.008	196	187	199	188	279	242	134 101
322-15-30	Cel. shift @ temp	rubber	0.008	206	200	260	206	212	378	129 98
322-15-31	Cel. shift @ temp	round	0.004	156	127	167	126	169	145	116 97
322-15-32	Cel. shift @ temp	round	0.004	129	130	160	133	168	147	128 108
322-15-33	Cel. shift @ temp	round	0.004	146	116	160	116	168	131	98 79
322-15-34	Cel. shift @ temp	round	0.008	193	174	195	172	234	206	132 107
322-15-35	Cel. shift @ temp	round	0.008	209	170	209	184	204	217	138 110
322-15-36	Cel. shift @ temp	round	0.008	206	204	244	216	216	260	143 118
Device #	Test	Diaphragm	Offset	Initial		Post Vac		250K cyc	500K cyc	Remarks
				act	rel	act	rel	act	rel	
322-15-37	Dirt over fil	control	0.004	130	102	130	102	123	99	124 107
322-15-38	Dirt over fil	control	0.004	122	112	135	113	180	108	126 109
322-15-39	Dirt over fil	control	0.004	129	101	128	103	129	103	125 108

**Rubber Diaphragm, Round Kapton Evaluation**

Drift over life										
Device #	Test	Diaphragm	Offset	act	rel	act	rel	act	rel	act
322-15-43	Drift over life	control	0.008	183	132	178	161	178	154	172
322-15-41	Drift over life	control	0.008	198	179	207	178	206	176	200
322-15-42	Drift over life	control	0.008	207	174	218	164	202	179	201
322-15-49	Drift over life	rubber	0.004	138	107	144	108	139	90	130
322-15-44	Drift over life	rubber	0.004	134	102	137	102	131	104	128
322-15-45	Drift over life	rubber	0.004	139	109	147	118	139	113	135
322-15-48	Drift over life	rubber	0.008	230	189	234	184	209	169	201
322-15-47	Drift over life	rubber	0.008	184	169	193	154	169	166	171
322-15-46	Drift over life	rubber	0.008	209	177	218	172	208	172	204
322-15-49	Drift over life	round	0.004	158	136	166	131	160	126	148
322-15-50	Drift over life	round	0.004	157	125	158	137	161	125	160
322-15-51	Drift over life	round	0.004	168	131	167	134	169	124	160
322-15-52	Drift over life	round	0.008	224	195	233	185	222	168	217
322-15-53	Drift over life	round	0.008	265	208	242	250	247	204	230
322-15-54	Drift over life	round	0.009	232	207	236	211	225	194	218
Initial Post Vac 450 hrs.										
Device #	Test	Diaphragm	Offset	act	rel	act	rel	act	rel	Remarks
322-15-55	High temp aging	control	0.004	191	104	132	99	108	104	
322-15-56	High temp aging	control	0.004	128	102	129	99	160	107	
322-15-57	High temp aging	control	0.008	158	132	161	136	160	140	
322-15-58	High temp aging	control	0.009	191	173	163	179	178	182	
322-15-59	High temp aging	rubber	0.004	148	109	148	112			device failure not related to aging
322-15-60	High temp aging	rubber	0.004	148	111	147	108	172	119	
322-15-61	High temp aging	rubber	0.008	198	160	201	168	198	161	
322-15-62	High temp aging	rubber	0.008	172	154	178	154	180	146	
322-15-63	High temp aging	round	0.004	157	180	158	132	170	138	
322-15-64	High temp aging	round	0.004	157	193	159	132	171	138	
322-15-65	High temp aging	round	0.008	181	168	185	168	187	161	
322-15-66	High temp aging	round	0.008	238	208	240	210	218	187	
Room temp. -40 deg.C +178 deg.C										
Device #	Test	Diaphragm	Offset	act	rel	act	rel	act	rel	Remarks
322-15-67	Char at temp.	control	prod.	129	93	138	112	98	66	
322-15-68	Char at temp.	control	prod.	129	104	144	128	97	77	
322-15-69	Char at temp.	control	prod.	131	99	141	117	94	49	
322-15-70	Char at temp.	control	prod.	128	108	159	108	49	35	
322-15-71	Char at temp.	rubber	prod.	117	84	133	100			no continuity at 178 deg. C
322-15-72	Char at temp.	rubber	prod.	140	106	180	125	74	58	
322-15-73	Char at temp.	rubber	prod.	142	108	169	132	84	78	
322-15-74	Char at temp.	rubber	prod.	140	109	162	126	77	59	
322-15-75	Char at temp.	round	prod.	124	120	165	139	100	66	
322-15-76	Char at temp.	round	prod.	146	121	163	129	104	87	
322-15-77	Char at temp.	round	prod.	180	122	153	141	116	86	
322-15-78	Char at temp.	round	prod.	140	124	176	142	103	93	